



**AUSTRALIAN ATOMIC ENERGY COMMISSION
RESEARCH ESTABLISHMENT**

LUCAS HEIGHTS RESEARCH LABORATORIES

**RESULTS OF PIPE BEND ANALYSIS
PART XI: STRESS DISTRIBUTIONS IN FLANGED PIPE ELBOWS
FROM IN-PLANE BENDING**

by

J.F. WHATHAM

July 1984

ISBN 0 642 59796 0

AUSTRALIAN ATOMIC ENERGY COMMISSION
RESEARCH ESTABLISHMENT
LUCAS HEIGHTS RESEARCH LABORATORIES

RESULTS OF PIPE BEND ANALYSIS
PART XI: STRESS DISTRIBUTIONS IN FLANGED PIPE ELBOWS
FROM IN-PLANE BENDING

by

J.F. WHATHAM

ABSTRACT

Graphs of outside surface stress distributions are presented and numerical values of stresses on inside and outside surfaces tabulated for a wide range of flange-ended pipe elbows subjected to in-plane bending; calculations are based on linear thin shell theory.

National Library of Australia card number and ISBN 0 642 59796 0

The following descriptors have been selected from the INIS Thesaurus to describe the subject content of this report for information retrieval purposes. For further details please refer to IAEA-INIS-12 (INIS: Manual for Indexing) and IAEA-INIS-13 (INIS: Thesaurus) published in Vienna by the International Atomic Energy Agency.

BENDING; EXPERIMENTAL DATA; FLANGES; PIPES; SPATIAL DISTRIBUTION; STRESS ANALYSIS; STRESSES

CONTENTS

1. INTRODUCTION	1
2. STRESS DERIVATION	1
3. ELBOW CONFIGURATION AND RESULTS	3
4. ACKNOWLEDGEMENT	3
5. REFERENCES	3
Figure 1 Pipe bend configuration	5
Figure 2 Pipe middle surface	6
Figure 3 Element of pipe middle surface	7
Appendix A Stresses in flanged pipe elbows from in-plane bending	9

1. INTRODUCTION

The objective of this report is to present the surface stress distributions for a range of flanged elbows subjected to an in-plane bending moment, to assist pipework design engineers, and to provide analytic solutions for checking numerical solution methods. Wall thicknesses vary from one to ten per cent of the pipe radius with bend radii two, three and five times the pipe radius.

Stresses were calculated by the thin shell theory of Novozhilov [1970] and details of the analysis have been published [Whatham 1982, 1983]. The assumptions were that:

- (i) the pipe wall is thin ($t/r < 0.3$),
- (ii) normal stresses through the wall are negligible,
- (iii) normals through the wall remain normal to it and unchanged in length, and
- (iv) the flanges are infinitely stiff.

2. STRESS DERIVATION

The flanged pipe elbow configuration is shown in Figure 1, and a segment of a curved pipe middle surface, an imaginary surface mid-way between the inner and outer surfaces, is shown in Figure 2 with one end flanged. An element of the middle surface in Figure 3 is supposed to have forces T_θ , T_η , $T_{\theta\eta}$, $T_{\eta\theta}$, N_θ , N_η and moments M_θ , M_η , $M_{\eta\theta}$, $M_{\theta\eta}$ per unit length acting on its edges and the solution was obtained by thin shell theory in terms of these forces and moments; the stresses presented in this report were then derived as follows:

$$\text{Hoop stress: } \sigma_{\theta\theta} = T_\theta/t + 12zM_\theta/t^3 ,$$

$$\text{Axial stress: } \sigma_{\eta\eta} = T_\eta/t + 12zM_\eta/t^3 , \quad (1)$$

$$\text{Shear stress: } \sigma_{\eta\theta} = S/t + z(12H/t^3 - S/rt) ,$$

where $S = T_{\theta\eta} - M_{\eta\theta}/r_\eta = T_{\eta\theta} - M_{\theta\eta}/r$, and $H = M_{\theta\eta} = M_{\eta\theta}$.

These stresses were close to those derived from beam theory when considering a straight pipe subjected to bending or torsion; if a moment M is

applied to a straight pipe, Novozhilov's theory [see Whatham 1981] gives

$$\begin{aligned} T_{\eta} &= \frac{12M \cos \theta}{(12+\gamma)\pi r^2} , \\ M_{\eta} &= \frac{\gamma M \cos \theta}{(12+\gamma)\pi r} , \\ T_{\theta} &= M_{\theta} = S = H = 0 , \end{aligned} \quad (2)$$

where $\gamma = (t/r)^2$.

Substituting in Equations (1), the axial stress distribution approximates that from beam theory, which is

$$\sigma_{\eta\eta} = \frac{M(r+z)\cos\theta}{\pi r^3 t(1+\gamma/4)} . \quad (3)$$

If torque T is applied to a straight pipe, Novozhilov's theory gives

$$\begin{aligned} S &= \frac{3T}{2(3+\gamma)\pi r^2} , \\ H &= \frac{\gamma T}{4(3+\gamma)\pi r} , \\ T_{\eta} &= M_{\eta} = T_{\theta} = M_{\theta} = 0 . \end{aligned} \quad (4)$$

Substituting in Equations (1), the shear stress distribution approximates that from beam theory, which is

$$\sigma_{\eta\theta} = \frac{T(r+z)}{2\pi r^3 t(1+\gamma/4)} . \quad (5)$$

Equations (1) assume that the strains are linearly distributed through the pipe wall; this is not true for curved shells and the stresses $\sigma'_{\theta\theta}$, $\sigma'_{\eta\eta}$, $\sigma'_{\eta\theta}$ derived by the following equations are theoretically more accurate, even though the results do not agree with beam theory in the case of straight pipes:

$$\begin{aligned} \sigma'_{\theta\theta} &= \sigma_{\theta\theta} - z(A+vB)/(1-v^2) , \\ \sigma'_{\eta\eta} &= \sigma_{\eta\eta} - z(B+vA)/(1-v^2) , \\ \sigma'_{\eta\theta} &= \sigma_{\eta\theta} - \frac{z}{2r} \left[(r/(r+z) + r/(r_{\eta}+z))\sigma_{\eta\theta} - (r_{\eta}-r)\bar{\sigma}_{\eta\theta}/(r_{\eta}+z) \right] , \end{aligned} \quad (6)$$

where

$$r_{\eta} = r + R/\cos \theta ,$$

$$A = (\sigma_{\theta\theta} - \nu\sigma_{\eta\eta})/(r+z) ,$$

$$B = (\sigma_{\eta\eta} - \nu\sigma_{\theta\theta})/(r_{\eta}+z) , \quad \text{and}$$

$$\bar{\sigma}_{\eta\theta} = \frac{1}{2}\sigma_{\eta\theta} \text{ (inside)} + \frac{1}{2}\sigma_{\eta\theta} \text{ (outside)}.$$

3. ELBOW CONFIGURATION AND RESULTS

Stresses are duplicated each side of the bend in Figure 1 because of lateral symmetry, and hoop and axial stresses on the outside surface from $\theta = 0$ to 180° are plotted in Appendix A, together with the stresses on theoretically unterminated or unflanged pipe bends in which case the stresses are independent of ϕ . The maxima and minima of the latter curves are given by the floating numbers.

The two graphs for each pipe bend are accompanied by tabulated stresses from which the graphs were constructed. In addition, since shear stresses and inside surface stresses are also given and a linear variation through the wall is assumed, the complete stress state in each pipe bend is provided.

A computer program package BENDPAC, written in FORTRAN IV and ASSEMBLER for an IBM3031 computer and designed to calculate the stresses in and the flexibility of flanged pipe elbows under in-plane or out-of-plane loading, is available from the Australian Atomic Energy Commission, the Risley Nuclear Power Development Establishment, Cheshire, UK, or the National Energy Software Center, Argonne National Laboratory, USA.

4. ACKNOWLEDGEMENT

The author acknowledges the advice and encouragement of Professor J.J. Thompson of the School of Nuclear Engineering, University of New South Wales.

5. REFERENCES

- Novozhilov, V.V. [1970] - Thin Shell Theory. 2nd Augmented and Revised Edition, Wolters-Noordhoff, Gröningen, The Netherlands.
- Whatham, J.F. [1981] - Thin shell equations for circular pipe bends. J. Nucl. Eng. Des., 65(1)77.

Whatham, J.F. [1982] - Analysis of circular pipe bends with flanged ends. J. Nucl. Eng. Des., 72(2)175.

Whatham, J.F. [1983] - Thin shell analysis of flanged pipe bends. Trans. Inst. Eng. Aust., CE25(1)1.

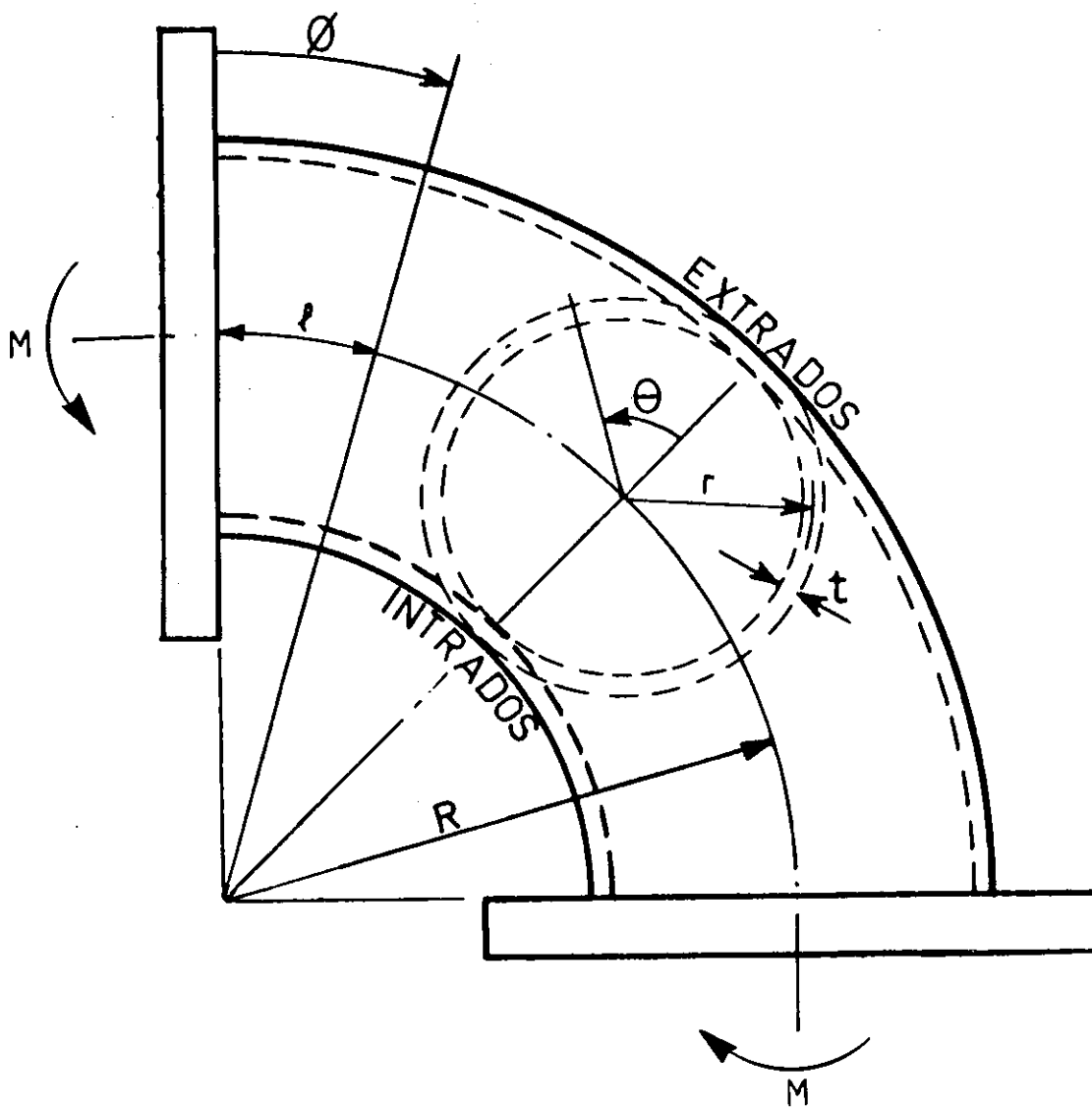


FIGURE 1. PIPE BEND CONFIGURATION

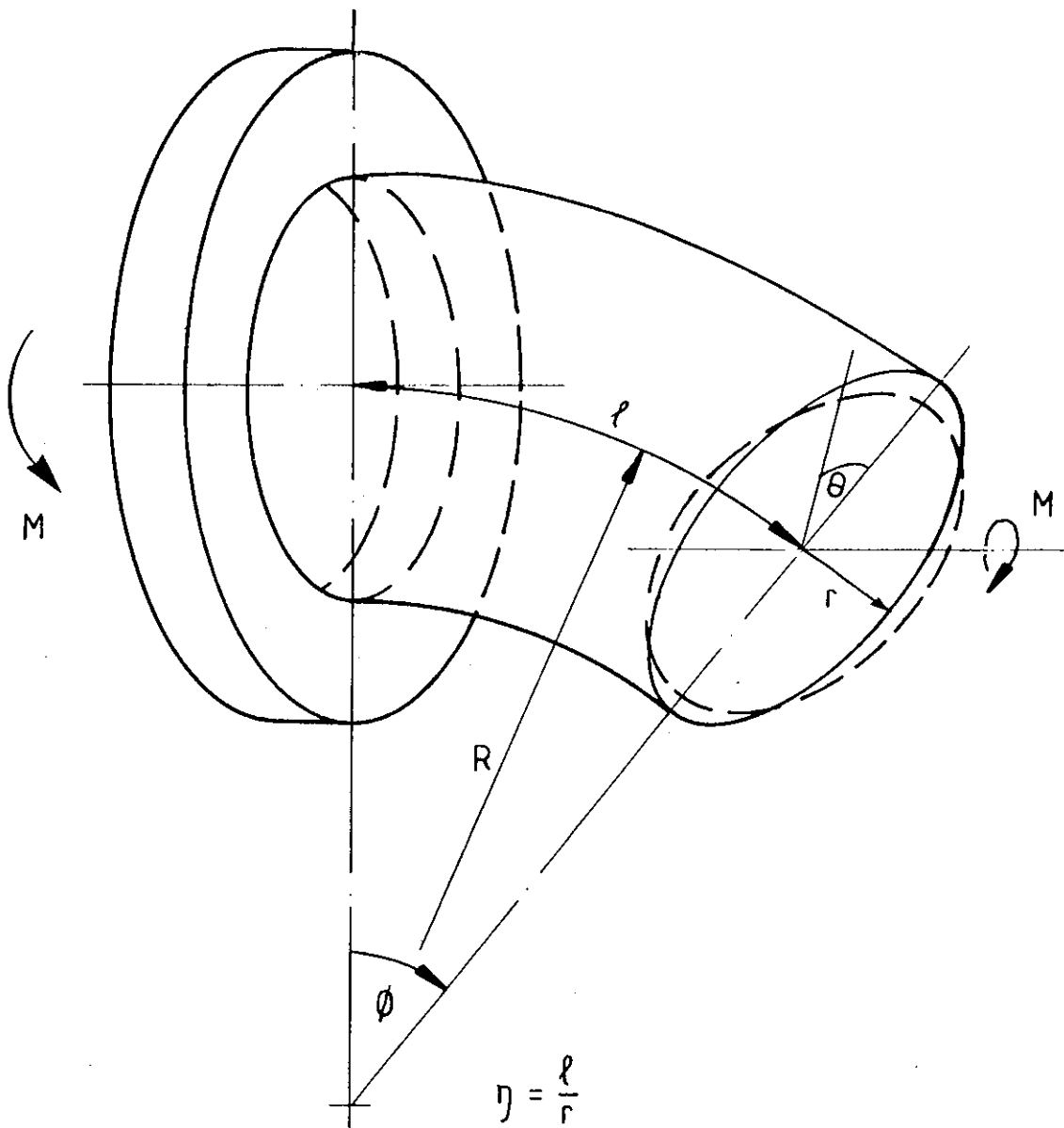


FIGURE 2. PIPE MIDDLE SURFACE

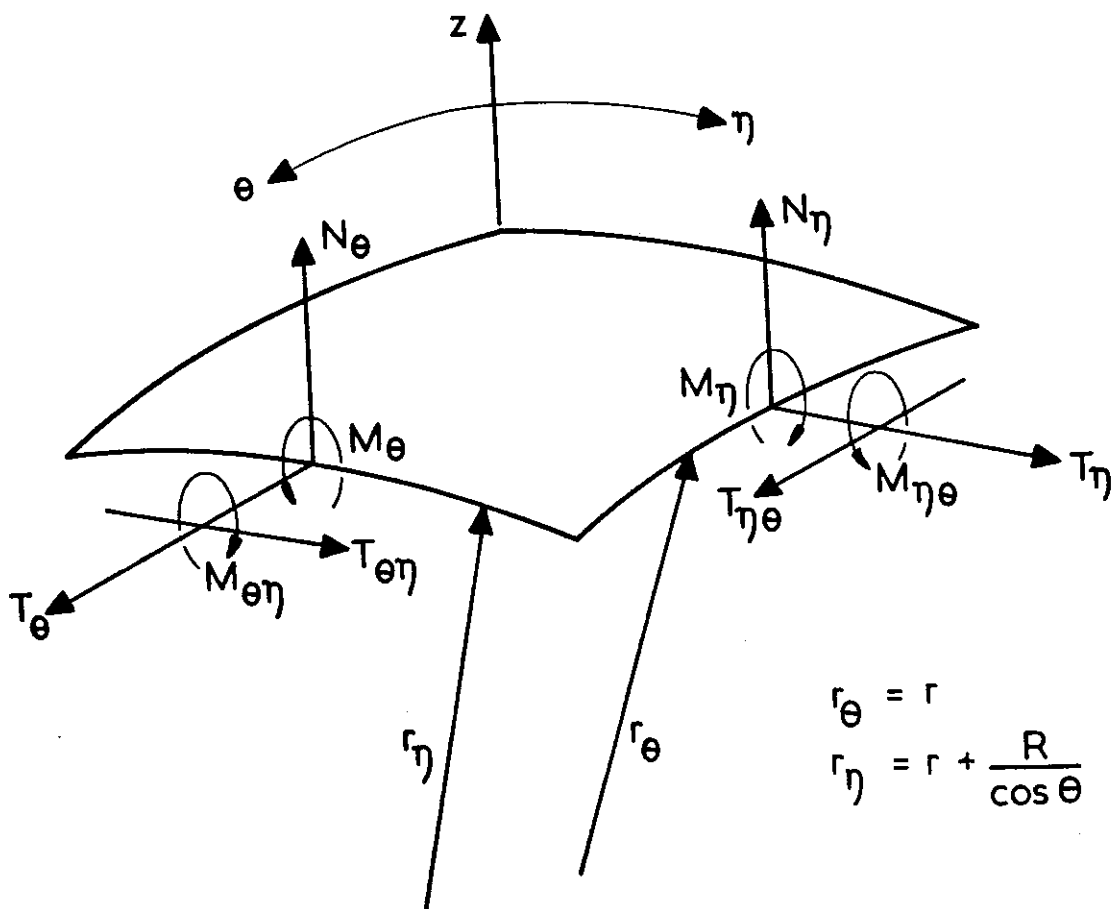


FIGURE 3. ELEMENT OF PIPE MIDDLE SURFACE

APPENDIX A

STRESSES IN FLANGED PIPE ELBOWS FROM IN-PLANE BENDING

Parameters of pipe elbows considered:

$$R/r = 2, 3, 5$$

$$t/r = 0.01, 0.02, 0.05, 0.1$$

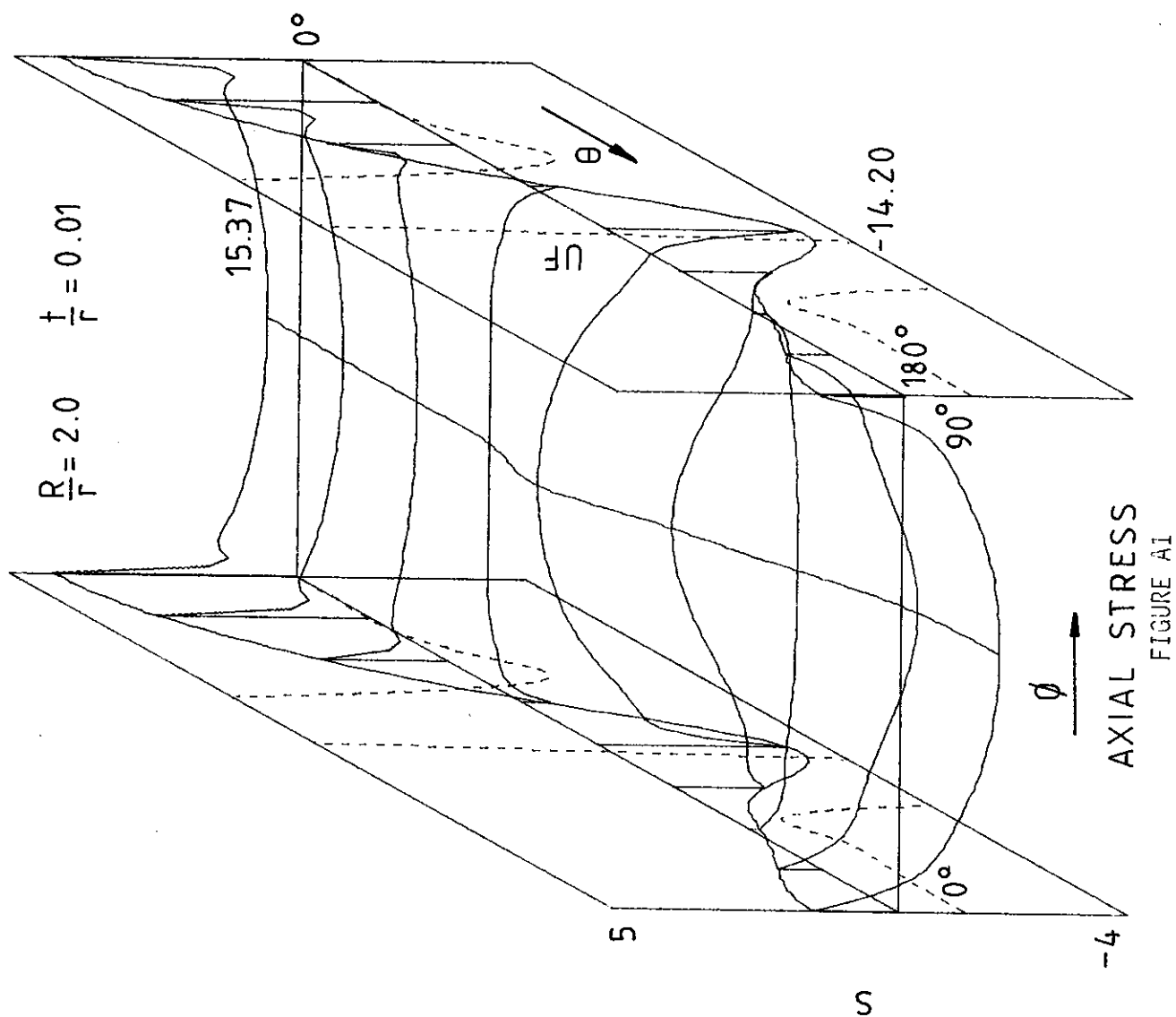
$$\nu = 0.3$$

$$\text{Stress factor (S)} = \text{stress} \cdot \pi r^2 t / M$$

$$UF = \text{unflanged pipe bend.}$$

$$\text{Diameter expansion factor} = \text{diameter expansion} \cdot \pi r E t / M$$

$$E = \text{Young's modulus}$$



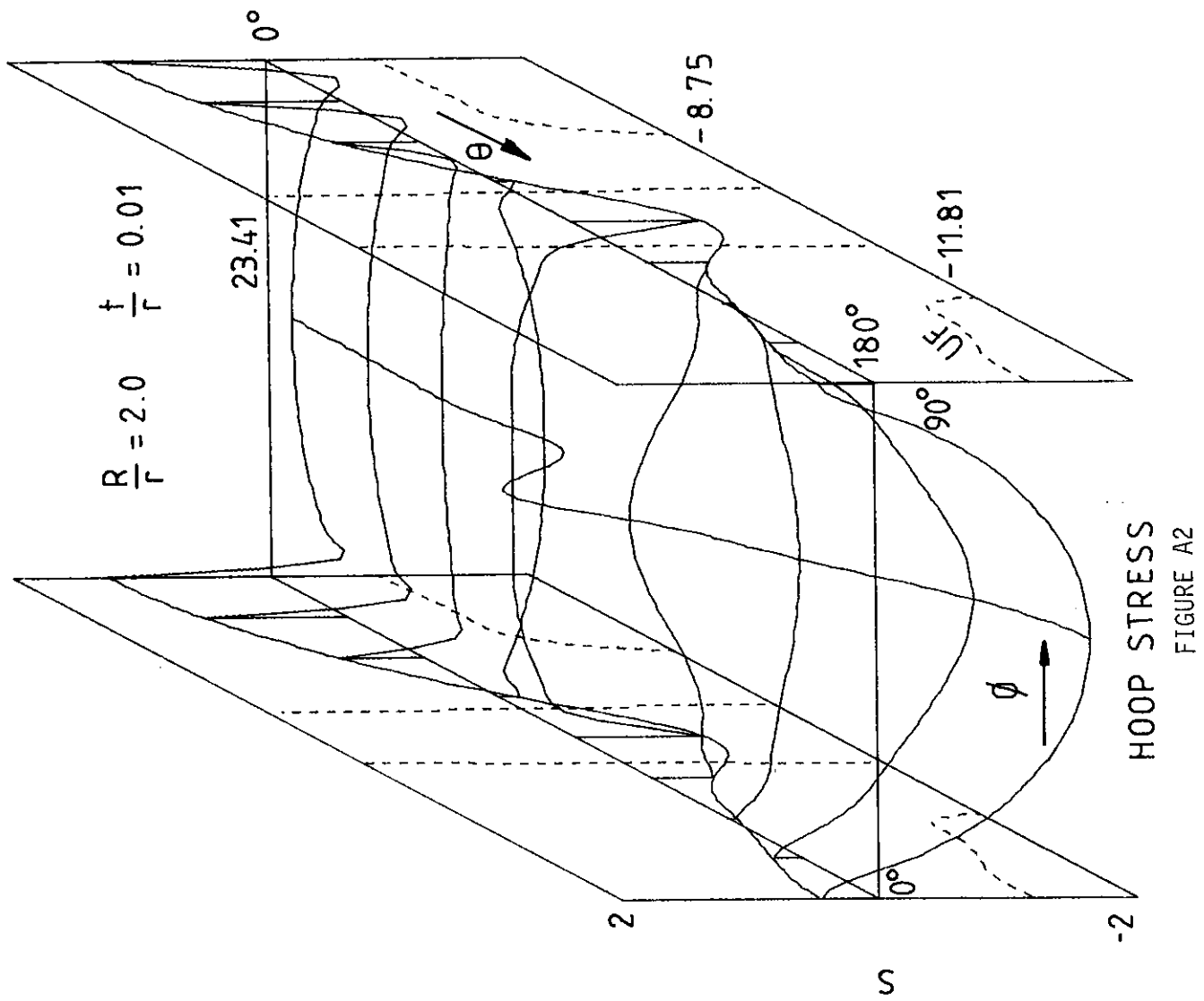


TABLE A1

$$R/r = 2.0 \quad t/r = 0.01$$

beta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.2642	-0.4366	-0.3301	-0.2614	-0.2185	-0.1949	-0.1873	-0.1949	-0.2185	-0.2614	-0.3301	-0.4366	1.2642
22.5	1.1105	-0.3917	-0.3062	-0.2491	-0.2122	-0.1914	-0.1847	-0.1914	-0.2122	-0.2491	-0.3062	-0.3917	1.1105
45.0	0.6531	-0.2617	-0.2318	-0.2057	-0.1845	-0.1711	-0.1666	-0.1711	-0.1845	-0.2057	-0.2318	-0.2617	0.6531
67.5	-0.1537	-0.0881	-0.2085	-0.2908	-0.3430	-0.3706	-0.3791	-0.3706	-0.3430	-0.2908	-0.2085	-0.0881	-0.1537
90.0	-1.0059	0.2297	0.3552	0.4429	0.4712	0.4712	0.4680	0.4712	0.4712	0.4429	0.3552	0.2297	-1.0059
12.5	-0.4848	-0.3985	-0.3785	-0.2335	-0.0488	0.0974	0.1523	0.0974	-0.0488	-0.2335	-0.3785	-0.3985	-0.4848
35.0	-0.0644	-0.3267	-0.3801	-0.4521	-0.5228	-0.5675	-0.5821	-0.5675	-0.5228	-0.4521	-0.3801	-0.3267	-0.0644
57.5	0.2323	-0.3029	-0.6319	-0.8747	-1.1080	-1.2798	-1.3433	-1.2798	-1.1080	-0.8747	-0.6319	-0.3029	0.2323
80.0	0.4401	-0.3127	-0.9126	-1.2575	-1.4895	-1.6136	-1.6506	-1.6136	-1.4895	-1.2575	-0.9126	-0.3127	0.4401
													-1.1951

beta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	4.2141	1.2811	0.9423	0.7368	0.6096	0.5402	0.5181	0.5402	0.6096	0.7368	0.9423	1.2811	4.2141	0.0234
22.5	3.7017	1.2050	0.9178	0.7355	0.6191	0.5542	0.5334	0.5542	0.6191	0.7355	0.9178	1.2050	3.7017	0.0548
45.0	2.1772	0.9353	0.8059	0.6994	0.6204	0.5725	0.5566	0.5725	0.6204	0.6994	0.8059	0.9353	2.1772	-0.9486
67.5	-0.5125	0.4046	0.5733	0.6278	0.6291	0.6169	0.6108	0.6169	0.6291	0.6278	0.5733	0.4046	-0.5125	5.2491
90.0	-3.3532	-0.5834	0.0925	0.5873	0.8884	1.0452	1.0931	1.0452	0.8884	0.5873	0.0925	-0.5834	-3.3532	8.0969
12.5	-1.6158	-1.2969	-1.0746	-0.6854	-0.3009	-0.0273	0.0711	-0.0273	-0.3009	-0.6854	-1.0746	-1.2969	-1.6158	-11.7260
35.0	-0.2148	-0.6335	-0.6765	-0.7523	-0.7970	-0.8101	-0.8112	-0.8101	-0.7970	-0.7523	-0.6765	-0.6335	-0.2148	-0.9160
57.5	0.7744	-0.4675	-0.7370	-1.0220	-1.3147	-1.5318	-1.6129	-1.5318	-1.3147	-1.0220	-0.7370	-0.4675	0.7744	-1.2391
80.0	1.4670	-0.4794	-0.9939	-1.3332	-1.5739	-1.6912	-1.7234	-1.6912	-1.5739	-1.3332	-0.9939	-0.4794	1.4670	-1.1878

beta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.5447	0.3874	0.2537	0.1634	0.0979	0.0459	0.0	-0.0459	-0.0979	-0.1634	-0.2537	-0.3874	-0.5447
45.0	0.9228	0.7183	0.4971	0.3331	0.2047	0.0975	0.0	-0.0975	-0.2047	-0.3331	-0.4971	-0.7183	-0.9228
67.5	0.8622	0.8088	0.6335	0.4661	0.3054	0.1509	0.0	-0.1509	-0.3054	-0.4661	-0.6335	-0.8088	-0.8622
90.0	0.0026	0.2156	0.2948	0.2862	0.2210	0.1196	0.0	-0.1196	-0.2210	-0.2862	-0.2948	-0.2156	-0.0026
12.5	-0.7430	-0.5493	-0.3260	-0.1758	-0.0868	-0.0352	0.0	0.0352	0.0868	0.1758	0.3260	0.5493	0.7430
35.0	-1.0313	-1.0286	-0.9460	-0.7618	-0.5225	-0.2627	0.0	0.2627	0.5225	0.7618	0.9460	1.0286	1.0313
57.5	-0.9344	-0.9010	-0.8819	-0.7750	-0.5839	-0.3160	0.0	0.3160	0.5839	0.7750	0.8819	0.9010	0.9344
80.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without Flanges	
beta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
90.0	0.0	5.264	9.739	13.381	15.897	17.326	17.784	17.326	15.897	13.381	9.739	5.264	0.0
													701.057

TABLE A2

$$R/r = 2.0 \quad t/r = 0.01$$

theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	-0.2058	-0.4191	-0.2997	-0.2313	-0.1894	-0.1667	-0.1595	-0.1667	-0.1894	-0.2313	-0.2997	-0.4191	-0.2058
22.5	-0.1631	-0.3736	-0.2743	-0.2163	-0.1798	-0.1596	-0.1532	-0.1596	-0.1798	-0.2163	-0.2743	-0.3736	-0.1631
45.0	-0.0487	-0.2346	-0.1923	-0.1657	-0.1471	-0.1362	-0.1326	-0.1362	-0.1471	-0.1657	-0.1923	-0.2346	-0.0487
67.5	0.1118	-0.0091	0.0459	0.1088	0.1567	0.1849	0.1941	0.1849	0.1567	0.1088	0.0459	-0.0091	0.1118
90.0	0.1124	-0.2356	-0.4170	-0.5084	-0.5324	-0.5265	-0.5208	-0.5265	-0.5324	-0.5084	-0.4170	-0.2356	0.1124
12.5	-0.2728	-0.1033	0.0130	0.0181	-0.0396	-0.1019	-0.1274	-0.1019	-0.0396	0.0181	0.0130	-0.1033	-0.2728
35.0	-0.2403	-0.3445	-0.4035	-0.3411	-0.2440	-0.1646	-0.1350	-0.1646	-0.2440	-0.3411	-0.4035	-0.3445	-0.2403
57.5	-0.3155	-0.2713	-0.5984	-0.8044	-0.9518	-1.0378	-1.0649	-1.0378	-0.9518	-0.8044	-0.5984	-0.2713	-0.3155
80.0	-0.4398	-0.2152	-0.6898	-1.0480	-1.3535	-1.5735	-1.6540	-1.5735	-1.3535	-1.0480	-0.6898	-0.2152	-0.4398
													1.1946

theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	-0.6861	1.2503	0.9468	0.7412	0.6138	0.5441	0.5220	0.5441	0.6138	0.7412	0.9468	1.2503	-0.6861
22.5	-0.5437	1.1740	0.9202	0.7390	0.6230	0.5583	0.5375	0.5583	0.6230	0.7390	0.9202	1.1740	-0.5437
45.0	-0.1623	0.9060	0.8018	0.6997	0.6225	0.5752	0.5593	0.5752	0.6225	0.6997	0.8018	0.9060	-0.1623
67.5	0.3725	0.3416	0.5859	0.6918	0.7293	0.7377	0.7383	0.7377	0.7293	0.6918	0.5859	0.3416	0.3725
90.0	0.3745	-0.8208	-0.2181	0.1801	0.4482	0.6013	0.6510	0.6013	0.4482	0.1801	-0.2181	-0.8208	0.3745
12.5	-0.9095	-1.0594	-0.7667	-0.4938	-0.2711	-0.1292	-0.0807	-0.1292	-0.2711	-0.4938	-0.7667	-1.0594	-0.9095
35.0	-0.8011	-0.5904	-0.6874	-0.6635	-0.5981	-0.5387	-0.5154	-0.5387	-0.5981	-0.6635	-0.6874	-0.5904	-0.8011
57.5	-1.0517	-0.3731	-0.6710	-0.9213	-1.0847	-1.1803	-1.2111	-1.1803	-1.0847	-0.9213	-0.6710	-0.3731	-1.0517
80.0	-1.4660	-0.3334	-0.6262	-0.9887	-1.2760	-1.4808	-1.5563	-1.4808	-1.2760	-0.9887	-0.6262	-0.3334	-1.4660
													1.1828

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
22.5	0.5393	0.3932	0.2559	0.1638	0.0977	0.0457	0.0	-0.0457	-0.1638	-0.2559	-0.3932	-0.5393	
45.0	0.9136	0.7302	0.5014	0.3330	0.2035	0.0966	0.0	-0.0966	-0.2035	-0.5014	-0.7302	-0.9136	
67.5	0.8536	0.9036	0.6990	0.5040	0.3241	0.1581	0.0	-0.1581	-0.3241	-0.5040	-0.6990	-0.8536	
90.0	0.0026	0.3140	0.4520	0.4494	0.3485	0.1881	0.0	-0.1881	-0.3485	-0.4494	-0.4520	-0.0026	
12.5	-0.7356	-0.7426	-0.5090	-0.2986	-0.1445	-0.0528	0.0	0.0528	0.1445	0.2986	0.5090	0.7356	
35.0	-1.0210	-1.0734	-1.0353	-0.9248	-0.7053	-0.3828	0.0	0.3828	0.7053	0.9248	1.0353	1.0210	
57.5	-0.9251	-1.0450	-0.9641	-0.7803	-0.5432	-0.2773	0.0	0.2773	0.5432	0.7803	0.9641	0.9251	
80.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	

theta	Phi=0.0	DIAMETER EXPANSION FACTORS										Without		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Flanges
0.0	0.0	-3.490	-5.974	-7.865	-9.156	-9.875	-10.100	-9.875	-9.156	-7.865	-5.974	-3.490	0.0	-330.349

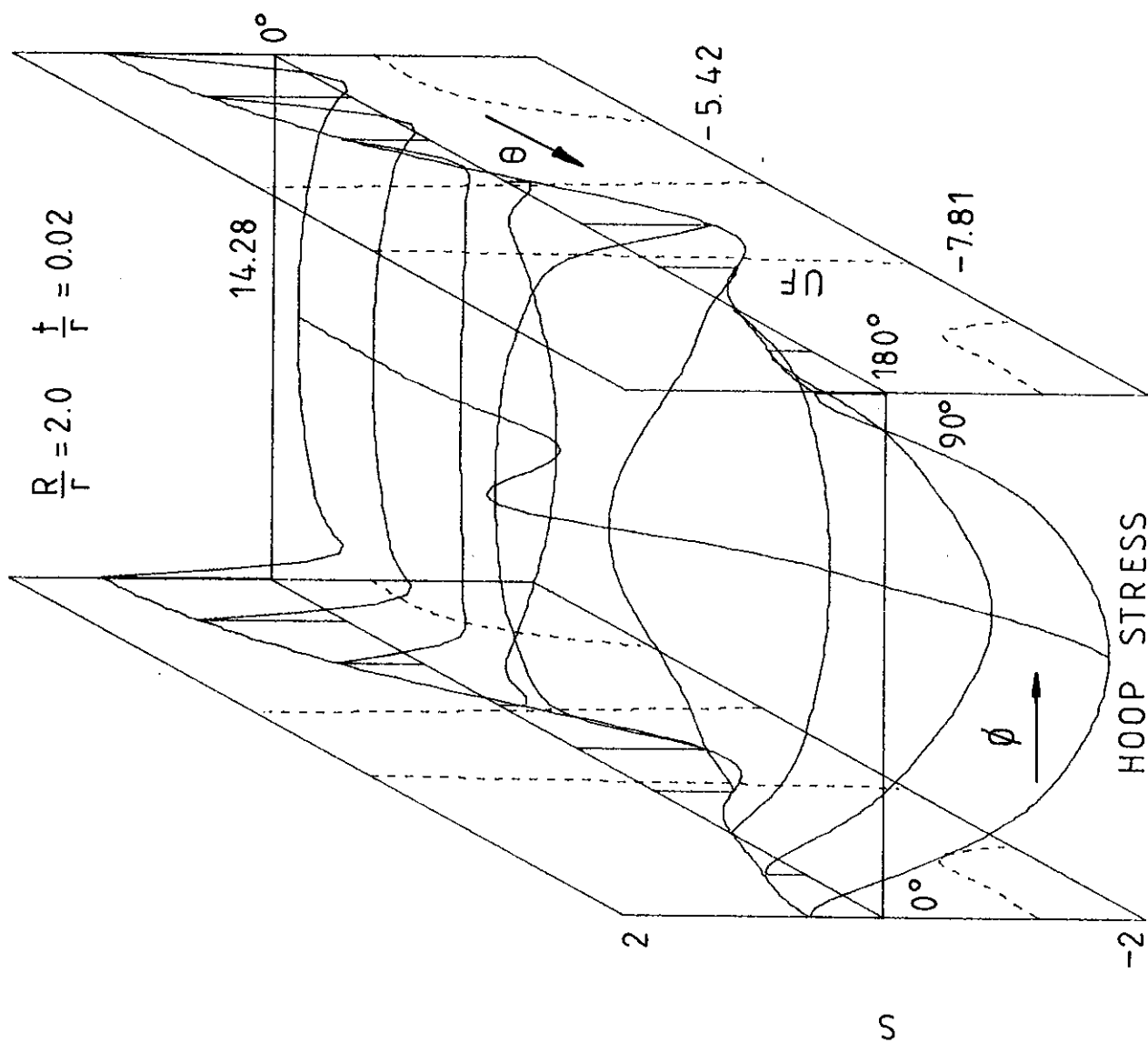


FIGURE A3

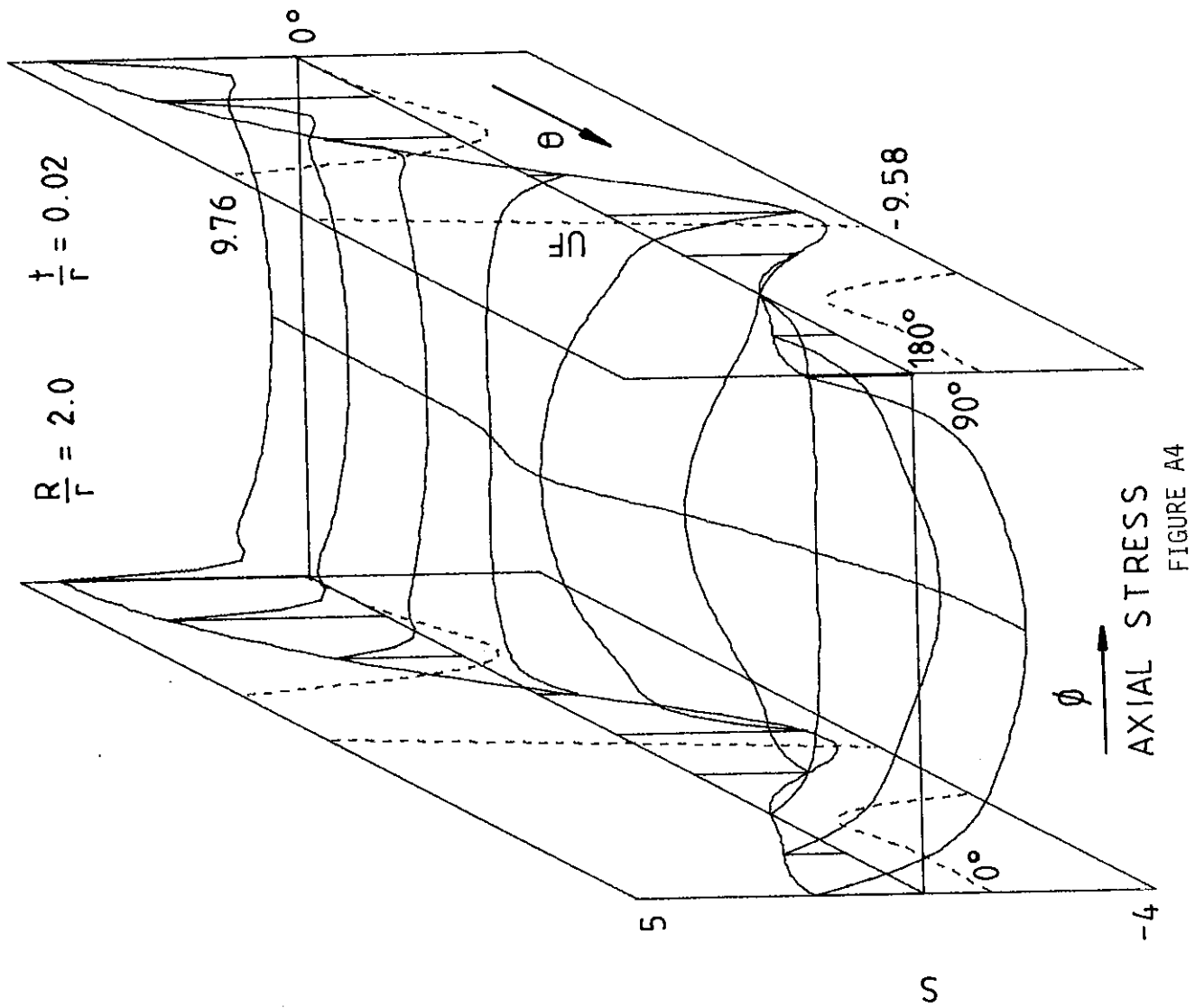


TABLE A3

$$R/r = 2.0 \quad t/r = 0.02$$

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.2901	-0.5004	-0.3373	-0.2706	-0.2275	-0.2037	-0.1961	-0.2037	-0.2275	-0.2706	-0.3373	-0.5004	1.2901
22.5	1.1325	-0.4496	-0.3101	-0.2524	-0.2130	-0.1906	-0.1834	-0.1906	-0.2130	-0.2524	-0.3101	-0.4496	1.1325
45.0	0.6516	-0.3186	-0.2917	-0.2870	-0.2774	-0.2693	-0.2662	-0.2693	-0.2774	-0.2870	-0.2917	-0.3186	0.6516
67.5	-0.2027	-0.0432	-0.1663	-0.2717	-0.3520	-0.4018	-0.4187	-0.4018	-0.3520	-0.2717	-0.1663	-0.0432	-0.2027
90.0	-0.9990	0.2113	0.4186	0.5356	0.6030	0.6311	0.6381	0.6311	0.6030	0.5356	0.4186	0.2113	-0.9990
112.5	-0.6073	-0.3908	-0.2572	-0.0637	0.1493	0.3068	0.3644	0.3068	0.1493	-0.0637	-0.2572	-0.3908	-0.6073
135.0	-0.0061	-0.3593	-0.5405	-0.6352	-0.6999	-0.7359	-0.7468	-0.7359	-0.6999	-0.6352	-0.5405	-0.3593	-0.0061
157.5	0.3261	-0.1604	-0.6312	-0.9259	-1.1643	-1.3373	-1.4010	-1.3373	-1.1643	-0.9259	-0.6312	-0.1604	0.3261
180.0	0.5584	-0.1211	-0.8544	-1.2623	-1.5097	-1.6579	-1.7082	-1.6579	-1.5097	-1.2623	-0.8544	-0.1211	0.5584

OUTSIDE AXIAL STRESS FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	4.3005	0.5985	0.5298	0.5080	0.5298	0.5985	0.7243	0.9279	1.2010	4.3005	0.0684
22.5	3.7749	0.6023	0.5382	0.5176	0.5382	0.6023	0.7177	0.9000	1.1237	3.7749	-0.1654
45.0	2.1721	0.5643	0.5098	0.4917	0.5098	0.5643	0.6545	0.7769	0.8813	2.1721	-0.6659
67.5	-0.6757	0.7641	0.7553	0.7498	0.7553	0.7641	0.7520	0.6723	0.5050	-0.6757	6.2809
90.0	-3.3300	0.9741	1.1515	1.2079	1.1515	0.9741	0.6519	0.1536	-0.4198	-3.3300	5.1399
112.5	-2.0243	-0.3027	-0.0443	0.0472	-0.0443	-0.3027	-0.6805	-1.0881	-1.3468	-2.0243	-9.5356
135.0	-0.0204	-0.9262	-0.9210	-0.9151	-0.9210	-0.9262	-0.9058	-0.8483	-0.6961	-0.0204	-1.3663
157.5	1.0869	-1.4068	-1.6368	-1.7209	-1.6368	-1.4068	-1.1029	-0.8058	-0.3278	1.0869	-1.2056
180.0	1.8613	-1.6563	-1.8226	-1.8798	-1.8226	-1.6563	-1.4069	-1.0709	-0.2876	1.8613	-1.1721

OUTSIDE SHEAR STRESS FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.5121	0.0975	0.0458	0.0	-0.0458	-0.0975	-0.1625	-0.2515	-0.3869	-0.5121	0.0
45.0	0.8594	0.7135	0.0990	0.0	-0.0990	-0.2070	-0.3345	-0.4947	-0.7135	-0.8594	0.0
67.5	0.7622	0.7229	0.1415	0.0	-0.1415	-0.2839	-0.4286	-0.5786	-0.7229	-0.7622	0.0
90.0	0.0190	0.1730	0.1635	0.0869	0.0	-0.0869	-0.1635	-0.2347	-0.1730	-0.0190	0.0
112.5	-0.7009	-0.4637	-0.0827	-0.0351	0.0	0.0351	0.0827	0.1600	0.4637	0.7009	0.0
135.0	-0.9862	-0.9463	-0.4246	-0.2075	0.0	0.2075	0.4246	0.6474	0.9463	0.9862	0.0
157.5	-0.8489	-0.7804	-0.5467	-0.2963	0.0	0.2963	0.5467	0.7158	0.7804	0.8489	0.0
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
90.0	0.0	13.889	15.308	15.778	15.308	13.889	11.520	8.287	4.292	0.0	309.870

TABLE A4

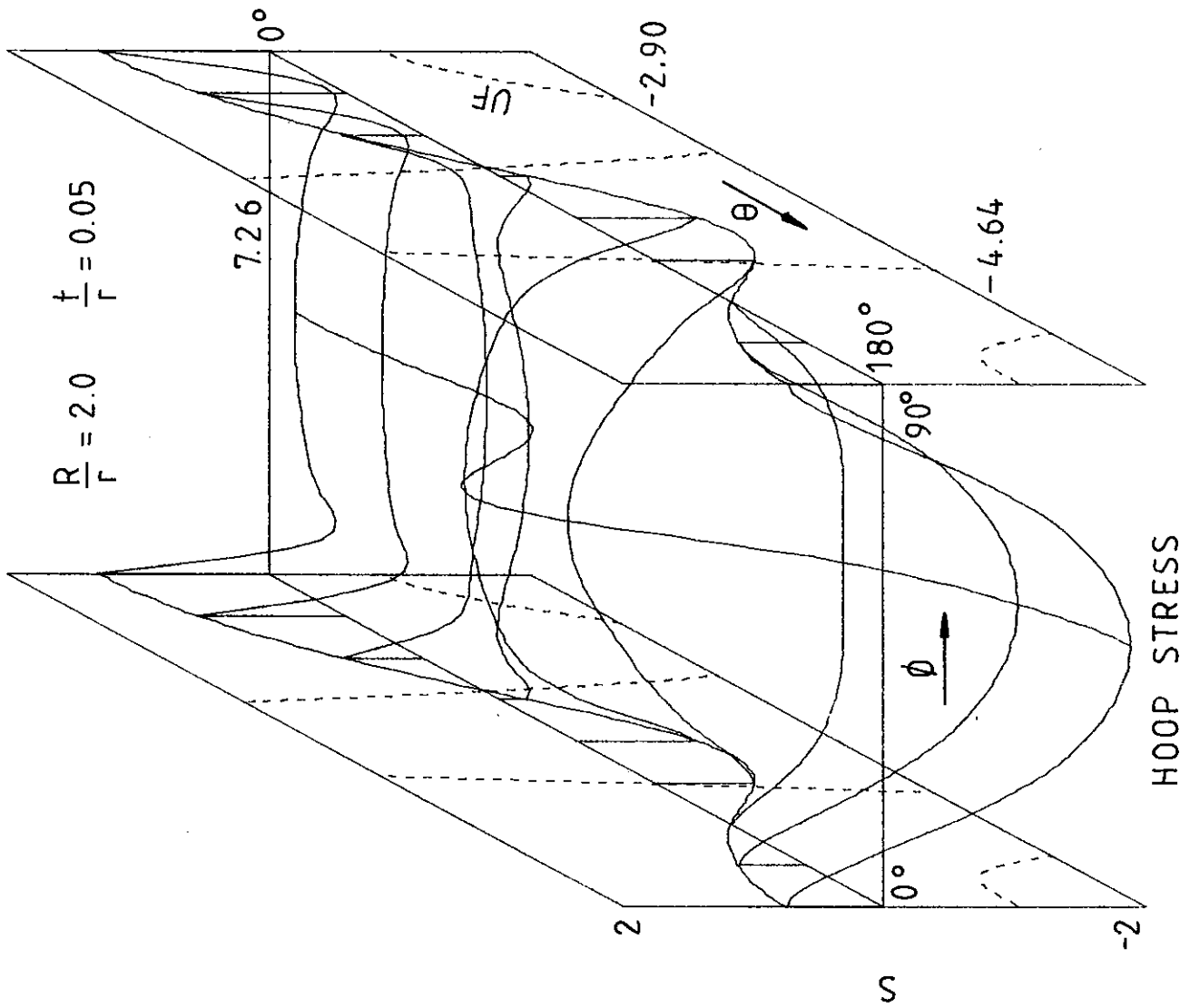
R/r = 2.0 t/r = 0.02

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.2496	-0.4186	-0.2802	-0.2142	-0.1731	-0.1509	-0.1439	-0.1509	-0.1731	-0.2142	-0.2802	-0.4186	-0.2496	0.7474
22.5	-0.2013	-0.3674	-0.2583	-0.2049	-0.1714	-0.1532	-0.1474	-0.1532	-0.1714	-0.2049	-0.2583	-0.3674	-0.2013	0.8725
45.0	-0.0627	-0.1939	-0.1175	-0.0708	-0.0407	-0.0246	-0.0196	-0.0246	-0.0407	-0.0708	-0.1175	-0.1939	-0.0627	3.8834
67.5	0.1590	-0.0938	0.0038	0.0906	0.1713	0.2248	0.2434	0.2248	0.1713	0.0906	0.0038	-0.0938	0.1590	1.5103
90.0	0.2087	-0.3621	-0.4998	-0.6438	-0.7085	-0.7308	-0.7351	-0.7308	-0.7085	-0.6438	-0.4998	-0.3621	0.2087	-17.0399
112.5	-0.1951	-0.1592	-0.0921	-0.1803	-0.3031	-0.4043	-0.4428	-0.4043	-0.3031	-0.1803	-0.0921	-0.1592	-0.1951	3.4071
135.0	-0.2963	-0.2576	-0.2719	-0.1761	-0.0608	0.0186	0.0460	0.0186	-0.0608	-0.1761	-0.2719	-0.2576	-0.2963	3.7993
157.5	-0.4107	-0.2149	-0.5464	-0.7615	-0.8526	-0.8831	-0.8893	-0.8831	-0.8526	-0.7615	-0.5464	-0.2149	-0.4107	1.2179
180.0	-0.5925	-0.1382	-0.5316	-0.9214	-1.2300	-1.4484	-1.5296	-1.4484	-1.2300	-0.9214	-0.5316	-0.1382	-0.5925	1.1829

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.8321	1.2971	0.9352	0.7321	0.6060	0.5370	0.5151	0.5370	0.6060	0.7321	0.9352	1.2971	-0.8321	-0.0336
22.5	-0.6710	1.2183	0.9007	0.7204	0.6043	0.5396	0.5188	0.5396	0.6043	0.7204	0.9007	1.2183	-0.6710	-0.2445
45.0	-0.2089	0.9365	0.7934	0.6942	0.6157	0.5666	0.5499	0.5666	0.6157	0.6942	0.7934	0.9365	-0.2089	0.6610
67.5	0.5301	0.2813	0.6005	0.7400	0.8080	0.8360	0.8433	0.8360	0.8080	0.7400	0.6005	0.2813	0.5301	4.9620
90.0	0.6957	-0.8504	-0.2325	0.1216	0.3649	0.5081	0.5552	0.5081	0.3649	0.1216	-0.2325	-0.8504	0.6957	-4.3042
112.5	-0.6503	-1.0652	-0.7953	-0.5870	-0.4302	-0.3330	-0.3005	-0.3330	-0.4302	-0.5870	-0.7953	-1.0652	-0.6503	-3.4875
135.0	-0.9875	-0.5720	-0.5983	-0.5856	-0.5259	-0.4735	-0.4540	-0.4735	-0.5259	-0.5856	-0.5983	-0.5720	-0.9875	2.7142
157.5	-1.3691	-0.4997	-0.5820	-0.8111	-0.9373	-0.9884	-1.0018	-0.9884	-0.9373	-0.8111	-0.5820	-0.4997	-1.3691	1.1833
180.0	-1.9750	-0.5473	-0.4867	-0.8089	-1.0856	-1.2603	-1.3213	-1.2603	-1.0856	-0.8089	-0.4867	-0.5473	-1.9750	1.1888

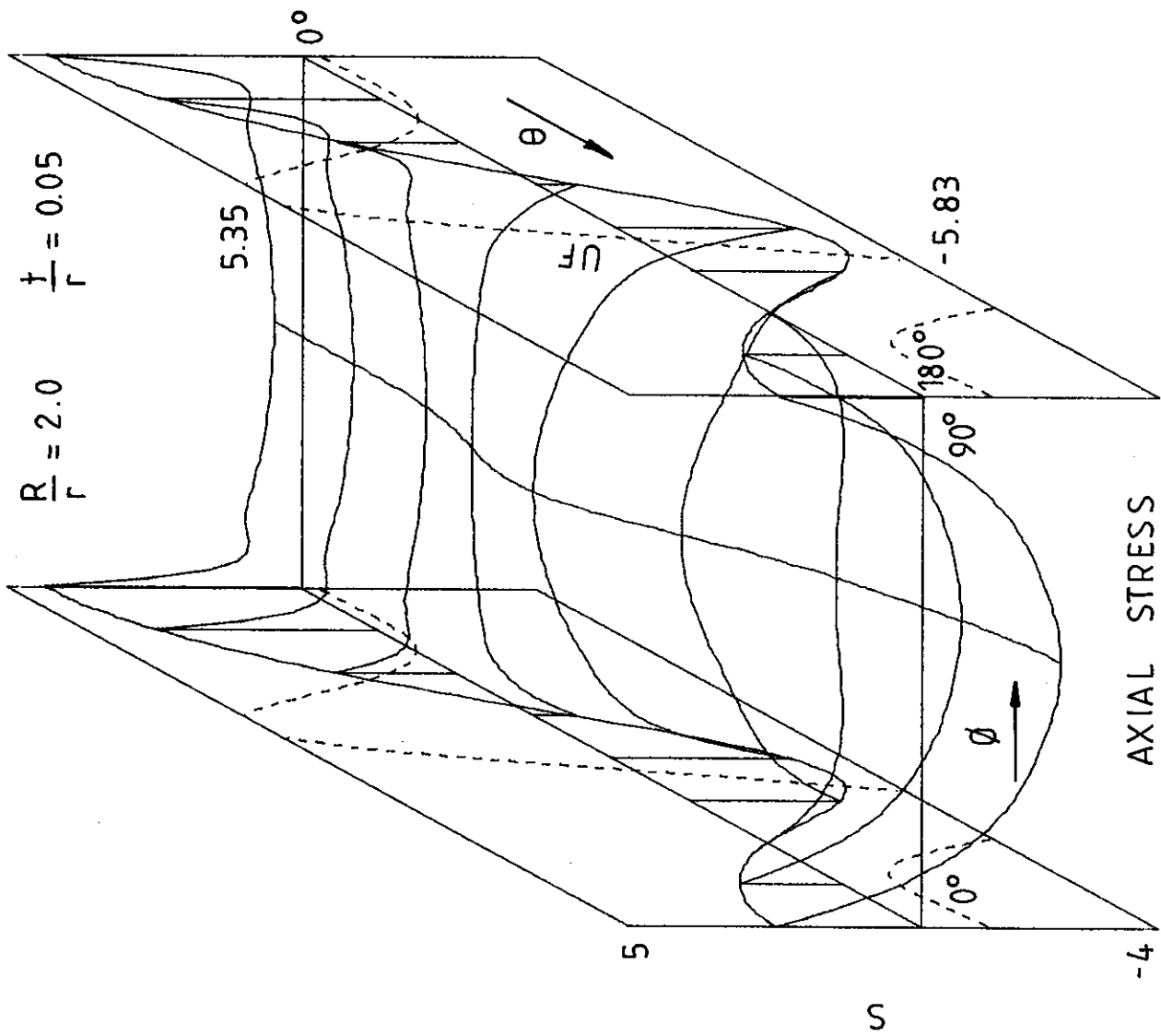
Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.5020	0.3906	0.2536	0.1616	0.0961	0.0449	0.0	-0.0449	-0.0961	-0.1616	-0.2536	-0.3906	-0.5020
45.0	0.8424	0.7508	0.5081	0.3360	0.2043	0.0966	0.0	-0.0966	-0.2043	-0.3360	-0.5081	-0.7508	-0.8424
67.5	0.7471	0.9453	0.7277	0.5323	0.3458	0.1698	0.0	-0.1698	-0.3458	-0.5323	-0.7277	-0.9453	-0.7471
90.0	0.0186	0.3162	0.4501	0.4454	0.3467	0.1878	0.0	-0.1878	-0.3467	-0.4454	-0.4501	-0.3162	-0.0186
112.5	-0.6870	-0.8033	-0.5312	-0.3141	-0.1606	-0.0639	0.0	0.0639	0.1606	0.3141	0.5312	0.8033	0.6870
135.0	-0.9666	-1.1086	-1.0931	-0.9540	-0.7076	-0.3766	0.0	0.3766	0.7076	0.9540	1.0931	1.1086	0.9666
157.5	-0.8321	-0.9809	-0.9433	-0.7897	-0.5692	-0.2992	0.0	0.2992	0.5692	0.7897	0.9433	0.9809	0.8321
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS											Without			
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Flanges
180.0	0.0	-3.203	-5.579	-7.544	-8.969	-9.839	-10.132	-9.839	-8.969	-7.544	-5.579	-3.203	0.0	-165.036



HOOP STRESS

FIGURE A5



AXIAL STRESS

FIGURE A6

TABLE A5

$$R/r = 2.0 \quad t/r = 0.05$$

heta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.3006	-0.5010	-0.3752	-0.2811	-0.2325	-0.2031	-0.1935	-0.2031	-0.2325	-0.2811	-0.3752	-0.5010	1.3006
22.5	1.1324	-0.4537	-0.3947	-0.3337	-0.3039	-0.2845	-0.2781	-0.2845	-0.3039	-0.3337	-0.3947	-0.4537	1.1324
45.0	0.6108	-0.2976	-0.3729	-0.4207	-0.4609	-0.4834	-0.4909	-0.4834	-0.4609	-0.4207	-0.3729	-0.2976	0.6108
67.5	-0.2195	-0.0414	-0.0039	-0.0904	-0.1658	-0.2146	-0.2319	-0.2146	-0.1658	-0.0904	-0.0039	-0.0414	-0.2195
90.0	-0.9245	-0.0109	0.4750	0.6499	0.7597	0.8261	0.8478	0.8261	0.7597	0.6499	0.4750	-0.0109	-0.9245
12.5	-0.7739	-0.4194	-0.0498	0.2165	0.4427	0.6014	0.6582	0.6014	0.4427	0.2165	-0.0498	-0.4194	-0.7739
35.0	-0.0101	-0.3992	-0.6893	-0.8256	-0.8671	-0.8693	-0.8663	-0.8693	-0.8671	-0.8256	-0.6893	-0.3992	-0.0101
57.5	0.5264	0.0419	-0.6041	-1.0815	-1.3853	-1.5558	-1.6115	-1.5558	-1.3853	-1.0815	-0.6041	0.0419	0.5264
80.0	0.7378	0.2087	-0.5839	-1.1951	-1.5882	-1.8075	-1.8787	-1.8075	-1.5882	-1.1951	-0.5839	0.2087	0.7378
													-1.0336

heta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	4.3353	0.8650	0.8662	0.6604	0.5359	0.4701	0.4491	0.4701	0.5359	0.6604	0.8662	0.8650	4.3353	-0.2856
22.5	3.7746	0.8463	0.8336	0.6370	0.5117	0.4432	0.4211	0.4432	0.5117	0.6370	0.8336	0.8463	3.7746	-0.3464
45.0	2.0359	0.7931	0.8189	0.6855	0.5858	0.5245	0.5034	0.5245	0.5858	0.6855	0.8189	0.7931	2.0359	1.0458
67.5	-0.7316	0.5311	0.8570	0.9572	1.0126	1.0344	1.0392	1.0344	1.0126	0.9572	0.8570	0.5311	-0.7316	4.8176
90.0	-3.0817	-0.3902	0.3329	0.7557	1.0755	1.2652	1.3268	1.2652	1.0755	0.7557	0.3329	-0.3902	-3.0817	2.8162
12.5	-2.5798	-1.3810	-0.9076	-0.5252	-0.1726	0.0648	0.1468	0.0648	-0.1726	-0.5252	-0.9076	-1.3810	-2.5798	-5.1786
35.0	-0.0338	-0.8938	-1.2250	-1.2918	-1.2468	-1.1891	-1.1657	-1.1891	-1.2468	-1.2918	-1.2250	-0.8938	-0.0338	-3.5269
57.5	1.7547	0.0504	-0.9146	-1.4270	-1.7342	-1.9187	-1.9825	-1.9187	-1.7342	-1.4270	-0.9146	0.0504	1.7547	-0.9343
80.0	2.4592	0.3003	-0.9441	-1.6083	-2.0123	-2.2629	-2.3518	-2.2629	-2.0123	-1.6083	-0.9441	0.3003	2.4592	-1.1432

heta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Withou t Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.4363	0.3650	0.2539	0.1658	0.1013	0.0482	0.0	-0.0482	-0.1013	-0.1658	-0.2539	-0.3650	-0.4363
45.0	0.6975	0.5879	0.4533	0.3104	0.1956	0.0948	0.0	-0.0948	-0.1956	-0.3104	-0.4533	-0.5879	-0.6975
67.5	0.5801	0.4840	0.4296	0.3166	0.2070	0.1025	0.0	-0.1025	-0.2070	-0.3166	-0.4296	-0.4840	-0.5801
90.0	0.0388	0.1045	0.1191	0.1048	0.0726	0.0365	0.0	-0.0365	-0.0726	-0.1048	-0.1191	-0.1045	-0.0388
12.5	-0.5645	-0.2674	-0.2082	-0.1374	-0.0812	-0.0380	0.0	0.0380	0.0812	0.1374	0.2082	0.2674	0.5645
35.0	-0.8511	-0.6185	-0.5070	-0.3800	-0.2449	-0.1182	0.0	0.1182	0.2449	0.3800	0.5070	0.6185	0.8511
57.5	-0.6993	-0.5863	-0.5402	-0.4701	-0.3477	-0.1840	0.0	0.1840	0.3477	0.4701	0.5402	0.5863	0.6993
80.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS

Without

TABLE A6

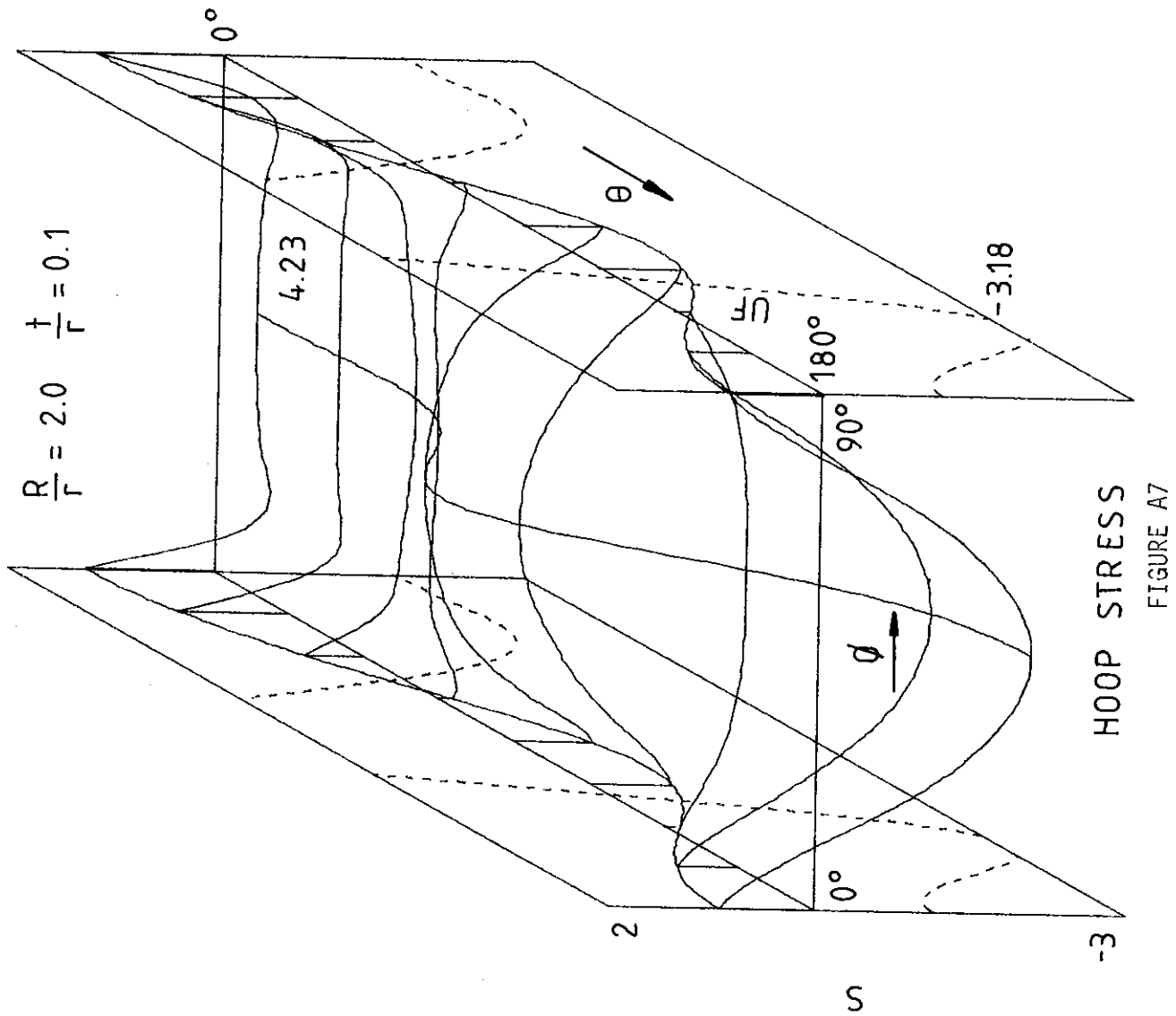
R/r = 2.0 t/r = 0.05

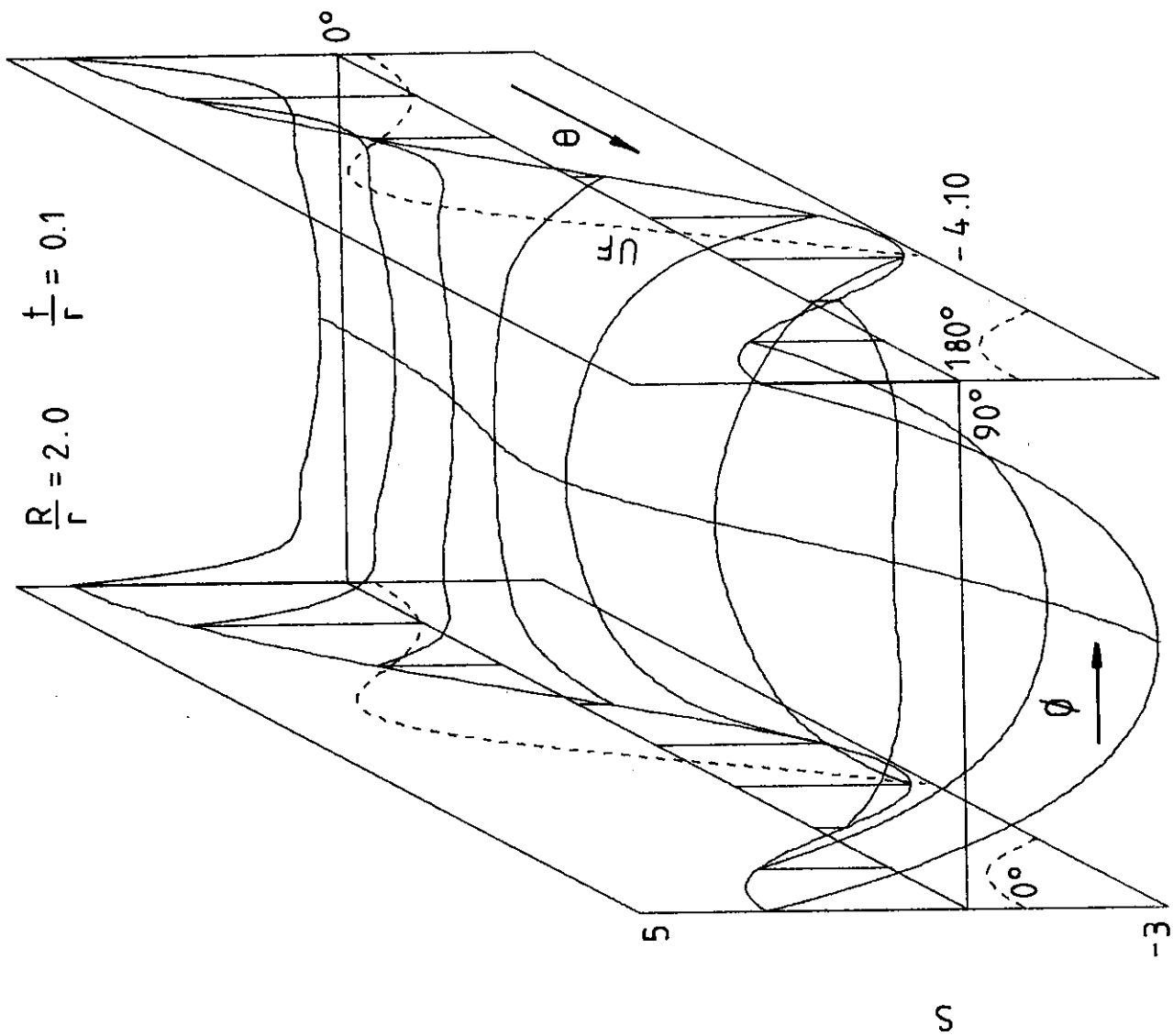
Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.3200	-0.1713	-0.2486	-0.1742	-0.1451	-0.1305	-0.1260	-0.1305	-0.1451	-0.1742	-0.2486	-0.1713	-0.3200	0.8654
22.5	-0.2586	-0.1295	-0.1725	-0.0896	-0.0520	-0.0324	-0.0263	-0.0324	-0.0520	-0.0896	-0.1725	-0.1295	-0.2586	1.7640
45.0	-0.0678	-0.0930	-0.0254	0.0999	0.1798	0.2283	0.2445	0.2283	0.1798	0.0999	-0.0254	-0.0930	-0.0678	2.8487
67.5	0.2129	-0.2780	-0.1976	-0.1067	-0.0289	0.0296	0.0511	0.0296	-0.0289	-0.1067	-0.1976	-0.2780	0.2129	-1.9492
90.0	0.3317	-0.5087	-0.6541	-0.8310	-0.9642	-1.0345	-1.0558	-1.0345	-0.9642	-0.8310	-0.6541	-0.5087	0.3317	-9.2749
112.5	0.0161	-0.2740	-0.3336	-0.5330	-0.7473	-0.8948	-0.9464	-0.8948	-0.7473	-0.5330	-0.3336	-0.2740	0.0161	-1.8869
135.0	-0.3827	-0.1227	-0.0023	0.0576	0.0927	0.1135	0.1204	0.1135	0.0927	0.0576	-0.0023	-0.1227	-0.3827	4.6040
157.5	-0.6403	-0.2136	-0.2901	-0.4053	-0.4442	-0.4356	-0.4265	-0.4356	-0.4442	-0.4053	-0.2901	-0.2136	-0.6403	1.5491
180.0	-0.8240	-0.2089	-0.3460	-0.6380	-0.8611	-0.9821	-1.0189	-0.9821	-0.8611	-0.6380	-0.3460	-0.2089	-0.8240	1.0297

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	-1.0668	1.5137	0.8686	0.6707	0.5450	0.4742	0.4517	0.4742	0.5450	0.6707	0.8686	1.5137	-1.0668
22.5	-0.8619	1.3964	0.8563	0.6815	0.5665	0.4996	0.4780	0.4996	0.5665	0.6815	0.8563	1.3964	-0.8619
45.0	-0.2259	1.0042	0.8068	0.7534	0.7035	0.6693	0.6576	0.6693	0.7035	0.7534	0.8068	1.0042	-0.2259
67.5	0.7096	0.2645	0.5001	0.7006	0.8015	0.8529	0.8692	0.8529	0.8015	0.7006	0.5001	0.2645	0.7096
90.0	1.1057	-0.5770	-0.3108	0.0142	0.2005	0.3034	0.3376	0.3034	0.2005	0.0142	-0.3108	-0.5770	1.1057
112.5	0.0537	-0.8149	-0.8051	-0.6704	-0.5949	-0.5569	-0.5444	-0.5569	-0.5949	-0.6704	-0.8051	-0.8149	0.0537
135.0	-1.2757	-0.6597	-0.5103	-0.5014	-0.5213	-0.5348	-0.5385	-0.5348	-0.5213	-0.5014	-0.5103	-0.6597	-1.2757
157.5	-2.1343	-0.9194	-0.4606	-0.4099	-0.4627	-0.5035	-0.5160	-0.5035	-0.4627	-0.4099	-0.4606	-0.9194	-2.1343
180.0	-2.7467	-1.1791	-0.5335	-0.4415	-0.5186	-0.5897	-0.6141	-0.5897	-0.5186	-0.4415	-0.5335	-1.1791	-2.7467

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.4150	0.4251	0.2538	0.1607	0.0947	0.0439	0.0	-0.0439	-0.0947	-0.1607	-0.2538	-0.4251	-0.4150
45.0	0.6635	0.8283	0.5454	0.3636	0.2237	0.1065	0.0	-0.1065	-0.2237	-0.3636	-0.5454	-0.8283	-0.6635
67.5	0.5518	0.9465	0.7577	0.5609	0.3719	0.1853	0.0	-0.1853	-0.3719	-0.5609	-0.7577	-0.9465	-0.5518
90.0	0.0369	0.3078	0.4356	0.4193	0.3208	0.1727	0.0	-0.1727	-0.3208	-0.4193	-0.4356	-0.3078	-0.0369
112.5	-0.5370	-0.7682	-0.5414	-0.3202	-0.1688	-0.0702	0.0	0.0702	0.1688	0.3202	0.5414	0.7682	0.5370
135.0	-0.8096	-1.1719	-1.1903	-1.0019	-0.7113	-0.3675	0.0	0.3675	0.7113	1.0019	1.1903	1.1719	0.8096
157.5	-0.6652	-0.8586	-0.9270	-0.8452	-0.6383	-0.3426	0.0	0.3426	0.6383	0.8452	0.9270	0.8586	0.6652
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without Flanges	
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-2.610	-4.843	-6.796	-8.327	-9.289	-9.618	-9.289	-8.327	-6.796	-4.843	-2.610	0.0





AXIAL STRESS

FIGURE A8

TABLE A3

$$R/r = 2.0 \quad t/r = 0.1$$

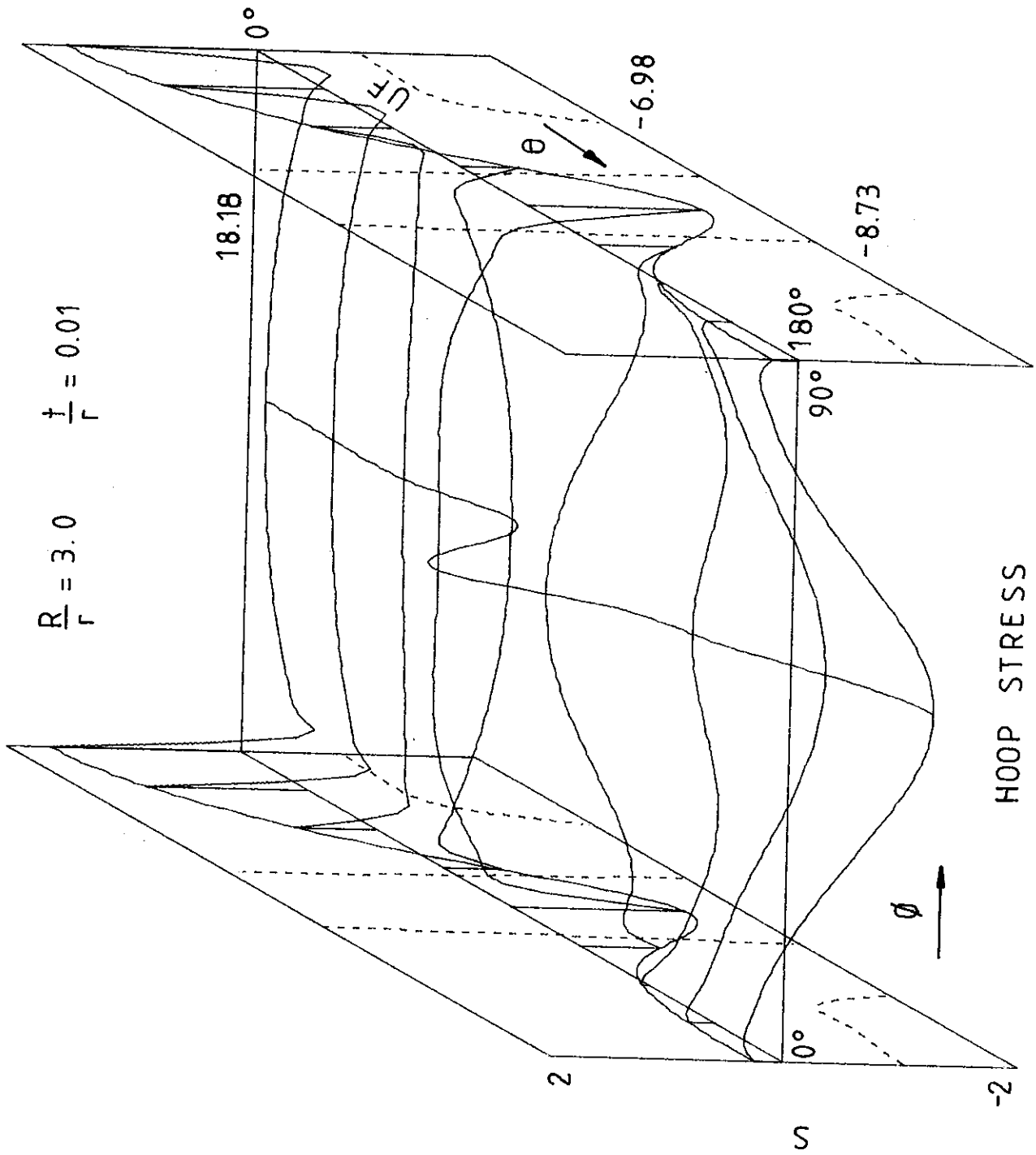
Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.3582	0.1462	-0.0466	0.0129	0.0660	0.0870	0.0927	0.0870	0.0660	0.0129	-0.0466	0.1462	-0.3582	1.7450
22.5	-0.2823	0.1088	-0.0221	0.0700	0.1466	0.1832	0.1944	0.1832	0.1466	0.0700	-0.0221	0.1088	-0.2823	1.9088
45.0	-0.0671	-0.0416	-0.0786	0.0587	0.1716	0.2353	0.2563	0.2353	0.1716	0.0587	-0.0786	-0.0416	-0.0671	1.1579
67.5	0.2170	-0.3055	-0.4108	-0.3624	-0.3116	-0.2796	-0.2683	-0.2796	-0.3116	-0.3624	-0.4108	-0.3055	0.2170	-2.3647
90.0	0.3805	-0.4513	-0.7502	-0.9472	-1.1047	-1.2058	-1.2400	-1.2058	-1.1047	-0.9472	-0.7502	-0.4513	0.3805	-5.8181
112.5	0.1935	-0.2701	-0.4854	-0.7300	-0.9769	-1.1533	-1.2163	-1.1533	-0.9769	-0.7300	-0.4854	-0.2701	0.1935	-2.8123
135.0	-0.3126	-0.1136	0.0055	0.0231	-0.0196	-0.0684	-0.0883	-0.0684	-0.0196	0.0231	0.0055	-0.1136	-0.3126	2.6379
157.5	-0.8360	-0.2603	-0.0462	0.0384	0.0855	0.1167	0.1283	0.1167	0.0855	0.0384	-0.0462	-0.2603	-0.8360	2.4216
180.0	-1.0778	-0.3710	-0.1887	-0.1859	-0.2076	-0.2154	-0.2157	-0.2154	-0.2076	-0.1859	-0.1887	-0.3710	-1.0778	1.1554

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-1.1940	1.4247	0.9434	0.6356	0.5124	0.4441	0.4206	0.4441	0.5124	0.6356	0.9434	1.4247	-1.1940	0.0954
22.5	-0.9411	1.2998	0.9287	0.6819	0.5857	0.5312	0.5121	0.5312	0.5857	0.6819	0.9287	1.2998	-0.9411	0.5047
45.0	-0.2237	0.9302	0.8070	0.7383	0.7303	0.7222	0.7178	0.7222	0.7303	0.7383	0.8070	0.9302	-0.2237	1.2400
67.5	0.7235	0.3838	0.3997	0.5355	0.6456	0.7012	0.7173	0.7012	0.6456	0.5355	0.3997	0.3838	0.7235	0.8195
90.0	1.2683	-0.1178	-0.2612	-0.1003	0.0385	0.1103	0.1313	0.1103	0.0385	-0.1003	-0.2612	-0.1178	1.2683	-1.3134
112.5	0.6448	-0.3773	-0.6533	-0.6499	-0.6174	-0.6060	-0.6049	-0.6060	-0.6174	-0.6499	-0.6533	-0.3773	0.6448	-2.0488
135.0	-1.0421	-0.6761	-0.5465	-0.5101	-0.5216	-0.5468	-0.5586	-0.5468	-0.5216	-0.5101	-0.5465	-0.6761	-1.0421	0.2982
157.5	-2.7867	-1.3546	-0.5600	-0.1965	-0.0605	-0.0218	-0.0152	-0.0218	-0.0605	-0.1965	-0.5600	-1.3546	-2.7867	1.6537
180.0	-3.5926	-1.7978	-0.7011	-0.1399	0.1093	0.2051	0.2291	0.2051	0.1093	-0.1399	-0.7011	-1.7978	-3.5926	1.4772

25

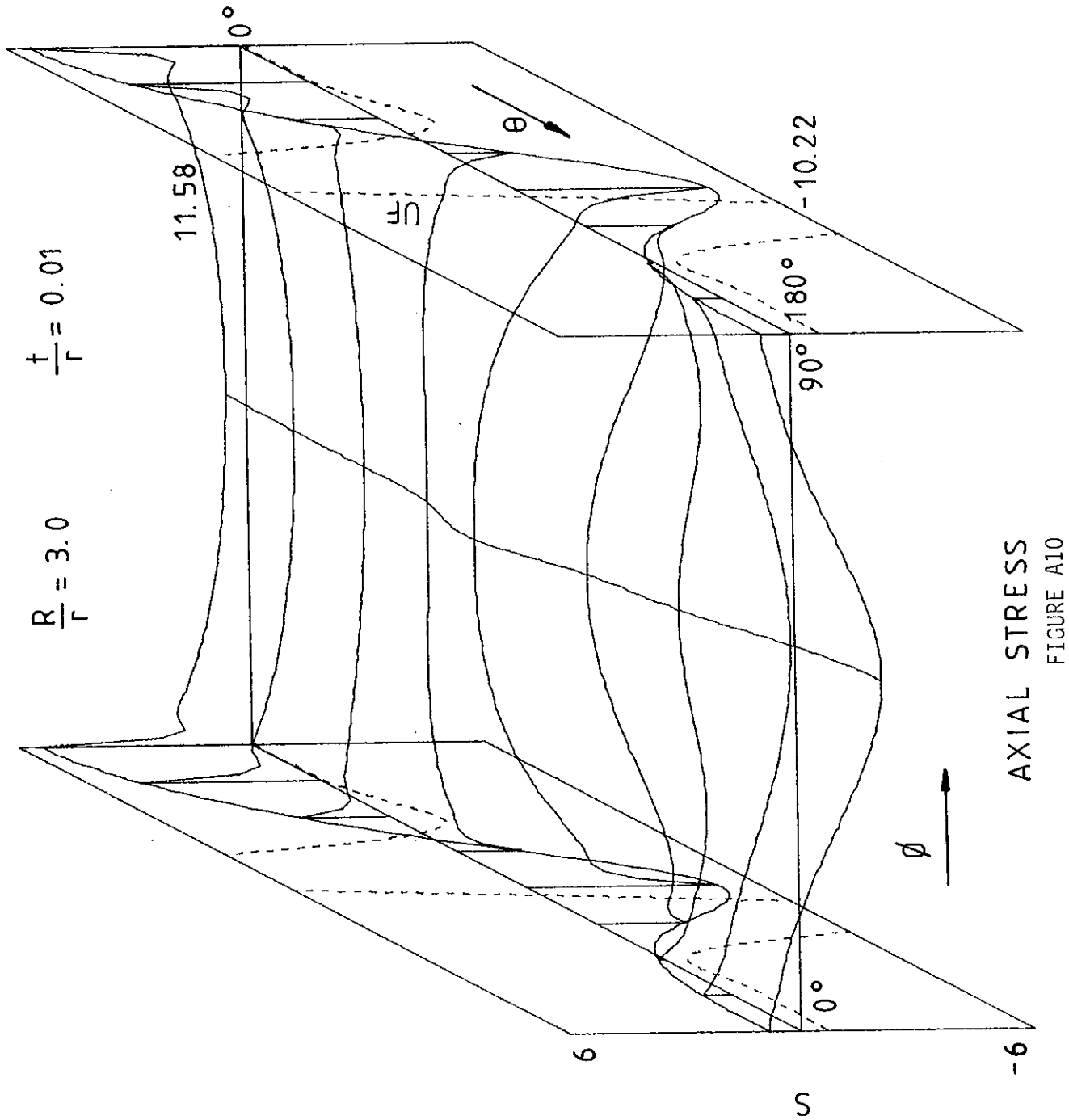
Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.2959	0.4579	0.2925	0.1831	0.1103	0.0521	0.0	-0.0521	-0.1103	-0.1831	-0.2925	-0.4579	-0.2959
45.0	0.4583	0.8128	0.5874	0.3944	0.2472	0.1195	0.0	-0.1195	-0.2472	-0.3944	-0.5874	-0.8128	-0.4583
67.5	0.3785	0.8377	0.7280	0.5450	0.3628	0.1813	0.0	-0.1813	-0.3628	-0.5450	-0.7280	-0.8377	-0.3785
90.0	0.0507	0.3012	0.3911	0.3670	0.2756	0.1464	0.0	-0.1464	-0.2756	-0.3670	-0.3911	-0.3012	-0.0507
112.5	-0.3541	-0.5782	-0.4659	-0.3008	-0.1695	-0.0749	0.0	0.0749	0.1695	0.3008	0.4659	0.5782	0.3541
135.0	-0.5865	-1.0648	-1.1376	-0.9691	-0.6862	-0.3528	0.0	0.3528	0.6862	0.9691	1.1376	1.0648	0.5865
157.5	-0.4809	-0.7902	-0.9186	-0.8588	-0.6521	-0.3495	0.0	0.3495	0.6521	0.8588	0.9186	0.7902	0.4809
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without	
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-1.893	-3.994	-5.698	-7.059	-7.934	-8.233	-7.934	-7.059	-5.698	-3.994	-1.893	0.0
													-33.606



HOOP STRESS

FIGURE A9



AXIAL STRESS
FIGURE A10

TABLE A9

$$R/r = 3.0 \quad t/r = 0.01$$

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS												Without Flanges
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	1.6240	-0.4216	-0.3043	-0.2293	-0.1837	-0.1591	-0.1513	-0.1591	-0.1837	-0.2293	-0.3043	-0.4216	1.6240	
22.5	1.3869	-0.3659	-0.2736	-0.2104	-0.1706	-0.1486	-0.1415	-0.1486	-0.1706	-0.2104	-0.2736	-0.3659	1.3869	
45.0	0.6880	-0.2366	-0.2271	-0.2075	-0.1890	-0.1767	-0.1724	-0.1767	-0.1890	-0.2075	-0.2271	-0.2366	0.6880	
67.5	-0.5023	-0.0714	-0.2439	-0.3762	-0.4591	-0.5013	-0.5138	-0.5013	-0.4591	-0.3762	-0.2439	-0.0714	-0.5023	
90.0	-1.4875	0.3224	0.5737	0.6915	0.7130	0.6978	0.6879	0.6978	0.7130	0.6915	0.5737	0.3224	-1.4875	
112.5	-0.6931	-0.4382	-0.3396	-0.0938	0.1472	0.3044	0.3572	0.3044	0.1472	-0.0938	-0.3396	-0.4382	-0.6931	
135.0	0.0488	-0.3152	-0.5045	-0.5755	-0.4993	-0.3740	-0.3162	-0.3740	-0.4993	-0.5755	-0.5045	-0.3152	0.0488	
157.5	0.2252	0.0387	-0.1531	-0.3986	-0.6450	-0.8186	-0.8791	-0.8186	-0.6450	-0.3986	-0.1531	0.0387	-0.2252	
180.0	0.2402	0.2294	-0.0156	-0.3630	-0.7569	-1.0875	-1.2192	-1.0875	-0.7569	-0.3630	-0.0156	0.2294	-0.2402	

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS												Without Flanges
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	5.4134	1.6012	1.1010	0.7996	0.6205	0.5253	0.4954	0.5253	0.6205	0.7996	1.1010	1.6012	5.4134	-0.0455
22.5	4.6230	1.4797	1.0585	0.7913	0.6268	0.5374	0.5090	0.5374	0.6268	0.7913	1.0585	1.4797	4.6230	-0.0773
45.0	2.2935	1.0469	0.8554	0.6900	0.5699	0.4984	0.4749	0.4984	0.5699	0.6900	0.8554	1.0469	2.2935	-1.1844
67.5	-1.6743	0.2606	0.5828	0.6681	0.6630	0.6391	0.6282	0.6391	0.6630	0.6681	0.5828	0.2606	-1.6743	5.8509
90.0	-4.9585	-1.0948	-0.0098	0.6531	1.0101	1.1724	1.2174	1.1724	1.0101	0.6531	-0.0098	-1.0948	-4.9585	6.0826
112.5	-2.3102	-2.0267	-1.5836	-0.9543	-0.3930	-0.0274	0.0975	-0.0274	-0.3930	-0.9543	-1.5836	-2.0267	-2.3102	-9.7407
135.0	0.1628	-0.8232	-1.1296	-1.1279	-0.8907	-0.6250	-0.5132	-0.6250	-0.8907	-1.1279	-1.1296	-0.8232	0.1628	-0.6012
157.5	0.7508	0.0826	-0.3480	-0.8408	-1.2733	-1.5489	-1.6399	-1.5489	-1.2733	-0.8408	-0.3480	0.0826	0.7508	-0.7955
180.0	0.8005	0.4503	-0.0131	-0.6599	-1.3829	-1.9801	-2.2151	-1.9801	-1.3829	-0.6599	-0.0131	0.4503	0.8005	-0.7180

Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS												Without Flanges
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
22.5	0.6782	0.4545	0.2813	0.1735	0.1008	0.0464	0.0	-0.0464	-0.1008	-0.1735	-0.2813	-0.4545	-0.6782	
45.0	1.0781	0.8061	0.5336	0.3445	0.2061	0.0965	0.0	-0.0965	-0.2061	-0.3445	-0.5336	-0.8061	-1.0781	
67.5	0.8056	0.7804	0.6085	0.4427	0.2869	0.1407	0.0	-0.1407	-0.2869	-0.4427	-0.6085	-0.7804	-0.8056	
90.0	-0.3366	0.0391	0.1983	0.2389	0.2013	0.1132	0.0	-0.1132	-0.2013	-0.2389	-0.1983	-0.0391	0.3366	
112.5	-0.9904	-0.7006	-0.3900	-0.1909	-0.0794	-0.0253	0.0	0.0253	0.0794	0.1909	0.3900	0.7006	0.9904	
135.0	-0.7650	-0.9093	-0.8696	-0.7043	-0.4830	-0.2438	0.0	0.2438	0.4830	0.7043	0.8696	0.9093	0.7650	
157.5	-0.3648	-0.5295	-0.6793	-0.7181	-0.5998	-0.3400	0.0	0.3400	0.5998	0.7181	0.6793	0.5295	0.3648	
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	

		DIAMETER EXPANSION FACTORS												Without Flanges
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
90.0	0.0	9.761	18.744	25.796	30.369	32.792	33.533	32.792	30.369	25.796	18.744	9.761	0.0	656.591

TABLE A10

$$R/r = 3.0 \quad t/r = 0.01$$

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.1687	-0.3740	-0.2479	-0.1724	-0.1285	-0.1054	-0.0983	-0.1054	-0.1285	-0.1724	-0.2479	-0.3740	-0.1687	0.8530
22.5	-0.1194	-0.3247	-0.2249	-0.1625	-0.1249	-0.1048	-0.0984	-0.1048	-0.1249	-0.1625	-0.2249	-0.3247	-0.1194	0.8137
45.0	0.0035	-0.1605	-0.1086	-0.0710	-0.0467	-0.0333	-0.0289	-0.0333	-0.0467	-0.0710	-0.1086	-0.1605	0.0035	3.3623
67.5	0.1490	0.0359	0.1443	0.2591	0.3413	0.3868	0.4010	0.3868	0.3413	0.2591	0.1443	0.0359	0.1490	4.0990
90.0	0.0226	-0.3539	-0.6432	-0.7658	-0.7835	-0.7638	-0.7521	-0.7638	-0.7835	-0.7658	-0.6432	-0.3539	0.0226	-20.2240
112.5	-0.4118	-0.0444	-0.0273	-0.1309	-0.2464	-0.3188	-0.3418	-0.3188	-0.2464	-0.1309	-0.0273	-0.0444	-0.4118	5.8506
135.0	-0.2575	-0.1842	-0.1128	-0.0043	0.0409	0.0327	0.0219	0.0327	0.0409	-0.0043	-0.1128	-0.1842	-0.2575	2.9345
157.5	0.0195	0.0110	-0.1843	-0.3276	-0.3881	-0.4002	-0.4006	-0.4002	-0.3881	-0.3276	-0.1843	0.0110	0.0195	1.1489
180.0	0.1603	0.1505	-0.0573	-0.2925	-0.5185	-0.6727	-0.7251	-0.6727	-0.5185	-0.2925	-0.0573	0.1505	0.1603	1.0445

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.5624	1.6000	1.1078	0.8073	0.6282	0.5327	0.5028	0.5327	0.6282	0.8073	1.1078	1.6000	-0.5624	0.0532
22.5	-0.3980	1.4707	1.0598	0.7943	0.6302	0.5407	0.5123	0.5407	0.6302	0.7943	1.0598	1.4707	-0.3980	-0.0414
45.0	0.0116	1.0313	0.8677	0.7139	0.5984	0.5287	0.5055	0.5287	0.5984	0.7139	0.8677	1.0313	0.0116	0.1606
67.5	0.4966	0.2163	0.6171	0.7812	0.8297	0.8346	0.8326	0.8346	0.8297	0.7812	0.6171	0.2163	0.4966	6.5881
90.0	0.0754	-1.2694	-0.4599	0.0892	0.4235	0.5976	0.6509	0.5976	0.4235	0.0892	-0.4599	-1.2694	0.0754	-5.4593
112.5	-1.3725	-1.6597	-1.2927	-0.8580	-0.4837	-0.2357	-0.1493	-0.2357	-0.4837	-0.8580	-1.2927	-1.6597	-1.3725	-3.6899
135.0	-0.8584	-0.7231	-0.8598	-0.7794	-0.6139	-0.4710	-0.4161	-0.4710	-0.6139	-0.7794	-0.8598	-0.7231	-0.8584	2.1455
157.5	0.0651	0.0274	-0.3939	-0.7712	-1.0466	-1.2056	-1.2569	-1.2056	-1.0466	-0.7712	-0.3939	0.0274	0.0651	0.7554
180.0	0.5345	0.3110	-0.1243	-0.6469	-1.1666	-1.5494	-1.6896	-1.5494	-1.1666	-0.6469	-0.1243	0.3110	0.5345	0.7349

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.6715	0.4646	0.2843	0.1738	0.1003	0.0461	0.0	-0.0461	-0.1003	-0.1738	-0.2843	-0.4646	-0.6715
45.0	1.0673	0.8289	0.5398	0.3444	0.2045	0.0954	0.0	-0.0954	-0.2045	-0.3444	-0.5398	-0.8289	-1.0673
67.5	0.7976	0.9232	0.7122	0.5043	0.3182	0.1531	0.0	-0.1531	-0.3182	-0.5043	-0.7122	-0.9232	-0.7976
90.0	-0.3332	0.1343	0.3807	0.4181	0.3307	0.1785	0.0	-0.1785	-0.3307	-0.4181	-0.3807	-0.1343	0.3332
112.5	-0.9805	-0.9675	-0.5840	-0.2507	-0.0521	0.0132	0.0	-0.0132	0.0521	0.2507	0.5840	0.9675	0.9805
135.0	-0.7574	-0.9396	-1.0162	-0.9173	-0.6702	-0.3468	0.0	0.3468	0.6702	0.9173	1.0162	0.9396	0.7574
157.5	-0.3611	-0.5338	-0.7323	-0.8396	-0.7602	-0.4566	0.0	0.4566	0.7602	0.8396	0.7323	0.5338	0.3611
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without Flanges	
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-4.378	-8.519	-13.341	-18.131	-21.775	-23.152	-21.775	-18.131	-13.341	-8.519	-4.378	0.0
													-330.412

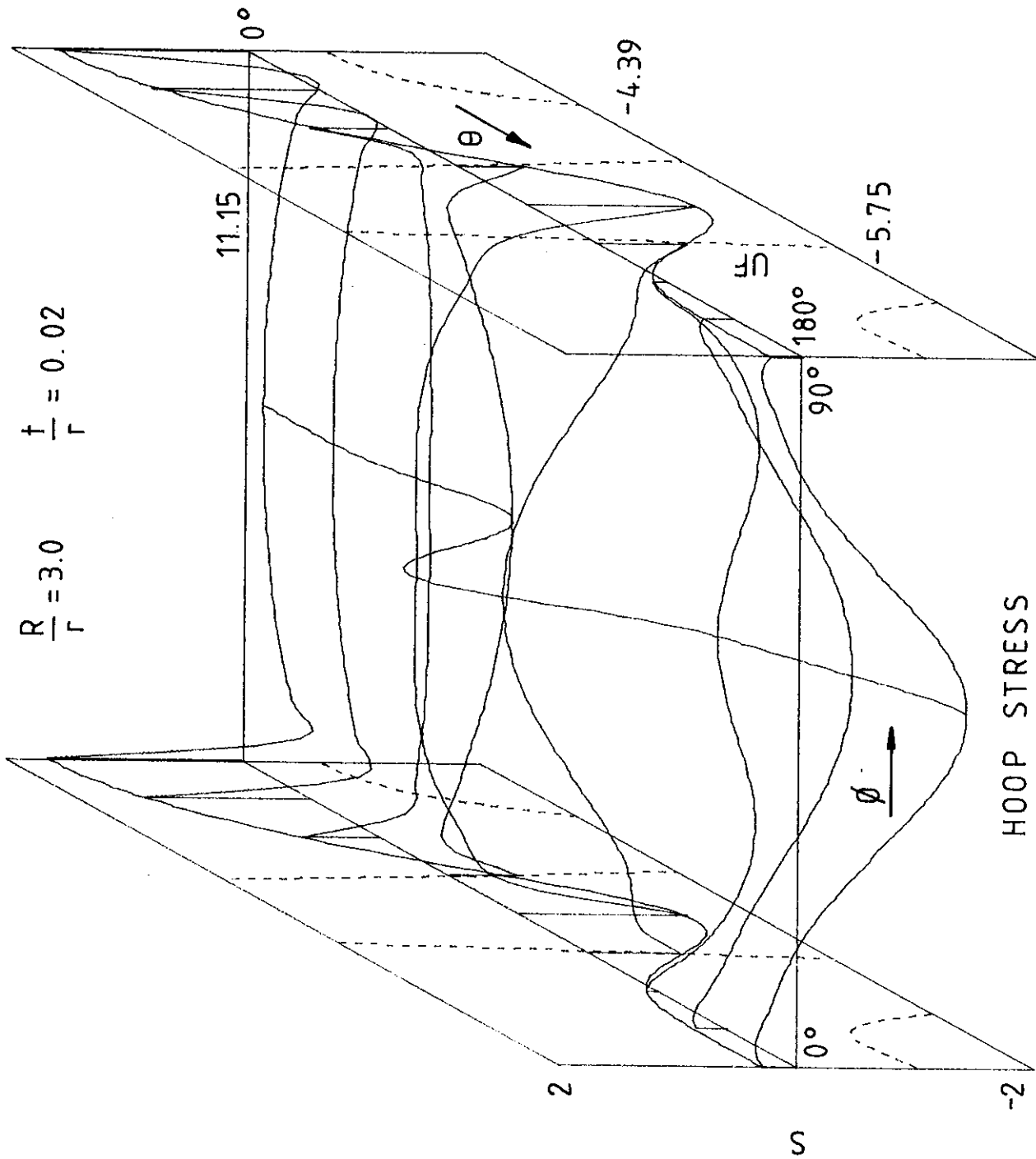


FIGURE A11

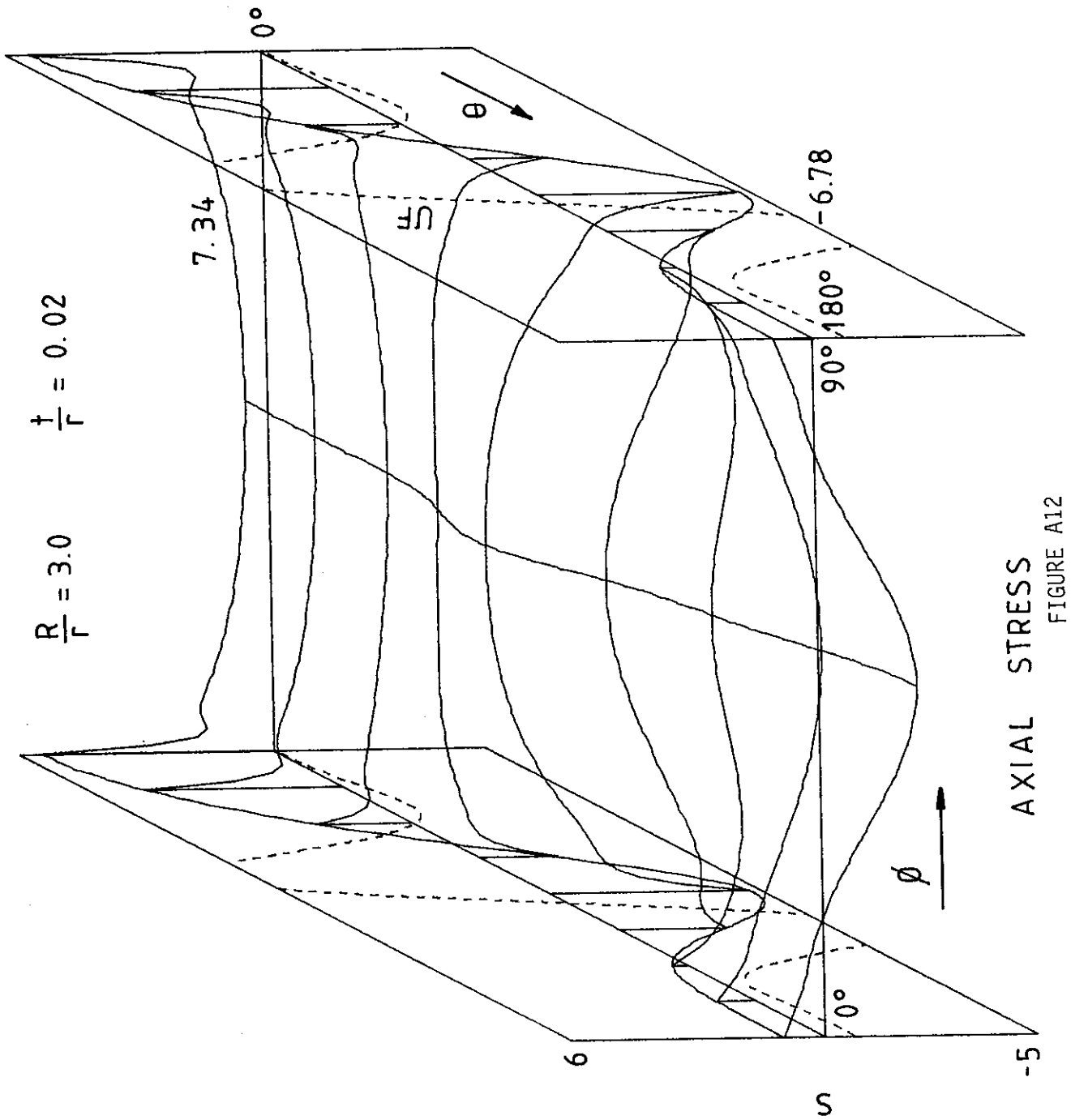


TABLE A11

$$R/r = 3.0 \quad t/r = 0.02$$

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.6363	-0.4383	-0.3103	-0.2336	-0.1866	-0.1611	-0.1531	-0.1611	-0.1866	-0.2336	-0.3103	-0.4383	1.6363
22.5	1.3970	-0.3928	-0.2928	-0.2282	-0.1857	-0.1616	-0.1538	-0.1616	-0.1857	-0.2282	-0.2928	-0.3928	1.3970
45.0	0.6595	-0.2989	-0.3530	-0.3826	-0.3925	-0.3934	-0.3929	-0.3934	-0.3925	-0.3826	-0.3530	-0.2989	0.6595
67.5	-0.5559	0.0360	-0.1338	-0.2852	-0.4018	-0.4719	-0.4948	-0.4719	-0.4018	-0.2852	-0.1338	0.0360	-0.5559
90.0	-1.4250	0.4035	0.6774	0.8488	0.9043	0.9045	0.8985	0.9045	0.9043	0.8488	0.6774	0.4035	-1.4250
112.5	-0.7719	-0.3530	-0.1702	0.1653	0.4777	0.6837	0.7540	0.6837	0.4777	0.1653	-0.1702	-0.3530	-0.7719
135.0	0.1010	-0.4634	-0.7426	-0.8095	-0.7004	-0.5440	-0.4738	-0.5440	-0.7004	-0.8095	-0.7426	-0.4634	0.1010
157.5	0.2715	0.0620	-0.1882	-0.4949	-0.7794	-0.9679	-1.0323	-0.9679	-0.7794	-0.4949	-0.1882	0.0620	0.2715
180.0	0.2815	0.2348	0.0072	-0.3599	-0.8317	-1.2416	-1.4050	-1.2416	-0.8317	-0.3599	0.0072	0.2348	0.2815
													-1.0278

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS									
		7.5	15.0	22.5	30.0	37.5	45.0	52.5			
0.0	5.4542	1.5736	1.0676	0.7742	0.6000	0.5074	0.4783	0.5074			
22.5	4.6565	1.4406	1.0037	0.7365	0.5730	0.4846	0.4567	0.4846			
45.0	2.1984	1.0296	0.8146	0.6305	0.4961	0.4165	0.3903	0.4165			
67.5	-1.8530	0.3762	0.7409	0.8651	0.8766	0.8565	0.8457	0.8565			
90.0	-4.7499	-0.9161	0.0676	0.7333	1.1142	1.2991	1.3528	1.2991			
112.5	-2.5730	-2.0001	-1.5436	-0.9126	-0.3532	0.0170	0.1452	0.0170			
135.0	0.3367	-1.0043	-1.3309	-1.3262	-1.0925	-0.8358	-0.7283	-0.8358			
157.5	0.9050	0.1013	-0.3551	-0.8854	-1.3275	-1.6058	-1.6993	-1.6058			
180.0	0.9382	0.4303	0.0138	-0.6489	-1.4320	-2.0756	-2.3248	-2.0756			

22

Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.6251	0.4469	0.2779	0.1720	0.1000	0.0461	0.0	-0.0461	-0.1000	-0.1720	-0.2779	-0.4469	-0.6251
45.0	0.9733	0.7788	0.5191	0.3395	0.2050	0.0965	0.0	-0.0965	-0.2050	-0.3395	-0.5191	-0.7788	-0.9733
67.5	0.6707	0.6741	0.5241	0.3889	0.2579	0.1286	0.0	-0.1286	-0.2579	-0.3889	-0.5241	-0.6741	-0.6707
90.0	-0.2859	0.0097	0.1248	0.1556	0.1341	0.0768	0.0	-0.0768	-0.1341	-0.1556	-0.1248	-0.0097	0.2859
112.5	-0.8915	-0.5913	-0.3455	-0.2026	-0.1158	-0.0537	0.0	0.0537	0.1158	0.2026	0.3455	0.5913	0.8915
135.0	-0.7333	-0.8160	-0.7162	-0.5560	-0.3812	-0.1957	0.0	0.1957	0.3812	0.5560	0.7162	0.8160	0.7333
157.5	-0.3663	-0.5238	-0.6190	-0.5837	-0.4355	-0.2274	0.0	0.2274	0.4355	0.5837	0.6190	0.5238	0.3663
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without Flanges	
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
90.0	0.0	8.089	15.634	21.879	26.189	28.614	29.386	28.614	26.189	21.879	15.634	8.089	0.0
													287.685

TABLE A12

$$R/r = 3.0 \quad t/r = 0.02$$
$$R/r = 3.0 \quad t/r = 0.02$$

INSIDE HOOP STRESS FACTORS

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										90.0	Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0			82.5
0.0	-0.2263	-0.3673	-0.2246	-0.1539	-0.1133	-0.0921	-0.0854	-0.0921	-0.1133	-0.1539	-0.2246	-0.3673	-0.2263	0.6525
22.5	-0.1690	-0.3067	-0.1883	-0.1300	-0.0968	-0.0795	-0.0742	-0.0795	-0.0968	-0.1300	-0.1883	-0.3067	-0.1690	1.1964
45.0	-0.0034	-0.0924	0.0386	0.1255	0.1776	0.2038	0.2116	0.2038	0.1776	0.1255	0.0386	-0.0924	-0.0034	3.9969
67.5	0.2189	-0.0639	0.0251	0.1651	0.2851	0.3606	0.3859	0.3606	0.2851	0.1651	0.0251	-0.0639	0.2189	-0.3298
90.0	0.1387	-0.4283	-0.7934	-0.9736	-1.0222	-1.0128	-1.0027	-1.0128	-1.0222	-0.9736	-0.7934	-0.4283	0.1387	-12.7638
112.5	-0.3472	-0.0672	-0.1928	-0.4283	-0.6484	-0.7872	-0.8327	-0.7872	-0.6484	-0.4283	-0.1928	-0.0672	-0.3472	0.3152
135.0	-0.3509	-0.0064	0.1685	0.2782	0.2506	0.1652	0.1224	0.1652	0.2506	0.2782	0.1685	-0.0064	-0.3509	4.6201
157.5	-0.0466	-0.0271	-0.1892	-0.2188	-0.1982	-0.1776	-0.1712	-0.1776	-0.1982	-0.2188	-0.1892	-0.0271	-0.0466	1.0419
180.0	0.0752	0.0996	-0.1471	-0.3332	-0.4008	-0.3745	-0.3490	-0.3745	-0.4008	-0.3332	-0.1471	0.0996	0.0752	1.0267

INSIDE AXIAL STRESS FACTORS

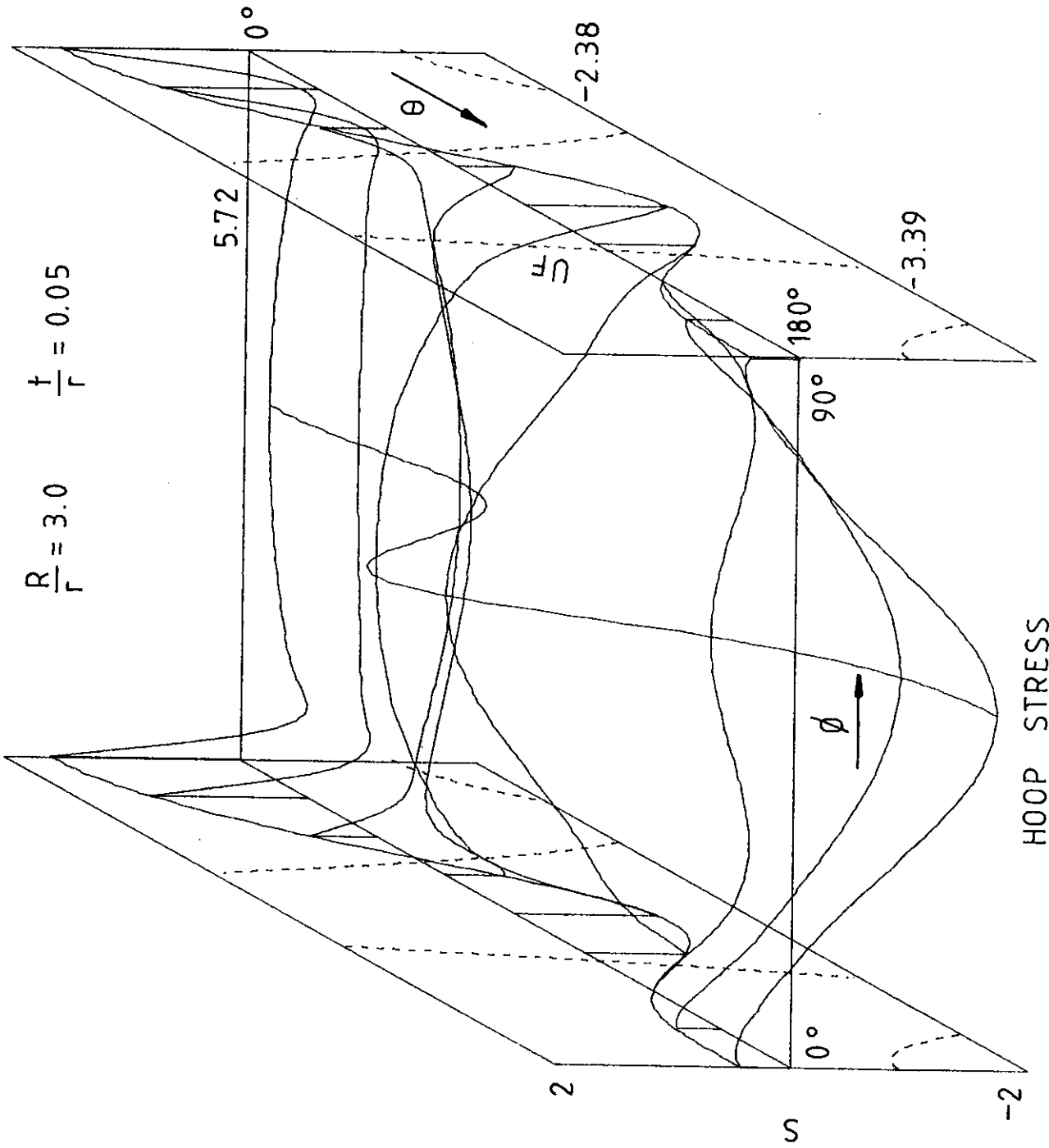
Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS											Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5		90.0
0.0	-0.7544	1.5370	1.0742	0.7802	0.6050	0.5116	0.4823	0.5116	0.6050	0.7802	1.0742	1.5370	-0.7544	-0.0547
22.5	-0.5632	1.4066	1.0116	0.7462	0.5817	0.4920	0.4636	0.4920	0.5817	0.7462	1.0116	1.4066	-0.5632	-0.1669
45.0	-0.0114	1.0053	0.8795	0.7416	0.6315	0.5626	0.5394	0.5626	0.6315	0.7416	0.8795	1.0053	-0.0114	1.4053
67.5	0.7296	0.1747	0.6320	0.8449	0.9334	0.9626	0.9684	0.9626	0.9334	0.8449	0.6320	0.1747	0.7296	4.0745
90.0	0.4623	-1.2201	-0.4824	0.0050	0.3260	0.5067	0.5648	0.5067	0.3260	0.0050	-0.4824	-1.2201	0.4623	-3.3500
112.5	-1.1572	-1.5782	-1.2635	-0.9468	-0.6730	-0.4879	-0.4223	-0.4879	-0.6730	-0.9468	-1.2635	-1.5782	-1.1572	-4.2242
135.0	-1.1697	-0.6527	-0.7378	-0.6812	-0.6039	-0.5502	-0.5319	-0.5502	-0.6039	-0.6812	-0.7378	-0.6527	-1.1697	1.8401
157.5	-0.1555	-0.0286	-0.3947	-0.6668	-0.8602	-0.9786	-1.0191	-0.9786	-0.8602	-0.6668	-0.3947	-0.0286	-0.1555	0.9351
180.0	0.2508	0.1905	-0.2396	-0.6738	-1.0175	-1.2220	-1.2874	-1.2220	-1.0175	-0.6738	-0.2396	0.1905	0.2508	0.7001

INSIDE SHEAR STRESS FACTORS

[illegible]

DIAMETER EXPANSION FACTORS

DIAMETER EXPANSION FACTORS													
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-4.215	-8.113	-12.719	-17.347	-20.857	-22.174	-20.857	-17.347	-12.719	-8.113	-4.215	0.0
													Without Flanges -165.306



HOOP STRESS

FIGURE A13

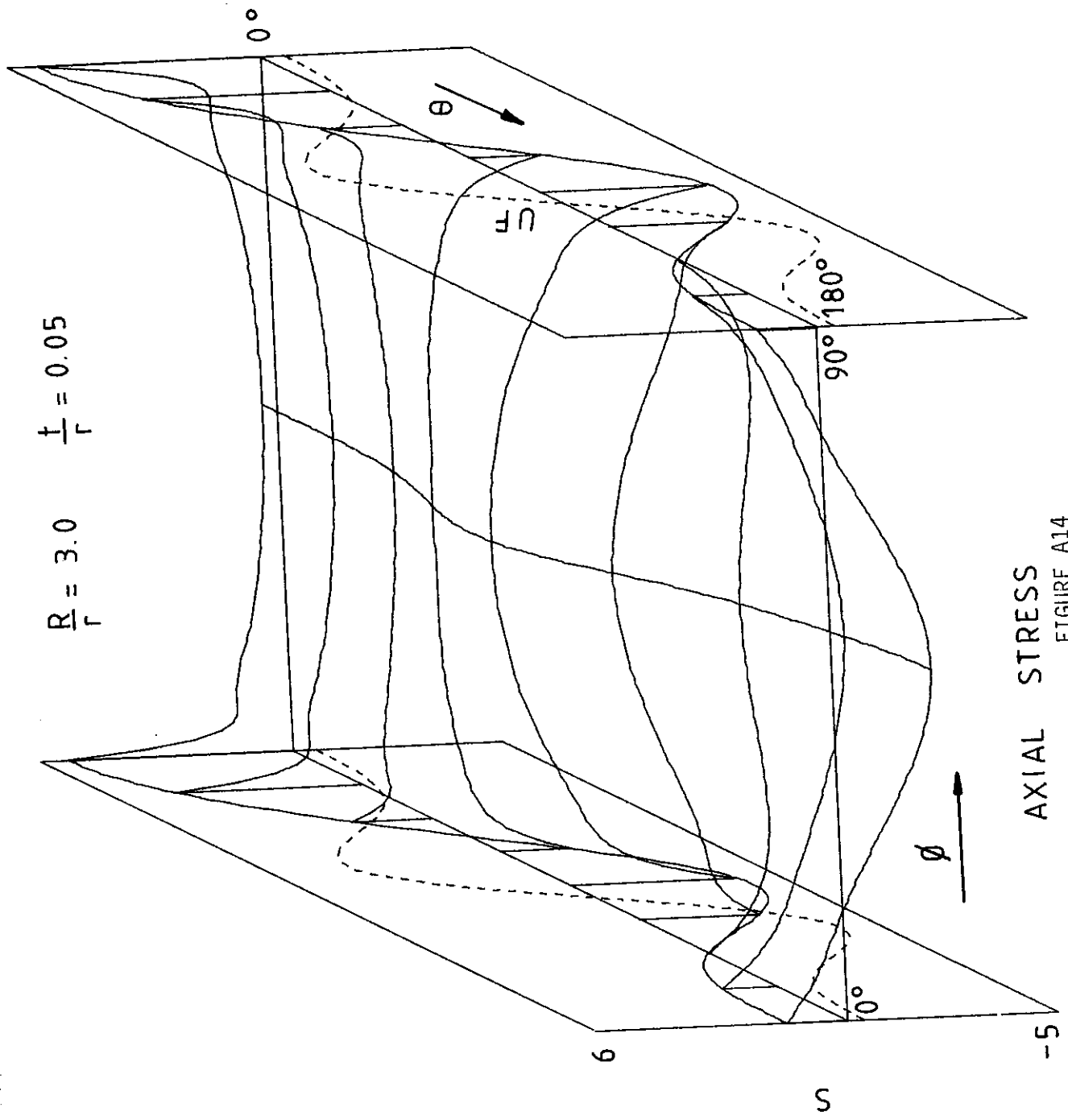


TABLE A13

R/r = 3.0 t/r = 0.05

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.5953	-0.5668	-0.3804	-0.3083	-0.2550	-0.2231	-0.2124	-0.2231	-0.2550	-0.3083	-0.3804	-0.5668	1.5953
22.5	1.3331	-0.5215	-0.4354	-0.4200	-0.4030	-0.3906	-0.3860	-0.3906	-0.4030	-0.4200	-0.4354	-0.5215	1.3331
45.0	0.5607	-0.3000	-0.4046	-0.5216	-0.6050	-0.6545	-0.6707	-0.6545	-0.6050	-0.5216	-0.4046	-0.3000	0.5607
67.5	-0.5132	0.1532	0.1390	0.0405	-0.0642	-0.1438	-0.1731	-0.1438	-0.0642	0.0405	0.1390	0.1532	-0.5132
90.0	-1.2077	0.3251	0.7317	0.9731	1.1221	1.1928	1.2128	1.1928	1.1221	0.9731	0.7317	0.3251	-1.2077
112.5	-0.8608	-0.2274	0.0938	0.4822	0.8569	1.1173	1.2102	1.1173	0.8569	0.4822	0.0938	-0.2274	-0.8608
135.0	-0.0022	-0.4952	-0.7757	-0.7758	-0.6464	-0.5079	-0.4502	-0.5079	-0.6464	-0.7758	-0.7757	-0.4952	-0.0022
157.5	0.3920	-0.0163	-0.4812	-0.8564	-1.1786	-1.3973	-1.4741	-1.3973	-1.1786	-0.8564	-0.4812	-0.0163	0.3920
180.0	0.4213	0.2778	-0.1270	-0.6093	-1.1368	-1.5474	-1.7013	-1.5474	-1.1368	-0.6093	-0.1270	0.2778	0.4213
													-0.9225

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	5.3178	1.2303	0.9064	0.6053	0.4340	0.3441	0.3161	0.3441	0.4340	0.6053	0.9064	1.2303	5.3178
22.5	4.4435	1.1695	0.8783	0.5932	0.4196	0.3244	0.2942	0.3244	0.4196	0.5932	0.8783	1.1695	4.4435
45.0	1.8689	1.0025	0.9038	0.7412	0.6093	0.5238	0.4944	0.5238	0.6093	0.7412	0.9038	1.0025	1.8689
67.5	-1.7105	0.5491	0.9188	1.1190	1.1949	1.2138	1.2157	1.2138	1.1949	1.1190	0.9188	0.5491	-1.7105
90.0	-4.0258	-0.5730	0.1694	0.7876	1.1957	1.4246	1.4985	1.4246	1.1957	0.7876	0.1694	-0.5730	-4.0258
112.5	-2.8692	-1.6759	-1.2764	-0.6996	-0.1918	0.1494	0.2697	0.1494	-0.1918	-0.6996	-1.2764	-1.6759	-2.8692
135.0	-0.0075	-1.2645	-1.5744	-1.5211	-1.3512	-1.1919	-1.1287	-1.1919	-1.3512	-1.5211	-1.5744	-1.6759	-3.1741
157.5	1.3067	-0.0119	-0.6255	-1.1314	-1.5645	-1.8488	-1.9471	-1.8488	-1.5645	-1.1314	-0.6255	-0.0119	-1.3067
180.0	1.4042	0.5172	-0.0921	-0.8260	-1.5927	-2.1526	-2.3556	-2.1526	-1.5927	-0.8260	-0.0921	0.5172	1.4042

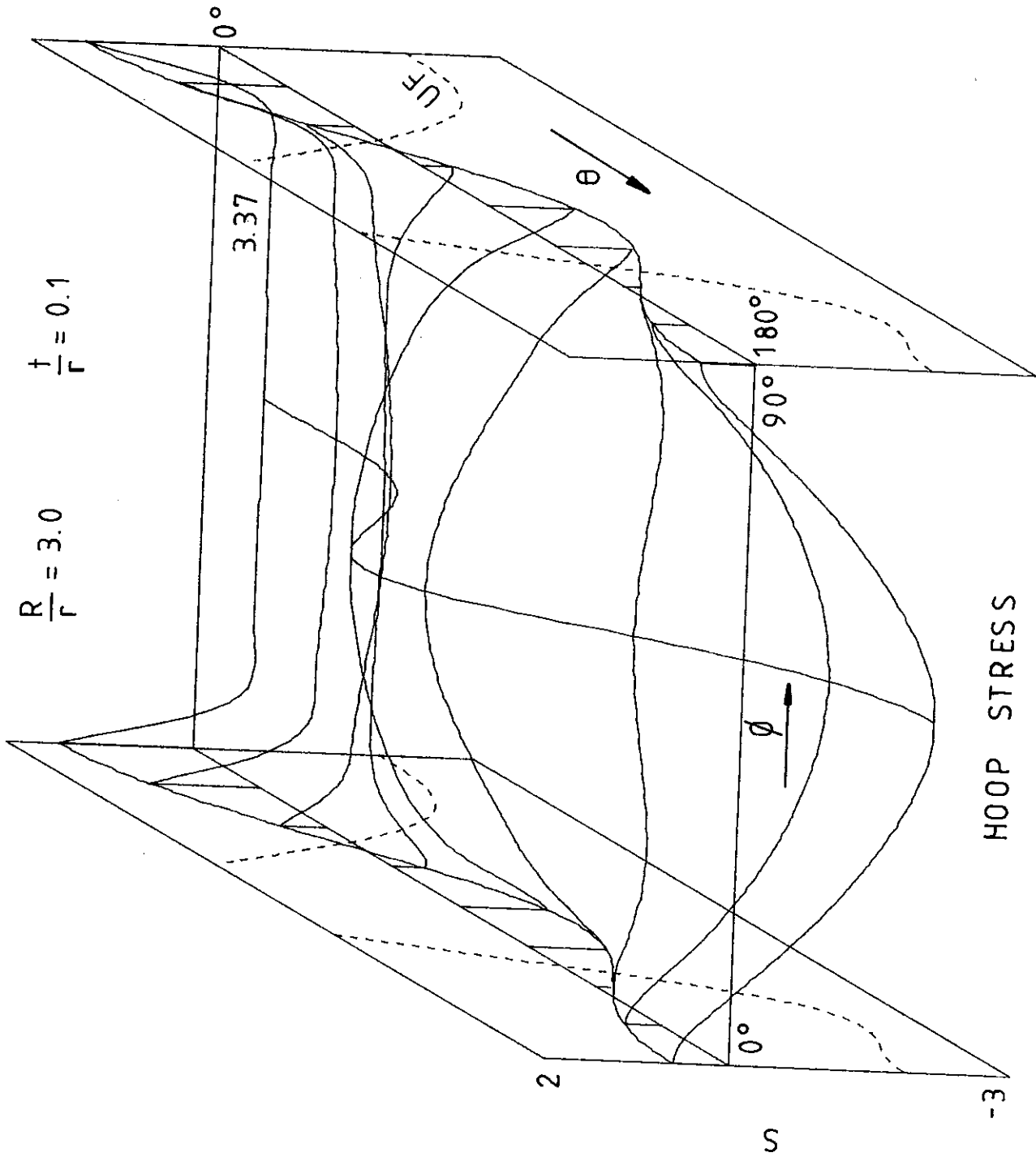
Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.4957	0.4093	0.2622	0.1685	0.1013	0.0477	0.0	-0.0477	-0.1013	-0.1685	-0.2622	-0.4093	-0.4957
45.0	0.7161	0.6114	0.4219	0.2844	0.1786	0.0866	0.0	-0.0866	-0.1786	-0.2844	-0.4219	-0.6114	-0.7161
67.5	0.4512	0.4278	0.3338	0.2392	0.1582	0.0795	0.0	-0.0795	-0.1582	-0.2392	-0.3338	-0.4278	-0.4512
90.0	-0.1830	-0.0090	0.0215	0.0159	0.0068	0.0017	0.0	-0.0017	-0.0068	-0.0159	-0.0215	-0.0090	0.1830
112.5	-0.6636	-0.3484	-0.2509	-0.1846	-0.1354	-0.0745	0.0	0.0745	0.1354	0.1846	0.2509	0.3484	0.6636
135.0	-0.6622	-0.5389	-0.4156	-0.2749	-0.1640	-0.0765	0.0	0.0765	0.1640	0.2749	0.4156	0.5389	0.6622
157.5	-0.3721	-0.4613	-0.3974	-0.2602	-0.1290	-0.0452	0.0	0.0452	0.1290	0.2602	0.3974	0.4613	0.3721
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

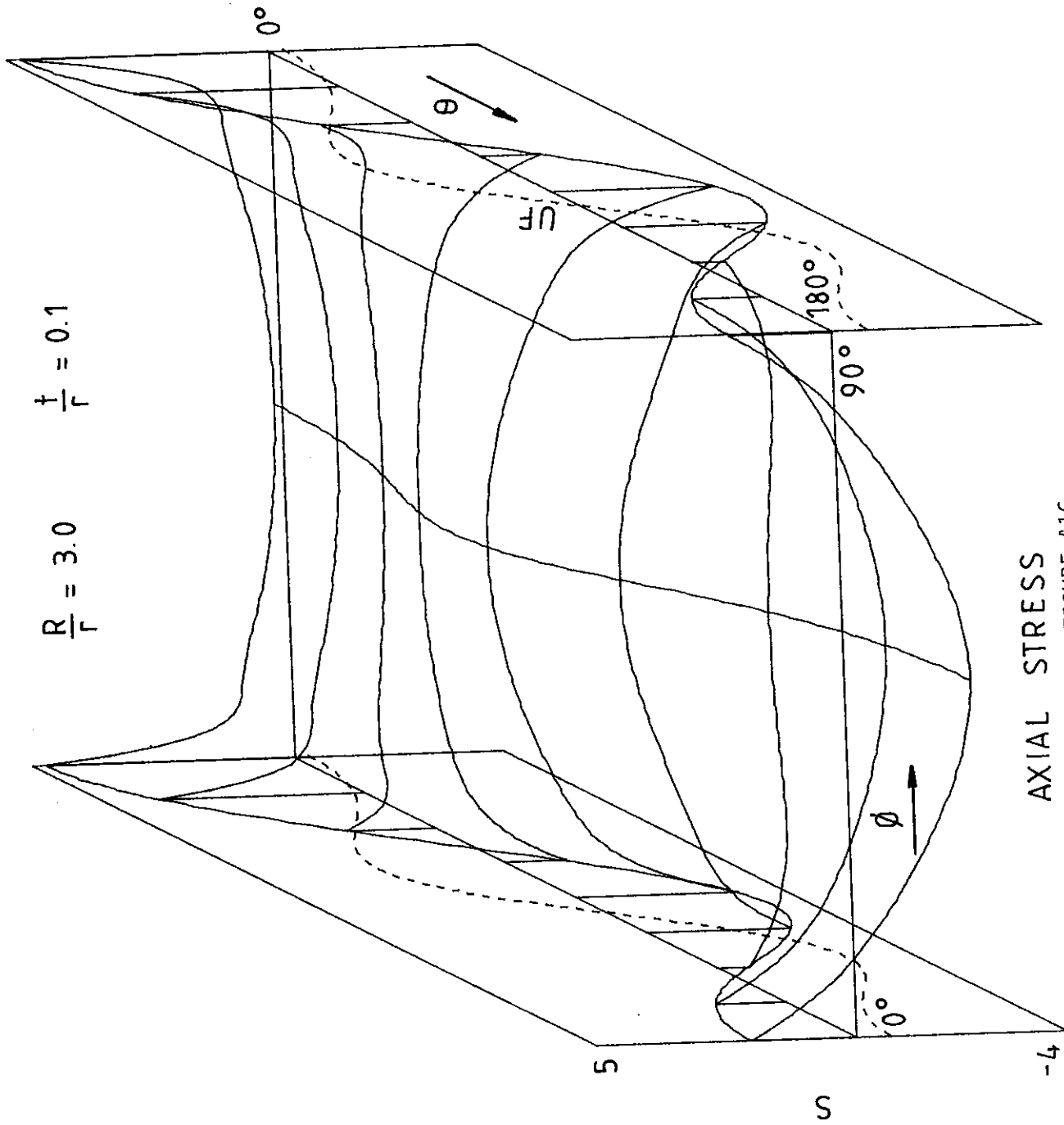
Theta	Phi=0.0	DIAMETER EXPANSION FACTORS					Without Flanges
		30.0	37.5	45.0	52.5	60.0	
90.0	0.0	5.197	10.730	15.461	22.052	21.301	92.780
						19.071	
						60.0	
						82.5	
						15.461	
						10.730	
						5.197	
						0.0	
						90.0	

TABLE A14

$$R/r = 3.0 \quad t/r = 0.05$$

K/I = 5.0														
Theta 0.0 22.5 45.0 67.5 90.0 112.5 135.0 157.5 180.0	Phi=0.0	INSIDE HOOP STRESS FACTORS									Without Flanges			
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0				
		-0.0100	0.0009	0.0038	0.0009	-0.0100	-0.0370	-0.0990	-0.2043	-0.3173	1.3121			
		0.1638	0.1893	0.1967	0.1893	0.1638	0.1108	0.0131	-0.1348	-0.2325	1.9794			
		0.4323	0.5083	0.5334	0.5083	0.4323	0.3040	0.1291	-0.0861	0.0092	2.0624			
		-0.0796	0.0158	0.0514	0.0158	-0.0796	-0.2014	-0.2886	-0.3389	0.2849	-2.2472			
		-1.3455	-1.4065	-1.4215	-1.4065	-1.3455	-1.1966	-0.9049	-0.5757	0.2748	-6.9128			
		-1.1641	-1.4178	-1.5081	-1.4178	-1.1641	-0.7987	-0.4024	-0.1935	-0.1405	-2.4216			
		0.1823	0.0237	-0.0434	0.0237	0.1823	0.3220	0.3315	0.1498	-0.4228	3.2124			
		-0.4228	0.1498	0.3315	0.3220	-0.4228	0.3326	0.0326	0.0312	-0.2648	2.1187			
157.5	-0.2648	0.0312	0.0326	0.4488	0.4902	0.4488	0.3326	0.1693	0.0326	0.9202				
180.0	-0.1197	-0.0369	-0.2551	0.3082	0.4103	0.3082	0.0590	-0.1902	-0.2551	-0.0369	-0.1197			
Without Flanges														
Theta 0.0 22.5 45.0 67.5 90.0 112.5 135.0 157.5 180.0	Phi=0.0	INSIDE AXIAL STRESS FACTORS									Without Flanges			
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0				
		0.4754	0.3791	0.3486	0.3791	0.4754	0.6545	0.9397	1.5819	-1.0577	-0.1552			
		0.5470	0.4582	0.4295	0.4582	0.5470	0.7050	0.9425	1.4497	-0.7750	0.3870			
		0.7971	0.7571	0.7427	0.7571	0.7971	0.8541	0.9077	0.9838	0.0305	1.7294			
		0.8900	0.9613	0.9835	0.9613	0.8900	0.7544	0.5206	0.0482	0.9498	1.5963			
		-0.1047	-0.4445	-0.1047	-0.4445	-0.1047	-0.1047	-0.4445	-1.0492	0.9161	-1.7255			
		-0.9064	-0.7494	-0.7306	-0.7494	-0.9064	-0.9064	-1.0516	-1.3240	-0.4683	-2.9195			
		-0.7490	-0.7909	-0.8087	-0.7909	-0.7490	-0.7050	-0.6638	-0.6982	-1.4092	0.0782			
		-0.5104	-0.5312	-0.5368	-0.5312	-0.5104	-0.4556	-0.3231	-0.2154	-0.8828	1.2644			
157.5	-0.8828	-0.2154	-0.3231	-0.4556	-0.5104	-0.4556	-0.3231	-0.2154	-0.8828	1.2644				
180.0	-0.3989	-0.1108	-0.3179	-0.4968	-0.5114	-0.4968	-0.3179	-0.1108	-0.3989	0.9518				
Without Flanges														
Theta 0.0 22.5 45.0 67.5 90.0 112.5 135.0 157.5 180.0	Phi=0.0	INSIDE SHEAR STRESS FACTORS									Without Flanges			
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0				
		0.0986	0.0446	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0			
		0.2298	0.1078	0.0	-0.0446	-0.0986	-0.1740	-0.2874	-0.4887	-0.4715	0.0			
		0.3627	0.1812	0.0	-0.1078	-0.2298	-0.3818	-0.5821	-0.8957	-0.6812	0.0			
		0.3056	0.1705	0.0	-0.1812	-0.3627	-0.5430	-0.7163	-0.8953	-0.4292	0.0			
		-0.1260	-0.0425	0.0	-0.1705	-0.3056	-0.3698	-0.3268	-0.1286	0.1741	0.0			
		-0.2852	-0.0425	0.0	0.0425	0.1260	0.2852	0.5407	0.8881	0.6312	0.0			
		-0.9158	-0.3506	0.0	0.3506	0.6669	0.9158	1.0654	1.0688	0.6299	0.0			
		-0.8287	-0.3921	0.0	0.3921	0.6928	0.8287	0.7680	0.5447	0.3540	0.0			
157.5	-0.3540	-0.5447	-0.7680	-0.6928	-0.3921	0.0	0.0	0.0	0.0	0.0				
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0				
Without Flanges														
Theta 180.0	Phi=0.0	DIAMETER EXPANSION FACTORS									Without Flanges			
		30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0				
180.0	0.0	-3.979	-7.777	-11.845	-15.532	-18.119	-19.049	-18.119	-15.532	-11.845	-7.777	-3.979	90.0	0.0
Without Flanges														
-68.001														





AXIAL STRESS
FIGURE A16

TABLE A15

 $R/r = 3.0$ $t/r = 0.1$

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS									Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5		75.0	82.5
0.0	1.4232	-0.5392	-0.6244	-0.6155	-0.6238	-0.6258	-0.6261	-0.6258	-0.6238	-0.6155	-0.6244	-0.5392	1.4232
22.5	1.1748	-0.4506	-0.5657	-0.6037	-0.6467	-0.6716	-0.6798	-0.6716	-0.6467	-0.6037	-0.5657	-0.4506	1.1748
45.0	0.4993	-0.1960	-0.2794	-0.3757	-0.4671	-0.5268	-0.5477	-0.5268	-0.4671	-0.3757	-0.2794	-0.1960	0.4993
67.5	-0.3539	0.0927	0.2909	0.3008	0.2808	0.2602	0.2516	0.2602	0.2808	0.3008	0.2909	0.0927	-0.3539
90.0	-0.9351	0.1136	0.6728	0.9545	1.1570	1.2822	1.3241	1.2822	1.1570	0.9545	0.6728	0.1136	-0.9351
112.5	-0.8402	-0.2024	0.2624	0.6237	0.9443	1.1657	1.2439	1.1657	0.9443	0.6237	0.2624	-0.2024	-0.8402
135.0	-0.1921	-0.3649	-0.4885	-0.4741	-0.3876	-0.3059	-0.2742	-0.3059	-0.3876	-0.4741	-0.4885	-0.3649	-0.1921
157.5	0.4032	-0.0926	-0.6749	-1.0948	-1.3939	-1.5831	-1.6488	-1.5831	-1.3939	-1.0948	-0.6749	-0.0926	0.4032
180.0	0.6006	0.1241	-0.5683	-1.1519	-1.6222	-1.9413	-2.0553	-1.9413	-1.6222	-1.1519	-0.5683	0.1241	0.6006
													-1.8824

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	4.7440	0.9548	0.7392	0.4397	0.2456	0.1406	0.1068	0.1406	0.2456	0.4397	0.7392	0.9548	4.7440
22.5	3.9160	0.9631	0.8105	0.5529	0.3787	0.2812	0.2493	0.2812	0.3787	0.5529	0.8105	0.9631	3.9160
45.0	1.6645	0.8946	0.9777	0.8844	0.8071	0.7577	0.7403	0.7577	0.8071	0.8844	0.9777	0.8946	1.6645
67.5	-1.1796	0.4809	0.9618	1.1736	1.3100	1.3836	1.4064	1.3836	1.3100	1.1736	0.9618	0.4809	-1.1796
90.0	-3.1171	-0.4428	0.3003	0.7807	1.1429	1.3610	1.4332	1.3610	1.1429	0.7807	0.3003	-0.4428	-3.1171
112.5	-2.8006	-1.3426	-0.8594	-0.4314	-0.0577	0.1821	0.2635	0.1821	-0.0577	-0.4314	-0.8594	-1.3426	-2.8006
135.0	-0.6404	-1.2331	-1.4403	-1.4219	-1.3389	-1.2760	-1.2541	-1.2760	-1.3389	-1.4219	-1.4403	-1.2331	-0.6404
157.5	1.3441	-0.2189	-1.0114	-1.4919	-1.8399	-2.0651	-2.1435	-2.0651	-1.8399	-1.4919	-1.0114	-0.2189	1.3441
180.0	2.0020	0.3662	-0.6163	-1.3251	-1.8945	-2.2744	-2.4075	-2.2744	-1.8945	-1.3251	-0.6163	0.3662	2.0020
													-0.6837

Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.3217	0.2427	0.1852	0.1198	0.0742	0.0359	0.0	-0.0359	-0.0742	-0.1198	-0.1852	-0.2427	-0.3217
45.0	0.4508	0.3237	0.2577	0.1649	0.1001	0.0478	0.0	-0.0478	-0.1001	-0.1649	-0.2577	-0.3237	-0.4508
67.5	0.2819	0.1945	0.1563	0.0863	0.0427	0.0171	0.0	-0.0171	-0.0427	-0.0863	-0.1563	-0.1945	-0.2819
90.0	-0.1055	-0.0149	-0.0340	-0.0579	-0.0591	-0.0365	0.0	0.0365	0.0591	0.0579	0.0340	0.0149	0.1055
112.5	-0.4504	-0.1467	-0.1324	-0.1079	-0.0796	-0.0433	0.0	0.0433	0.0796	0.1079	0.1324	0.1467	0.4504
135.0	-0.5240	-0.2171	-0.1223	-0.0346	0.0134	0.0187	0.0	-0.0187	-0.0134	0.0346	0.1223	0.2171	0.5240
157.5	-0.3293	-0.2055	-0.0942	0.0109	0.0638	0.0521	0.0	-0.0521	-0.0638	-0.0109	0.0942	0.2055	0.3293
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

Theta	Phi=0.0	DIAMETER EXPANSION FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
90.0	0.0	2.971	6.858	10.147	12.713	14.350	14.911	14.350	12.713	10.147	6.858	2.971	0.0	36.846

TABLE A16

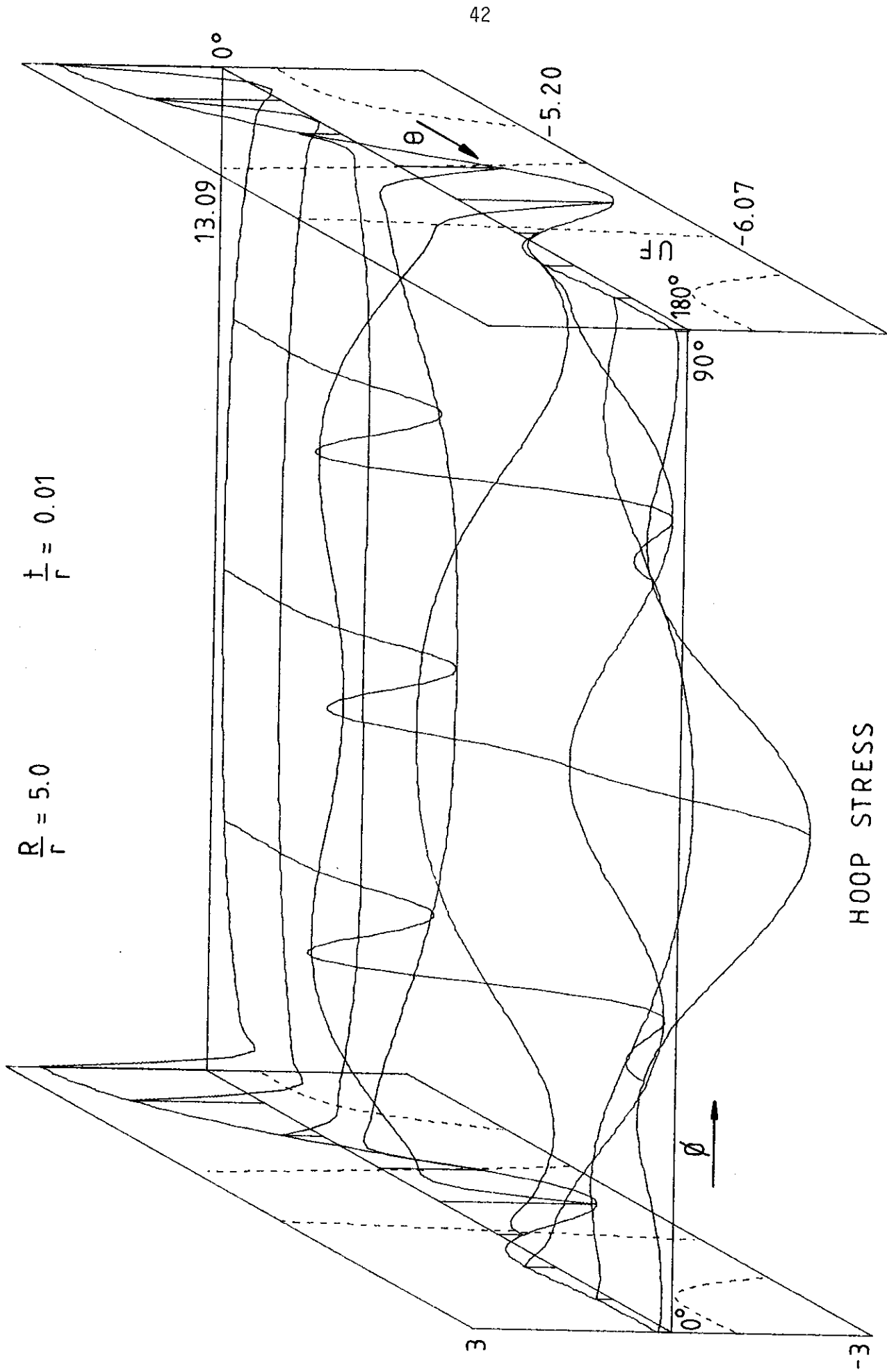
R/r = 3.0 t/r = 0.1

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.3431	0.1553	0.1925	0.3395	0.4152	0.4534	0.4651	0.4534	0.4152	0.3395	0.1925	0.1553	-0.3431	1.9872
22.5	-0.2454	0.0988	0.1790	0.3546	0.4606	0.5208	0.5404	0.5208	0.4606	0.3546	0.1790	0.0988	-0.2454	1.8291
45.0	0.0069	-0.1128	-0.0100	0.1726	0.3111	0.4019	0.4336	0.4019	0.3111	0.1726	-0.0100	-0.1128	0.0069	0.6534
67.5	0.2732	-0.4440	-0.5177	-0.5136	-0.4816	-0.4447	-0.4293	-0.4447	-0.4816	-0.5136	-0.5177	-0.4440	0.2732	-2.1465
90.0	0.3296	-0.5810	-0.9136	-1.2435	-1.4872	-1.6279	-1.6737	-1.6279	-1.4872	-1.2435	-0.9136	-0.5810	0.3296	-4.3099
112.5	0.0591	-0.2878	-0.5445	-0.9727	-1.3719	-1.6417	-1.7367	-1.6417	-1.3719	-0.9727	-0.5445	-0.2878	0.0591	-2.5210
135.0	-0.3146	0.0539	0.1528	0.0483	-0.1145	-0.2471	-0.2976	-0.2471	-0.1145	0.0483	0.1528	0.0539	-0.3146	1.1824
157.5	-0.4707	0.0491	0.2507	0.4709	0.6985	0.8638	0.9232	0.8638	0.6985	0.4709	0.2507	0.0491	-0.4707	2.2678
180.0	-0.4692	-0.0371	0.0956	0.3968	0.7892	1.1014	1.2181	1.1014	0.7892	0.3968	0.0956	-0.0371	-0.4692	1.8776

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-1.1435	1.5870	0.9231	0.6680	0.5139	0.4248	0.3960	0.4248	0.5139	0.6680	0.9231	1.5870	-1.1435	0.4816
22.5	-0.8179	1.4239	0.9198	0.7350	0.6207	0.5535	0.5316	0.5535	0.6207	0.7350	0.9198	1.4239	-0.8179	0.7644
45.0	0.0230	0.9213	0.8013	0.8052	0.8007	0.7954	0.7937	0.7954	0.8007	0.8052	0.8013	0.9213	0.0230	1.1429
67.5	0.9108	0.1378	0.3475	0.5488	0.6606	0.7225	0.7431	0.7225	0.6606	0.5488	0.3475	0.1378	0.9108	0.5752
90.0	1.0985	-0.6141	-0.3913	-0.1611	-0.0427	0.0203	0.0408	0.0203	-0.0427	-0.1611	-0.3913	-0.6141	1.0985	-1.0452
112.5	0.1970	-0.9018	-0.8499	-0.7756	-0.7713	-0.7815	-0.7862	-0.7815	-0.7713	-0.7756	-0.8499	-0.9018	0.1970	-1.8210
135.0	-1.0486	-0.7306	-0.6670	-0.7053	-0.7667	-0.8097	-0.8245	-0.8097	-0.7667	-0.7053	-0.6670	-0.7306	-1.0486	-0.5912
157.5	-1.5688	-0.5412	-0.3178	-0.3099	-0.2988	-0.2733	-0.2609	-0.2733	-0.2988	-0.3099	-0.3178	-0.5412	-1.5688	0.8534
180.0	-1.5641	-0.5007	-0.2098	-0.1494	-0.0744	0.0057	0.0394	0.0057	-0.0744	-0.1494	-0.2098	-0.5007	-1.5641	1.2586

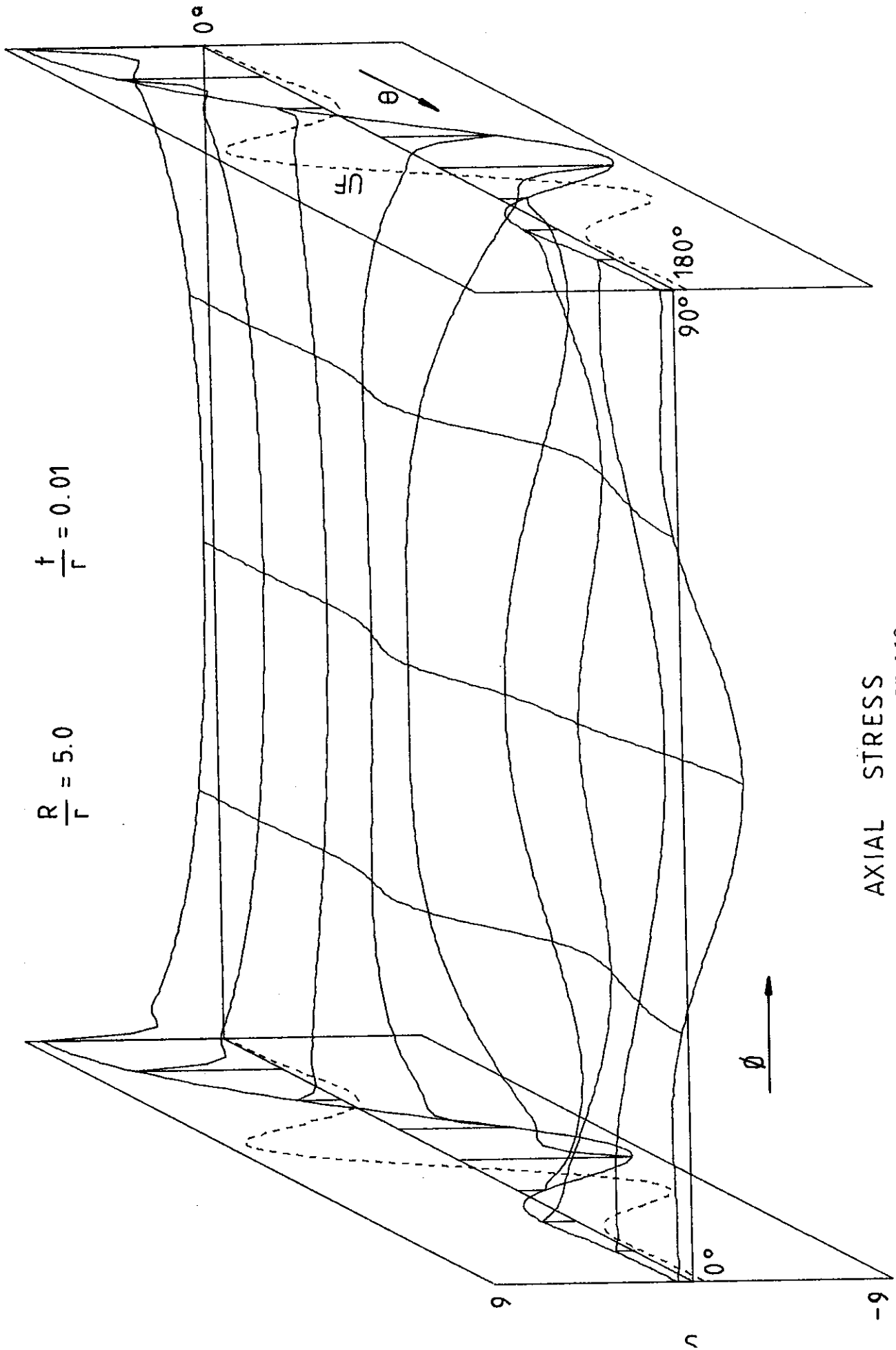
Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.2911	0.5119	0.3115	0.2003	0.1215	0.0575	0.0	-0.0575	-0.1215	-0.2003	-0.3115	-0.5119	-0.2911
45.0	0.4078	0.8551	0.5794	0.3969	0.2521	0.1228	0.0	-0.1228	-0.2521	-0.3969	-0.5794	-0.8551	-0.4078
67.5	0.2551	0.7838	0.6417	0.4889	0.3321	0.1681	0.0	-0.1681	-0.3321	-0.4889	-0.6417	-0.7838	-0.2551
90.0	-0.0954	0.1584	0.2726	0.2791	0.2188	0.1187	0.0	-0.1187	-0.2188	-0.2791	-0.2726	-0.1584	0.0954
112.5	-0.4075	-0.6756	-0.4650	-0.2831	-0.1604	-0.0726	0.0	0.0726	0.1604	0.2831	0.4650	0.6756	0.4075
135.0	-0.4741	-1.0191	-0.9957	-0.8094	-0.5668	-0.2915	0.0	0.2915	0.5668	0.8094	0.9957	1.0191	0.4741
157.5	-0.2979	-0.6622	-0.7902	-0.7294	-0.5482	-0.2920	0.0	0.2920	0.5482	0.7294	0.7902	0.6622	0.2979
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS											Without Flanges		
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-3.131	-6.537	-9.682	-12.280	-13.987	-14.582	-13.987	-12.280	-9.682	-6.537	-3.131	0.0
													-32.792



HOO P STRESS

FIGURE A17



AXIAL STRESS
FIGURE A18

TABLE A17

$$R/r = 5.0 \quad t/r = 0.01$$

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										90.0	Flanges	Without	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0				82.5
0.0	2.4725	-0.4347	-0.2950	-0.2158	-0.1714	-0.1488	-0.1418	-0.1488	-0.1714	-0.2158	-0.2950	-0.4347	2.4725	-0.7686	
22.5	2.0026	-0.3594	-0.2461	-0.1766	-0.1377	-0.1187	-0.1131	-0.1187	-0.1377	-0.1766	-0.2461	-0.3594	2.0026	-0.9662	
45.0	0.6022	-0.3379	-0.4703	-0.5173	-0.5184	-0.5091	-0.5051	-0.5091	-0.5184	-0.5173	-0.4703	-0.3379	0.6022	-4.1015	
67.5	-1.5706	0.0506	-0.3314	-0.6911	-0.9103	-0.9961	-1.0129	-0.9961	-0.9103	-0.6911	-0.3314	0.0506	-1.5706	-1.1074	
90.0	-2.3670	0.8402	1.6726	1.9783	1.8777	1.6594	1.5577	1.6594	1.8777	1.9783	1.6726	0.8402	-2.3670	13.0851	
112.5	-0.3880	-0.8010	-0.5660	0.2219	0.9151	1.2573	1.3402	1.2573	0.9151	0.2219	-0.5660	-0.8010	-0.3880	-1.7045	
135.0	0.4575	-0.2893	-1.1675	-1.5631	-1.1722	-0.4354	-0.0735	-0.4354	-1.1722	-1.5631	-1.1675	-0.2893	0.4575	-4.2777	
157.5	0.2564	0.3434	0.2676	-0.2078	-0.6911	-0.9766	-1.0666	-0.9766	-0.6911	-0.2078	0.2676	0.3434	0.2564	-0.9446	
180.0	0.2042	0.2111	0.4878	0.5009	-0.2904	-1.4194	-1.9432	-1.4194	-0.2904	0.5009	0.4878	0.2111	0.2042	-0.9729	

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	8.2417	2.2647	1.3557	0.8614	0.5897	0.4525	0.4106	0.4525	0.5897	0.8614	1.3557	2.2647	8.2417	-0.0788
22.5	6.6753	2.0021	1.2458	0.8092	0.5595	0.4298	0.3894	0.4298	0.5595	0.8092	1.2458	2.0021	6.6753	-0.3898
45.0	2.0073	1.1319	0.8096	0.5199	0.3241	0.2174	0.1841	0.2174	0.3241	0.5199	0.8096	1.1319	2.0073	-0.8044
67.5	-5.2355	-0.0245	0.8357	1.0203	0.9639	0.8810	0.8488	0.8810	0.9639	1.0203	0.8357	-0.0245	-5.2355	5.6610
90.0	-7.8899	-2.0979	-0.0676	1.1629	1.7190	1.8676	1.8781	1.8676	1.7190	1.1629	-0.0676	-2.0979	-7.8899	4.2467
112.5	-1.2933	-3.0192	-2.8195	-1.8650	-0.8423	-0.1169	0.1422	-0.1169	-0.8423	-1.8650	-2.8195	-3.0192	-1.2933	-6.8774
135.0	1.5251	-0.2561	-1.4314	-1.8336	-1.4778	-0.8583	-0.5647	-0.8583	-1.4778	-1.8336	-1.4314	-0.2561	1.5251	-1.4846
157.5	0.8545	0.7782	0.3374	-0.4274	-1.1343	-1.6365	-1.8258	-1.6365	-1.1343	-0.4274	0.3374	0.7782	0.8545	-0.3459
180.0	0.6807	0.6942	0.7819	0.2875	-0.9300	-2.1657	-2.6707	-2.1657	-0.9300	0.2875	0.7819	0.6942	0.6807	-0.5141

Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.9549	0.5738	0.3165	0.1779	0.0967	0.0428	0.0	-0.0428	-0.0967	-0.1779	-0.3165	-0.5738	-0.9549
45.0	1.3466	0.9399	0.5693	0.3414	0.1919	0.0860	0.0	-0.0860	-0.1919	-0.3414	-0.5693	-0.9399	-1.3466
67.5	0.5267	0.6075	0.4946	0.3791	0.2549	0.1268	0.0	-0.1268	-0.2549	-0.3791	-0.4946	-0.6075	-0.5267
90.0	-0.9798	-0.3612	-0.0881	0.0541	0.1151	0.0903	0.0	-0.0903	-0.1151	-0.0541	0.0881	0.3612	0.9798
112.5	-0.9435	-0.7016	-0.4218	-0.3031	-0.2274	-0.1292	0.0	0.1292	0.2274	0.3031	0.4218	0.7016	0.9435
135.0	-0.1833	-0.5289	-0.4954	-0.3965	-0.3348	-0.2143	0.0	0.2143	0.3348	0.3965	0.4954	0.5289	0.1833
157.5	0.0008	-0.1490	-0.3866	-0.3902	-0.1918	-0.0312	0.0	0.0312	0.1918	0.3902	0.3866	0.1490	-0.0008
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS														
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
90.0	0.0	24.991	56.521	82.171	97.442	104.044	105.650	104.044	97.442	82.171	56.521	24.991	0.0	598.848

TABLE A13

R/r = 5.0 t/r = 0.01

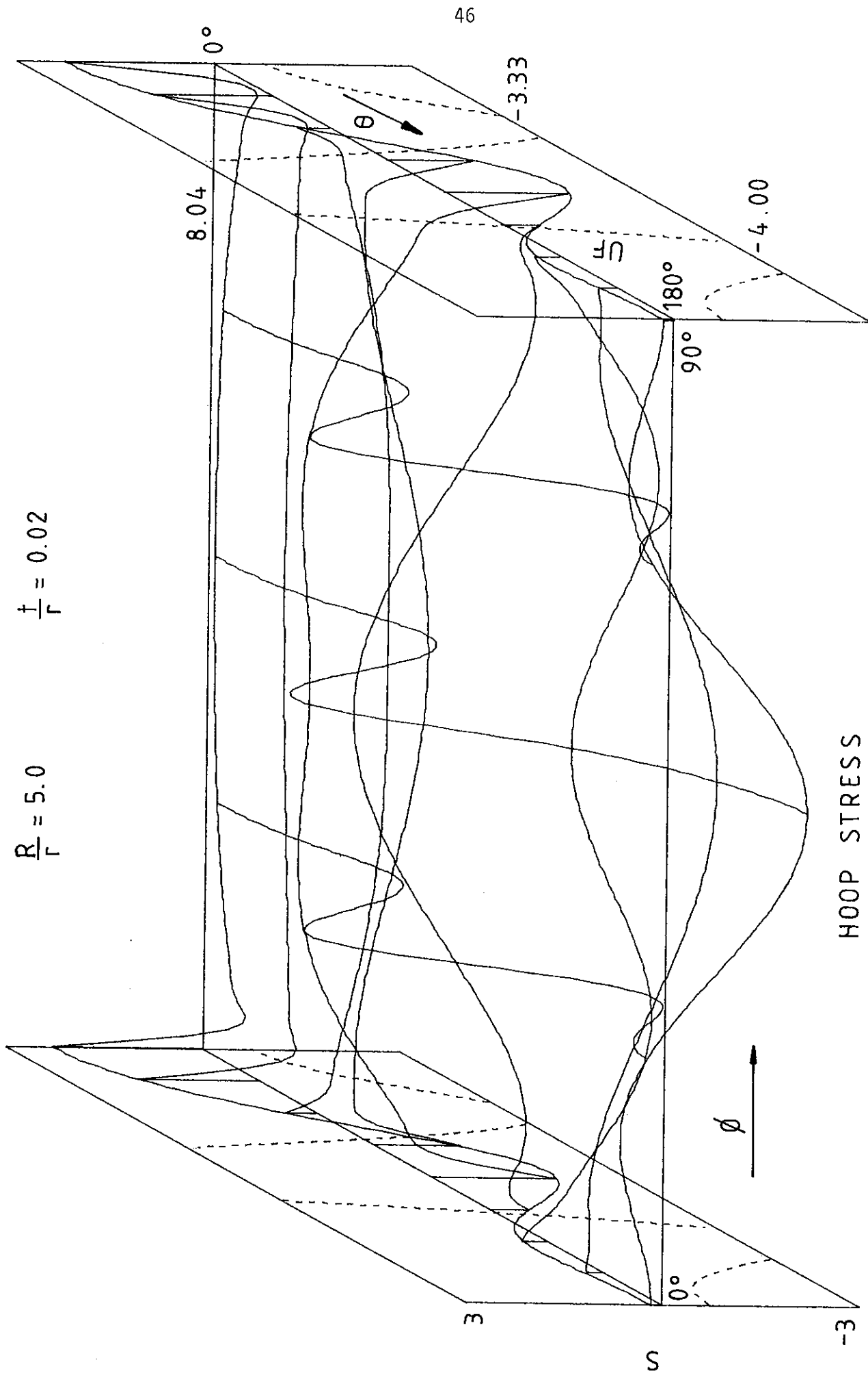
Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										90.0	82.5	75.0	67.5	60.0	52.5	45.0	37.5	30.0	22.5	15.0	7.5	Without Flanges
0.0	-0.0812	-0.3192	-0.1573	-0.0723	-0.0263	-0.0033	0.0037	-0.0033	-0.0263	-0.0723	-0.1573	-0.3192	-0.0812	0.7688	-0.0263	-0.0033	-0.0033	-0.0263	-0.0723	-0.1573	-0.3192	-0.0812	0.7688	
22.5	-0.0291	-0.2669	-0.1491	-0.0838	-0.0449	-0.0234	-0.0165	-0.0234	-0.0449	-0.0838	-0.1491	-0.2669	-0.0291	0.9700	-0.0449	-0.0234	-0.0234	-0.0449	-0.0838	-0.1491	-0.2669	-0.0291	0.9700	
45.0	0.0895	0.0550	0.2463	0.3504	0.3931	0.4086	0.4128	0.4086	0.3931	0.3504	0.2463	0.0550	0.0895	4.1374	0.3931	0.4086	0.4086	0.3931	0.3504	0.2463	0.0550	0.0895	4.1374	
67.5	0.1633	-0.0095	0.2886	0.6343	0.8568	0.9458	0.9634	0.9458	0.8568	0.6343	0.2886	-0.0095	0.1633	0.8029	0.8568	0.9458	0.9458	0.8568	0.6343	0.2886	-0.0095	0.1633	0.8029	
90.0	-0.2463	-0.9301	-1.8270	-2.1339	-2.0081	-1.7683	-1.6591	-1.7683	-2.0081	-2.1339	-1.8270	-0.9301	-0.2463	-14.1285	-2.0081	-1.7683	-1.7683	-2.0081	-2.1339	-1.8270	-0.9301	-0.2463	-14.1285	
112.5	-0.5546	0.4580	0.2348	-0.5034	-1.1083	-1.3534	-1.3926	-1.3534	-1.1083	-0.5034	0.2348	0.4580	-0.5546	1.3597	-1.1083	-1.3534	-1.3534	-1.1083	-0.5034	0.2348	0.4580	-0.5546	1.3597	
135.0	-0.0010	0.1882	0.8816	1.2498	0.9028	0.2102	-0.1364	0.2102	0.9028	1.2498	0.8816	0.1882	-0.0010	4.3291	0.9028	0.2102	0.2102	0.9028	1.2498	0.8816	0.1882	-0.0010	4.3291	
157.5	0.1966	-0.0689	-0.2247	0.0134	0.2844	0.3973	0.4178	0.3973	0.2844	0.0134	-0.2247	-0.0689	0.1966	0.9459	0.3973	0.4178	0.4178	0.3973	0.2844	-0.2247	-0.0689	0.1966	0.9459	
180.0	0.1815	0.1450	-0.2074	-0.5604	-0.1910	0.6502	1.0798	0.6502	-0.1910	-0.5604	-0.2074	0.1450	0.1815	0.9726	-0.1910	1.0798	1.0798	-0.1910	-0.5604	-0.2074	0.1450	0.1815	0.9726	

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.2705	2.2663	1.3710	0.8810	0.6105	0.4735	0.4315	0.4735	0.6105	0.8810	1.3710	2.2663	-0.2705	0.1070
22.5	-0.0969	1.9907	1.2465	0.8121	0.5630	0.4341	0.3941	0.4341	0.5630	0.8121	1.2465	1.9907	-0.0969	-0.1054
45.0	0.2982	1.1849	0.9789	0.7412	0.5596	0.4541	0.4203	0.4541	0.5596	0.7412	0.9789	1.1849	0.2982	1.1871
67.5	0.5444	-0.1619	0.8590	1.2499	1.3340	1.3105	1.2907	1.3105	1.3340	1.2499	0.8590	-0.1619	0.5444	5.2009
90.0	-0.8212	-2.5117	-1.1961	-0.2533	0.3444	0.6474	0.7347	0.6474	0.3444	-0.2533	-1.1961	-2.5117	-0.8212	-3.9271
112.5	-1.8486	-2.2759	-2.1996	-1.8477	-1.3732	-0.9299	-0.7434	-0.9299	-1.3732	-1.8477	-2.1996	-2.2759	-1.8486	-4.7217
135.0	-0.0035	-0.1854	-0.5662	-0.6026	-0.5933	-0.6433	-0.6809	-0.6433	-0.5933	-0.6026	-0.5662	-0.1854	-0.0035	1.7084
157.5	0.6554	0.5051	0.0435	-0.3357	-0.6492	-0.9200	-1.0346	-0.9200	-0.6492	-0.3357	0.0435	0.5051	0.6554	0.6539
180.0	0.6051	0.6544	0.3395	-0.3302	-0.8651	-1.0342	-1.0376	-1.0342	-0.8651	-0.3302	0.3395	0.6544	0.6051	0.4826

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.9454	0.5872	0.3197	0.1795	0.0979	0.0435	0.0	-0.0435	-0.0979	-0.1795	-0.3197	-0.5872	-0.9454
45.0	1.3332	1.0079	0.5967	0.3503	0.1955	0.0880	0.0	-0.0880	-0.1955	-0.3503	-0.5967	-1.0079	-1.3332
67.5	0.5215	0.9446	0.7536	0.5170	0.3059	0.1379	0.0	-0.1379	-0.3059	-0.5170	-0.7536	-0.9446	-0.5215
90.0	-0.9700	-0.3156	0.2292	0.4165	0.3714	0.2068	0.0	-0.2068	-0.3714	-0.4165	-0.2292	0.3156	0.9700
112.5	-0.9341	-1.2756	-0.8427	-0.3382	0.0053	0.0980	0.0	-0.0980	-0.0053	0.3382	0.8427	1.2756	0.9341
135.0	-0.1815	-0.4719	-0.8638	-0.9215	-0.7218	-0.3920	0.0	0.3920	0.7218	0.9215	0.8638	0.4719	0.1815
157.5	0.0008	0.0345	-0.2201	-0.6520	-0.8527	-0.5944	0.0	0.5944	0.8527	0.6520	0.2201	-0.0345	-0.0008
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS														
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
Theta	Phi=0.0	-7.795	-16.543	-32.557	-56.578	-79.619	-89.190	-79.619	-56.578	-32.557	-16.543	-7.795	0.0	-330.281
180.0	0.0	-7.795	-16.543	-32.557	-56.578	-79.619	-89.190	-79.619	-56.578	-32.557	-16.543	-7.795	0.0	-330.281

$$\frac{R}{r} = 5.0 \quad \frac{t}{r} = 0.02$$



HOOP STRESS
FIGURE A19

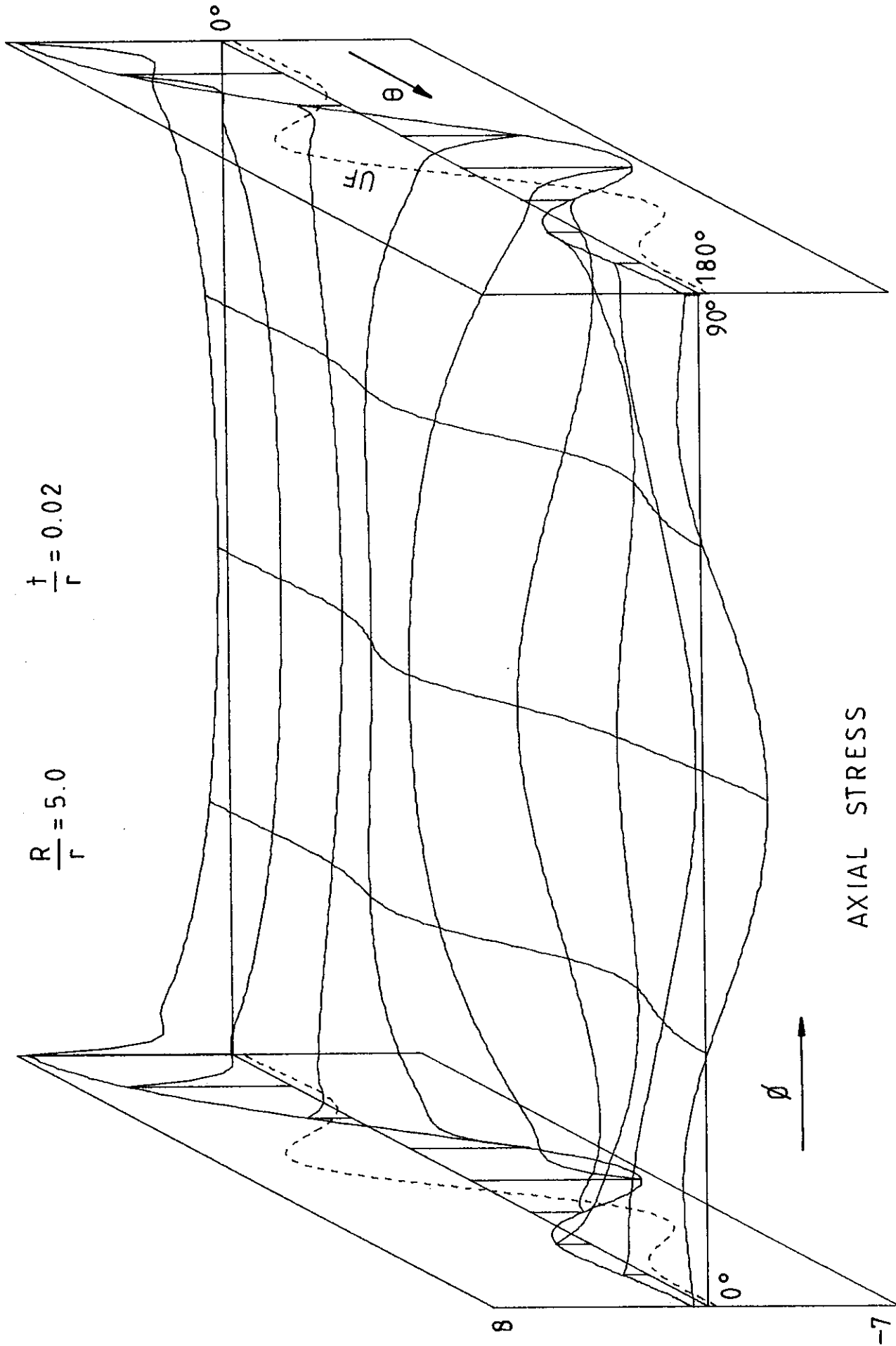


FIGURE A20

TABLE A19

$$R/r = 5.0 \quad t/r = 0.02$$

OUTSIDE HOOP STRESS FACTORS

Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
0.0	2.2792	-0.4104	-0.2699	-0.1815	-0.1320	-0.1086	-0.1020	-0.1086	-0.1320	-0.1815	-0.2699	-0.4104	2.2792	-0.7486
22.5	1.8322	-0.4314	-0.3983	-0.3593	-0.3226	-0.2968	-0.2877	-0.2968	-0.3226	-0.3593	-0.3983	-0.4314	1.8322	-1.7219
45.0	0.4774	-0.3790	-0.6724	-0.8728	-0.9726	-1.0048	-1.0095	-1.0048	-0.9726	-0.8728	-0.6724	-0.3790	0.4774	-3.3249
67.5	-1.3291	0.3121	0.1735	-0.1233	-0.4306	-0.6509	-0.7302	-0.6509	-0.4306	-0.1233	0.1735	0.3121	-1.3291	1.3038
90.0	-1.9080	0.7204	1.5346	1.9975	2.0966	2.0120	1.9534	2.0120	2.0966	1.9975	1.5346	0.7204	-1.9080	8.0354
112.5	-0.5983	-0.5410	-0.1935	0.5636	1.3583	1.9441	2.1601	1.9441	1.3583	0.5636	-0.1935	-0.5410	-0.5983	1.1808
135.0	0.3703	-0.6356	-1.3781	-1.5313	-1.1004	-0.5269	-0.2733	-0.5269	-1.1004	-1.5313	-1.3781	-0.6356	0.3703	-3.9828
157.5	0.2868	0.2470	-0.0334	-0.5517	-1.0840	-1.4812	-1.6298	-1.4812	-1.0840	-0.5517	-0.0334	0.2470	0.2868	-1.6276
180.0	0.1694	0.4234	0.6666	0.2708	-0.7216	-1.7097	-2.1130	-1.7097	-0.7216	0.2708	0.6666	0.4234	0.1694	-0.7488

OUTSIDE AXIAL STRESS FACTORS

Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
0.0	7.5975	1.9827	1.1543	0.7108	0.4707	0.3503	0.3136	0.3503	0.4707	0.7108	1.1543	1.9827	7.5975	-0.3361
22.5	6.1073	1.7323	1.0021	0.5778	0.3392	0.2198	0.1838	0.2198	0.3392	0.5778	1.0021	1.7323	6.1073	-0.5878
45.0	1.5914	1.1267	0.8669	0.5732	0.3429	0.2041	0.1584	0.2041	0.3429	0.5732	0.8669	1.1267	1.5914	0.5610
67.5	-4.4304	0.2082	1.1149	1.4100	1.4137	1.3276	1.2828	1.3276	1.4137	1.4100	1.1149	0.2082	-4.4304	4.5687
90.0	-6.3601	-1.6323	-0.0202	1.0623	1.6971	2.0085	2.0994	2.0085	1.6971	1.0623	-0.0202	-1.6323	-6.3601	2.6695
112.5	-1.9942	-2.6447	-2.3028	-1.5128	-0.6394	0.0445	0.3053	0.0445	-0.6394	-1.5128	-2.3028	-2.6447	-1.9942	-4.0627
135.0	1.2345	-0.8642	-1.7398	-1.9139	-1.6572	-1.3430	-1.2133	-1.3430	-1.6572	-1.9139	-1.7398	-0.8642	1.2345	-2.7148
157.5	0.9559	0.6382	0.0866	-0.6031	-1.2550	-1.7389	-1.9204	-1.7389	-1.2550	-0.6031	0.0866	0.6382	0.9559	-0.3891
180.0	0.5645	0.8356	0.7532	-0.0330	-1.1462	-2.0305	-2.3564	-2.0305	-1.1462	-0.0330	0.7532	0.8356	0.5645	-0.3233

48

OUTSIDE SHEAR STRESS FACTORS

Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.7908	0.5171	0.2918	0.1663	0.0901	0.0394	0.0	-0.0394	-0.0901	-0.1663	-0.2918	-0.5171	-0.7908	0.0
45.0	1.0479	0.7658	0.4761	0.3021	0.1813	0.0855	0.0	-0.0855	-0.1813	-0.3021	-0.4761	-0.7658	-1.0479	0.0
67.5	0.3684	0.4152	0.3157	0.2498	0.1856	0.1022	0.0	-0.1022	-0.1856	-0.2498	-0.3157	-0.4152	-0.3684	0.0
90.0	-0.6701	-0.2372	-0.1090	-0.0587	-0.0288	-0.0108	0.0	0.0108	0.0288	0.0587	0.1090	0.2372	0.6701	0.0
112.5	-0.8083	-0.4904	-0.2815	-0.2491	-0.2405	-0.1577	0.0	0.1577	0.2405	0.2491	0.2815	0.4904	0.8083	0.0
135.0	-0.2743	-0.4585	-0.3284	-0.2109	-0.1315	-0.0643	0.0	0.0643	0.1315	0.2109	0.3284	0.4585	0.2743	0.0
157.5	-0.0048	-0.2605	-0.3394	-0.1631	0.0405	0.0916	0.0	-0.0916	-0.0405	0.1631	0.3394	0.2605	0.0048	0.0
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS

Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
0.0	0.0	16.817	38.288	57.918	72.166	80.354	82.972	80.354	72.166	57.918	38.288	16.817	0.0	258.322

TABLE A20

$$R/r = 5.0 \quad t/r = 0.02$$

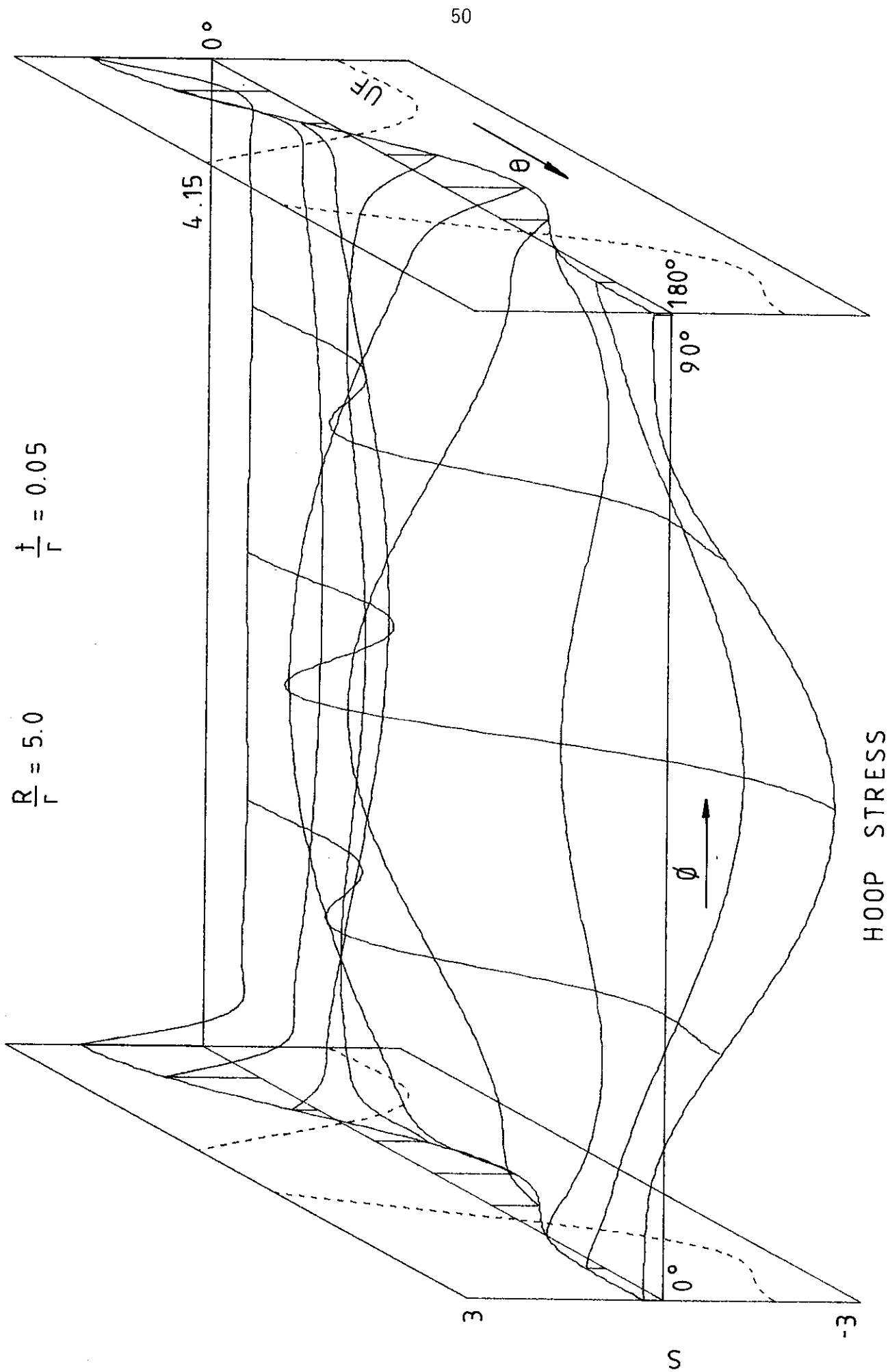
Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.1657	-0.2507	-0.1268	-0.0685	-0.0367	-0.0187	-0.0126	-0.0187	-0.0367	-0.0685	-0.1268	-0.2507	-0.1657	0.7491
22.5	-0.0950	-0.1123	0.0583	0.1396	0.1722	0.1825	0.1845	0.1825	0.1722	0.1396	0.0583	-0.1123	-0.0950	1.7356
45.0	0.1068	0.1308	0.4843	0.7424	0.8823	0.9363	0.9478	0.9363	0.8823	0.7424	0.4843	0.1308	0.1068	3.3199
67.5	0.2550	-0.3294	-0.2647	0.0304	0.3566	0.5935	0.6790	0.5935	0.3566	0.0304	-0.2647	-0.3294	0.2550	-1.6739
90.0	-0.0691	-0.8206	-1.7300	-2.2207	-2.2988	-2.1777	-2.1019	-2.1777	-2.2988	-2.2207	-1.7300	-0.8206	-0.0691	-8.8648
112.5	-0.4939	0.2577	-0.0876	-0.8389	-1.6075	-2.1629	-2.3660	-2.1629	-1.6075	-0.8389	-0.0876	0.2577	-0.4939	-1.6151
135.0	-0.2213	0.4470	1.1476	1.3116	0.8609	0.2401	-0.0382	0.2401	0.8609	1.3116	1.1476	0.4470	-0.2213	3.9928
157.5	0.1434	-0.0909	-0.0104	0.3291	0.7063	0.9948	1.1044	0.9948	0.7063	0.3291	-0.0104	-0.0909	0.1434	1.6420
180.0	0.2027	-0.0840	-0.5604	-0.5028	0.2440	1.1193	1.4947	1.1193	0.2440	-0.5028	-0.5604	-0.0840	0.2027	0.7481

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	-0.5523	1.9740	1.1558	0.7067	0.4611	0.3386	0.3015	0.3386	0.4611	0.7067	1.1558	1.9740	-0.5523
22.5	-0.3168	1.7583	1.0921	0.6869	0.4478	0.3229	0.2843	0.3229	0.4478	0.6869	1.0921	1.7583	-0.3168
45.0	0.3560	1.1560	1.1047	0.9579	0.8065	0.6978	0.6586	0.6978	0.8065	0.9579	1.1047	1.1560	0.3560
67.5	0.8499	-0.1165	0.7550	1.1934	1.3862	1.4504	1.4625	1.4504	1.3862	1.1934	0.7550	-0.1165	0.8499
90.0	-0.2302	-1.9134	-1.0365	-0.3758	0.1278	0.4644	0.5854	0.4644	0.1278	-0.3758	-1.0365	-1.9134	-0.2302
112.5	-1.6464	-1.9513	-1.8382	-1.6362	-1.4195	-1.2474	-1.1814	-1.2474	-1.4195	-1.6362	-1.8382	-1.9513	-1.6464
135.0	-0.7376	-0.4539	-0.5950	-0.6240	-0.7627	-0.9639	-1.0593	-0.9639	-0.7627	-0.6240	-0.5950	-0.4539	-0.7376
157.5	0.4781	0.3170	-0.0204	-0.2590	-0.4240	-0.5258	-0.5603	-0.5258	-0.4240	-0.2590	-0.0204	0.3170	0.4781
180.0	0.6756	0.4871	-0.0024	-0.4580	-0.5926	-0.5006	-0.4304	-0.5006	-0.5926	-0.4580	-0.0024	0.4871	0.6756

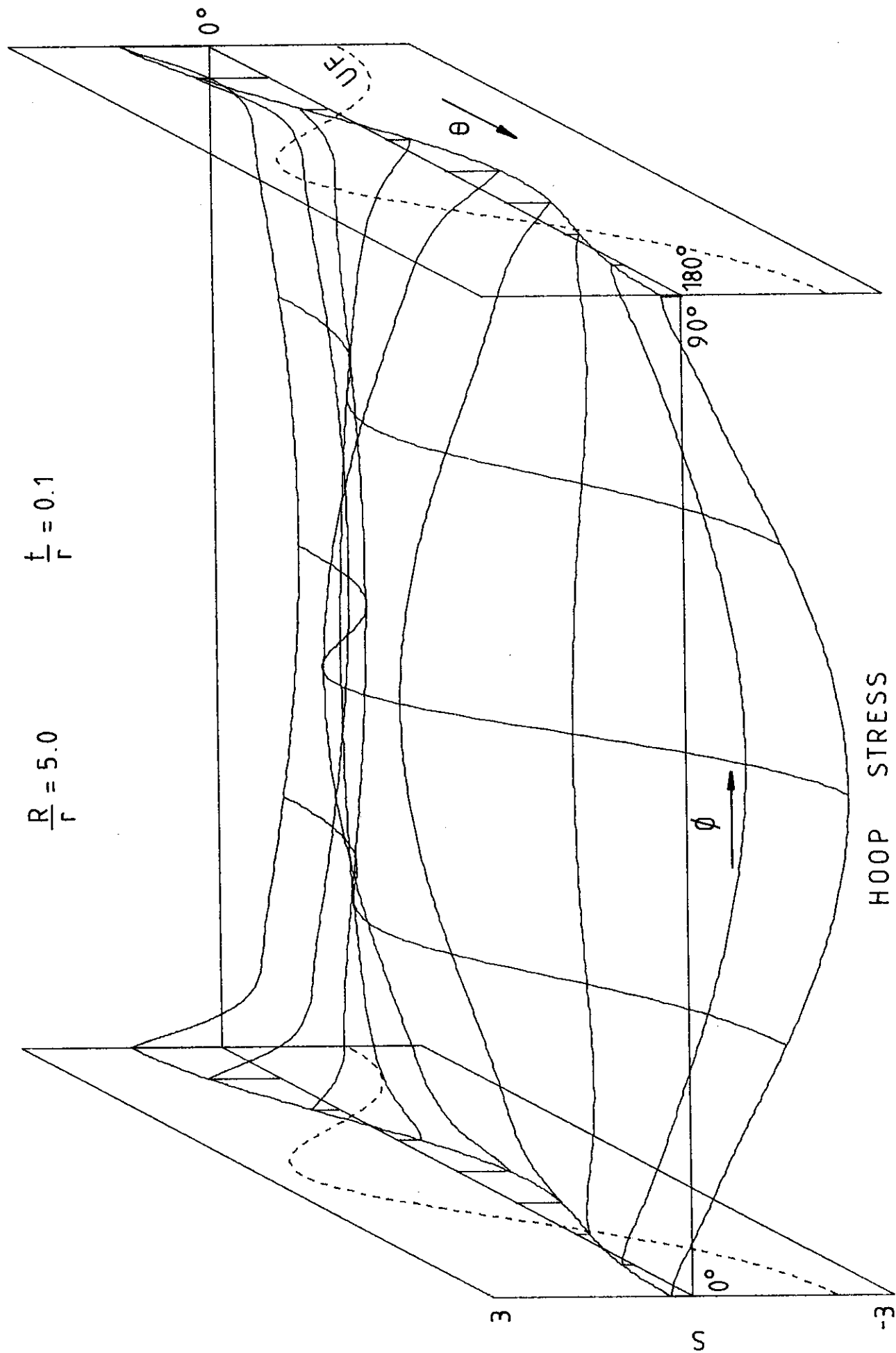
49

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
22.5	0.7752	0.5476	0.2957	0.1627	0.0866	0.0378	0.0	-0.0378	-0.0866	-0.1627	-0.2957	-0.5476	-0.7752	0.0
45.0	1.0272	0.9684	0.5945	0.3531	0.1931	0.0838	0.0	-0.0838	-0.1931	-0.3531	-0.5945	-0.9684	-1.0272	0.0
67.5	0.3611	0.8367	0.7176	0.5350	0.3450	0.1668	0.0	-0.1668	-0.3450	-0.5350	-0.7176	-0.8367	-0.3611	0.0
90.0	-0.6568	-0.2349	0.1930	0.3691	0.3631	0.2211	0.0	-0.2211	-0.3631	-0.3691	-0.1930	0.2349	0.6568	0.0
112.5	-0.7923	-1.0902	-0.7002	-0.3235	-0.0849	0.0030	0.0	-0.0030	0.0849	0.3235	0.7002	1.0902	0.7923	0.0
135.0	-0.2689	-0.6118	-0.8552	-0.8657	-0.7042	-0.3990	0.0	0.3990	0.7042	0.8657	0.8552	0.6118	0.2689	0.0
157.5	-0.0047	-0.0005	-0.3759	-0.7019	-0.7260	-0.4466	0.0	0.4466	0.7260	0.7019	0.3759	0.0005	0.0047	0.0
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS														
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
180.0	0.0	-6.970	-16.377	-31.526	-49.556	-64.249	-69.892	-64.249	-49.556	-31.526	-16.377	-6.970	0.0	-167.112



$$\frac{R}{r} = 5.0 \quad \frac{t}{r} = 0.1$$



HOOP STRESS

FIGURE A22

TABLE A21

R/r = 5.0 t/r = 0.05

Theta	Phi=0.0	OUTSIDE HOOP STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	1.8460	-0.5977	-0.6332	-0.6524	-0.6459	-0.6314	-0.6244	-0.6314	-0.6459	-0.6524	-0.6332	-0.5977	1.8460
22.5	1.4392	-0.5210	-0.6516	-0.7609	-0.8314	-0.8696	-0.8816	-0.8696	-0.8314	-0.7609	-0.6516	-0.5210	1.4392
45.0	0.3749	-0.1643	-0.3684	-0.6027	-0.8141	-0.9618	-1.0149	-0.9618	-0.8141	-0.6027	-0.3684	-0.1643	0.3749
67.5	-0.7969	0.4308	0.5433	0.5024	0.3799	0.2650	0.2194	0.2650	0.3799	0.5024	0.5433	0.4308	-0.7969
90.0	-1.2622	0.5453	1.1214	1.6112	1.9627	2.1783	2.2518	2.1783	1.9627	1.6112	1.1214	0.5453	-1.2622
112.5	-0.7404	-0.1584	0.2384	0.8977	1.5696	2.0598	2.2383	2.0598	1.5696	0.8977	0.2384	-0.1584	-0.7404
135.0	0.0198	-0.5732	-0.8038	-0.6921	-0.4324	-0.2094	-0.1252	-0.2094	-0.4324	-0.6921	-0.8038	-0.5732	0.0198
157.5	0.2966	-0.1114	-0.5359	-1.0673	-1.5625	-1.9093	-2.0335	-1.9093	-1.5625	-1.0673	-0.5359	-0.1114	0.2966
180.0	0.2880	0.2599	-0.0843	-0.8484	-1.6909	-2.3094	-2.5328	-2.3094	-1.6909	-0.8484	-0.0843	0.2599	0.2880

OUTSIDE AXIAL STRESS FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		0.0397	-0.0893	-0.1295	-0.0893	0.0397	0.2859	0.7133	1.4693	6.1534	
0.0	6.1534	1.4693	0.7133	0.7954	0.7954	0.1735	0.4164	0.7954	1.3933	4.7972	-0.0882 1.4157 2.7620 1.4342 -1.6735 -2.4956 -1.1395 -0.3935
22.5	4.7972	1.3933	0.7954	1.1233	1.0515	0.7682	0.9111	1.0515	1.1233	1.2495	
45.0	1.2495	1.1233	0.4078	1.0853	1.0853	1.6242	1.4391	1.0853	0.4078	-2.6562	
67.5	-2.6562	0.4078	-0.8829	0.1118	0.8687	1.4241	0.8687	0.1118	-0.8829	-4.2075	
90.0	-4.2075	-0.8829	-1.8202	-1.3540	-0.7630	-0.2473	-0.7630	-1.3540	-1.8202	-2.4681	
112.5	-2.4681	-1.8202	-1.2891	-1.5715	-1.5876	-1.5553	-1.5876	-1.5715	-1.2891	0.0661	
135.0	0.0661	-1.2891	-0.0133	-0.5659	-1.1531	-1.6289	-1.1531	-0.5659	0.0133	0.9887	
157.5	0.9387	0.0133	0.6198	0.0308	-0.7859	-1.4672	-0.7859	0.0308	0.6198	0.9599	-0.2880
180.0	0.9599	0.6198	0.0308	-0.7859	-1.4672	-1.8804	-0.7859	0.0308	0.6198	0.9599	

OUTSIDE SHEAR STRESS FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0
22.5	0.4686	0.3322	0.1903	0.2433	0.2433	0.0957	0.1212	-0.1903	-0.3322	-0.4686	
45.0	0.5798	0.4308	0.2433	0.2433	0.0957	0.0957	-0.1531	-0.2433	-0.4308	-0.5798	
67.5	0.2207	0.2248	0.1003	0.1003	0.0051	-0.0029	-0.0344	-0.1003	-0.2248	-0.2207	
90.0	-0.3049	-0.0767	-0.0953	-0.0953	-0.1358	0.1358	0.1352	0.0953	0.0767	0.3049	
112.5	-0.5244	-0.2422	-0.1406	-0.1406	-0.0640	0.1097	0.1274	0.1406	0.2422	0.5244	
135.0	-0.3589	-0.3026	-0.0946	-0.0946	-0.0765	-0.1041	-0.0495	0.0946	0.3026	0.3589	0.0 0.0 0.0 0.0 0.0 0.0 0.0
157.5	-0.1255	-0.2495	-0.0703	-0.0703	-0.1345	-0.1929	-0.1208	0.0703	0.2495	0.1255	
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	

DIAMETER EXPANSION FACTORS

Theta	Phi=0.0	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
		39.388	45.420	47.540	45.420	39.388	30.335	19.538	8.753	0.0	
0.0	0.0	39.388	45.420	47.540	45.420	39.388	30.335	19.538	8.753	0.0	80.504
22.5	7.5	19.538	15.0	15.0	15.0	60.0	67.5	75.0	82.5	90.0	
45.0	8.753	19.538	15.0	15.0	15.0	60.0	67.5	75.0	82.5	90.0	
67.5	0.0	8.753	19.538	15.0	15.0	60.0	67.5	75.0	82.5	90.0	
90.0	0.0	0.0	0.0	0.0	0.0	60.0	67.5	75.0	82.5	90.0	
112.5	0.0	0.0	0.0	0.0	0.0	60.0	67.5	75.0	82.5	90.0	
135.0	0.0	0.0	0.0	0.0	0.0	60.0	67.5	75.0	82.5	90.0	
157.5	0.0	0.0	0.0	0.0	0.0	60.0	67.5	75.0	82.5	90.0	80.504
180.0	0.0	0.0	0.0	0.0	0.0	60.0	67.5	75.0	82.5	90.0	

TABLE A22

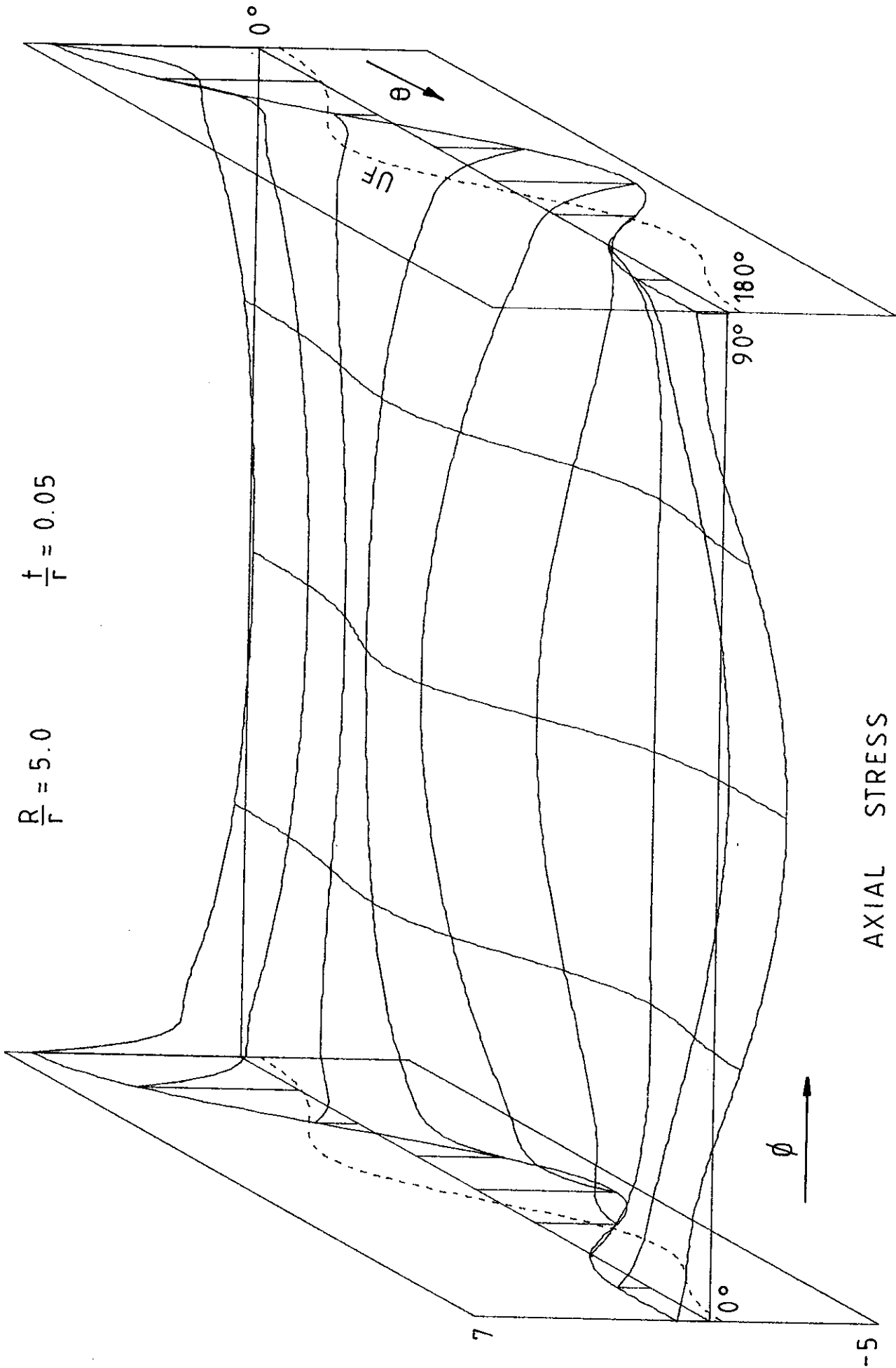
$$R/r = 5.0 \quad t/r = 0.05$$

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.2491	0.0646	0.3463	0.4774	0.5317	0.5473	0.5494	0.5473	0.5317	0.4774	0.3463	0.0646	-0.2491	1.9006
22.5	-0.1408	0.0793	0.4045	0.6111	0.7385	0.8067	0.8281	0.8067	0.7385	0.6111	0.4045	0.0793	-0.1408	2.0037
45.0	0.1127	-0.0716	0.1936	0.4857	0.7445	0.9234	0.9875	0.9234	0.7445	0.4857	0.1936	-0.0716	0.1127	1.0762
67.5	0.2824	-0.5070	-0.7065	-0.6651	-0.5169	-0.3764	-0.3205	-0.3764	-0.5169	-0.6651	-0.7065	-0.5070	0.2824	-2.2132
90.0	0.1199	-0.6131	-1.3269	-1.8678	-2.2409	-2.4651	-2.5411	-2.4651	-2.2409	-1.8678	-1.3269	-0.6131	0.1199	-4.7591
112.5	-0.2457	0.0119	-0.4308	-1.1447	-1.8882	-2.4381	-2.6398	-2.4381	-1.8882	-1.1447	-0.4308	0.0119	-0.2457	-2.3859
135.0	-0.3422	0.4092	0.6460	0.4971	0.1616	-0.1280	-0.2378	-0.1280	0.1616	0.4971	0.6460	0.4092	-0.3422	1.5193
157.5	-0.0945	0.0575	0.3456	0.7999	1.2604	1.5981	1.7216	1.5981	1.2604	0.7999	0.3456	0.0575	-0.0945	2.2301
180.0	0.0699	-0.2276	-0.1375	0.5084	1.3432	1.9932	2.2333	1.9932	1.3432	0.5084	-0.1375	-0.2276	0.0699	1.7163

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	-0.8304	1.5509	0.9385	0.5640	0.3334	0.2047	0.1629	0.2047	0.3334	0.5640	0.9385	1.5509	-0.8304
22.5	-0.4695	1.4256	1.0058	0.7303	0.5514	0.4480	0.4139	0.4480	0.5514	0.7303	1.0058	1.4256	-0.4695
45.0	0.3757	0.9423	1.0137	1.0232	1.0193	1.0152	1.0139	1.0152	1.0193	1.0232	1.0137	0.9423	0.3757
67.5	0.9412	-0.0346	0.4724	0.7823	0.9972	1.1321	1.1790	1.1321	0.9972	0.7823	0.4724	-0.0346	0.9412
90.0	0.3997	-1.0939	-0.6154	-0.3095	-0.0985	0.0289	0.0718	0.0289	-0.0985	-0.3095	-0.6154	-1.0939	0.3997
112.5	-0.8191	-1.3173	-1.1734	-1.1409	-1.1910	-1.2610	-1.2920	-1.2610	-1.1910	-1.1409	-1.1734	-1.3173	-0.8191
135.0	-1.1407	-0.6644	-0.7352	-0.8415	-1.0022	-1.1439	-1.1995	-1.1439	-1.0022	-0.8415	-0.7352	-0.6644	-1.1407
157.5	-0.3150	-0.0700	-0.2691	-0.3148	-0.2826	-0.2317	-0.2089	-0.2317	-0.2826	-0.3148	-0.2691	-0.0700	-0.3150
180.0	0.2330	0.0936	-0.1853	-0.1848	-0.0101	0.1804	0.2592	0.1804	-0.0101	-0.1848	-0.1853	0.0936	0.2330

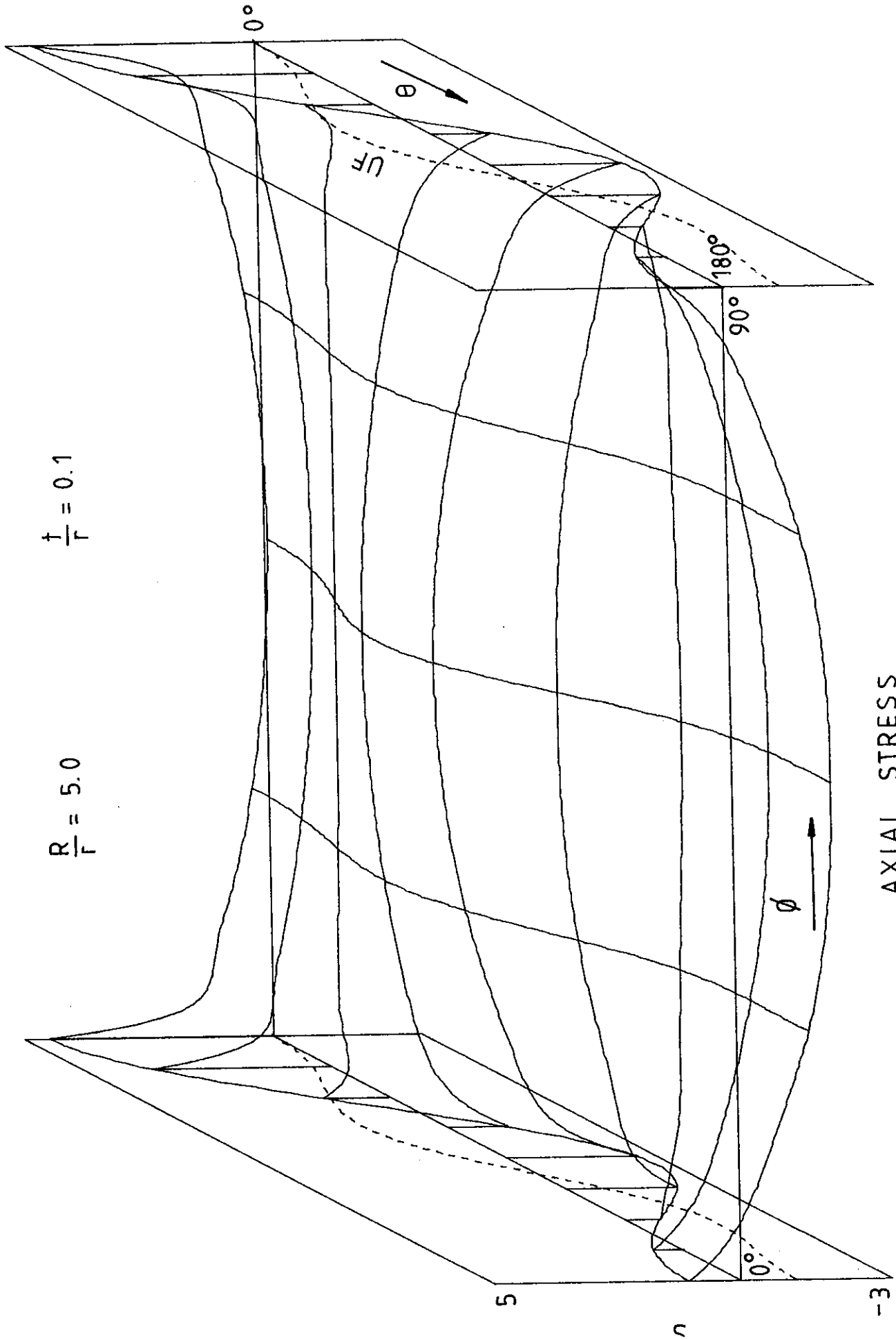
Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
22.5	0.4457	0.5008	0.2976	0.1783	0.1014	0.0461	0.0	-0.0461	-0.1014	-0.1783	-0.2976	-0.5008	-0.4457
45.0	0.5516	0.8064	0.5466	0.3680	0.2315	0.1130	0.0	-0.1130	-0.2315	-0.3680	-0.5466	-0.8064	-0.5516
67.5	0.2100	0.6358	0.5646	0.4566	0.3241	0.1692	0.0	-0.1692	-0.3241	-0.4566	-0.5646	-0.6358	-0.2100
90.0	-0.2900	-0.0668	0.1613	0.2320	0.2017	0.1135	0.0	-0.1135	-0.2017	-0.2320	-0.1613	0.0668	0.2900
112.5	-0.4988	-0.7292	-0.4674	-0.2947	-0.1831	-0.0915	0.0	0.0915	0.1831	0.2947	0.4674	0.7292	0.4988
135.0	-0.3414	-0.7240	-0.7786	-0.6975	-0.5172	-0.2736	0.0	0.2736	0.5172	0.6975	0.7786	0.7240	0.3414
157.5	-0.1194	-0.3055	-0.5435	-0.5752	-0.4436	-0.2339	0.0	0.2339	0.4436	0.5752	0.5435	0.3055	0.1194
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

DIAMETER EXPANSION FACTORS													
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
Theta	Phi=0.0	-6.606	-15.205	-25.354	-34.708	-41.201	-43.513	-41.201	-34.708	-25.354	-15.205	-6.606	0.0
180.0	0.0	-6.606	-15.205	-25.354	-34.708	-41.201	-43.513	-41.201	-34.708	-25.354	-15.205	-6.606	0.0
Without Flanges													
-67.994													



AXIAL STRESS

FIGURE A23



$$\frac{R}{r} = 5.0$$

$$\frac{t}{r} = 0.1$$

AXIAL STRESS

FIGURE A24

TABLE A23

$$R/r = 5.0 \quad t/r = 0.1$$

Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	Without Flanges
0.0	1.3465	-0.6406	-0.8134	-0.9994	-1.1411	-1.2319	-1.2634	-1.2319	-1.1411	-0.9994	-0.8134	-0.6406	1.3465	-1.8737
22.5	1.0778	-0.4959	-0.6612	-0.8479	-0.9952	-1.0906	-1.1237	-1.0906	-0.9952	-0.8479	-0.6612	-0.4959	1.0778	-1.5312
45.0	0.3937	-0.1088	-0.1753	-0.2933	-0.3978	-0.4675	-0.4917	-0.4675	-0.3978	-0.2933	-0.1753	-0.1088	0.3937	-0.3864
67.5	-0.3680	0.3082	0.5117	0.6316	0.7104	0.7584	0.7750	0.7584	0.7104	0.6316	0.5117	0.3082	-0.3680	1.3368
90.0	-0.7928	0.4031	0.8737	1.2934	1.6260	1.8392	1.9126	1.8392	1.6260	1.2934	0.8737	0.4031	-0.7928	2.3883
112.5	-0.6744	0.0782	0.4758	0.9316	1.3190	1.5690	1.6545	1.5690	1.3190	0.9316	0.4758	0.0782	-0.6744	1.5639
135.0	-0.2174	-0.2688	-0.3011	-0.2166	-0.1238	-0.0629	-0.0425	-0.0629	-0.1238	-0.2166	-0.3011	-0.2688	-0.2174	-0.4198
157.5	0.1780	-0.3000	-0.7478	-1.1513	-1.4786	-1.6860	-1.7565	-1.6860	-1.4786	-1.1513	-0.7478	-0.3000	0.1780	-1.8049
180.0	0.3134	-0.2301	-0.8248	-1.4404	-1.9573	-2.2850	-2.3958	-2.2850	-1.9573	-1.4404	-0.8248	-0.2301	0.3134	-2.1632

OUTSIDE AXIAL STRESS FACTORS

Theta	Phi=0.0	OUTSIDE AXIAL STRESS FACTORS												Without Flanges
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	4.4883	1.1074	0.6417	0.2984	0.0801	-0.0452	-0.0864	-0.0452	0.0801	0.2984	0.6417	1.1074	4.4883	0.0116
22.5	3.5926	1.0927	0.7542	0.4908	0.3224	0.2268	0.1957	0.2268	0.3224	0.4908	0.7542	1.0927	3.5926	0.3788
45.0	1.3125	0.9563	0.9664	0.9390	0.9245	0.9210	0.9210	0.9210	0.9245	0.9390	0.9664	0.9563	1.3125	1.2039
67.5	-1.2265	0.4953	0.9170	1.2047	1.4030	1.5251	1.5668	1.5251	1.4030	1.2047	0.9170	0.4953	-1.2265	1.6493
90.0	-2.6428	-0.3196	0.2853	0.7574	1.0747	1.2584	1.3184	1.2584	1.0747	0.7574	0.2853	-0.3196	-2.6428	0.8777
112.5	-2.2479	-1.0453	-0.6664	-0.3218	-0.1110	-0.0068	0.0235	-0.0068	-0.1110	-0.3218	-0.6664	-1.0453	-2.2479	-0.7144
135.0	-0.7247	-1.0950	-1.1923	-1.2154	-1.2604	-1.3067	-1.3261	-1.3067	-1.2604	-1.2154	-1.1923	-1.0950	-0.7247	-1.6442
157.5	0.5934	-0.5619	-1.0660	-1.4471	-1.7031	-1.8433	-1.8874	-1.8433	-1.7031	-1.4471	-1.0660	-0.5619	0.5934	-1.4255
180.0	1.0448	-0.2298	-0.8831	-1.4079	-1.7362	-1.8976	-1.9443	-1.8976	-1.7362	-1.4079	-0.8831	-0.2298	1.0448	-1.1114

OUTSIDE SHEAR STRESS FACTORS

Theta	Phi=0.0	OUTSIDE SHEAR STRESS FACTORS												Without Flanges
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
22.5	0.2191	0.1392	0.0582	0.0224	0.0063	0.0006	0.0	-0.0006	-0.0063	-0.0224	-0.0582	-0.1392	-0.2191	
45.0	0.2754	0.1736	0.0517	-0.0033	-0.0209	-0.0162	0.0	0.0162	0.0209	0.0033	-0.0517	-0.1736	-0.2754	
67.5	0.1180	0.0883	-0.0211	-0.0721	-0.0750	-0.0457	0.0	0.0457	0.0750	0.0721	0.0211	-0.0883	-0.1180	
90.0	-0.1436	-0.0269	-0.0753	-0.1000	-0.0882	-0.0502	0.0	0.0502	0.0882	0.1000	0.0753	0.0269	0.1436	
112.5	-0.3197	-0.0847	-0.0272	-0.0013	0.0080	0.0071	0.0	-0.0071	-0.0080	0.0013	0.0272	0.0847	0.3197	
135.0	-0.3110	-0.0934	0.0720	0.1568	0.1530	0.0907	0.0	-0.0907	-0.1530	-0.1568	-0.0720	0.0934	0.3110	
157.5	-0.1734	-0.0709	0.0892	0.1767	0.1691	0.0988	0.0	-0.0988	-0.1691	-0.1767	-0.0892	0.0709	0.1734	
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	

DIAMETER EXPANSION FACTORS

		DIAMETER EXPANSION FACTORS												Without Flanges
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0	
90.0	0.0	4.565	10.140	15.433	19.712	22.468	23.417	22.468	19.712	15.433	10.140	4.565	0.0	28.768

TABLE A24

R/r = 5.0 $\tau/r = 0.1$

Theta	Phi=0.0	INSIDE HOOP STRESS FACTORS										Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0		82.5	90.0
0.0	-0.2167	0.2482	0.6187	0.8899	1.0827	1.2014	1.2419	1.2014	1.0827	0.8899	0.6187	0.2482	-0.2167	1.8790
22.5	-0.1301	0.1468	0.4768	0.7398	0.9363	1.0595	1.1017	1.0595	0.9363	0.7398	0.4768	0.1468	-0.1301	1.5097
45.0	0.0697	-0.1439	0.0025	0.1648	0.3053	0.3969	0.4285	0.3969	0.3053	0.1648	0.0025	-0.1439	0.0697	0.2535
67.5	0.2278	-0.4775	-0.6960	-0.8312	-0.9122	-0.9612	-0.9783	-0.9612	-0.9122	-0.8312	-0.6960	-0.4775	0.2278	-1.6729
90.0	0.1927	-0.5346	-1.0689	-1.5697	-1.9607	-2.2122	-2.2992	-2.2122	-1.9607	-1.5697	-1.0689	-0.5346	0.1927	-2.8760
112.5	-0.0274	-0.2040	-0.6508	-1.2085	-1.6843	-1.9934	-2.0994	-1.9934	-1.6843	-1.2085	-0.6508	-0.2040	-0.0274	-1.9741
135.0	-0.2351	0.1299	0.1209	-0.0171	-0.1594	-0.2517	-0.2823	-0.2517	-0.1594	-0.0171	0.1209	0.1299	-0.2351	0.2374
157.5	-0.2837	0.1364	0.4849	0.9077	1.2687	1.5044	1.5857	1.5044	1.2687	0.9077	0.4849	0.1364	-0.2837	1.7682
180.0	-0.2648	0.0533	0.5028	1.1704	1.7650	2.1525	2.2850	2.1525	1.7650	1.1704	0.5028	0.0533	-0.2648	2.1553

Theta	Phi=0.0	INSIDE AXIAL STRESS FACTORS									Without Flanges		
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5		75.0	82.5
0.0	-0.7222	1.3686	0.9546	0.7684	0.6574	0.5987	0.5803	0.5987	0.6574	0.7684	0.9546	1.3686	-0.7222
22.5	-0.4336	1.2265	0.9408	0.8209	0.7534	0.7202	0.7103	0.7202	0.7534	0.8209	0.9408	1.2265	-0.4336
45.0	0.2323	0.7818	0.7920	0.8204	0.8478	0.8686	0.8763	0.8686	0.8478	0.8204	0.7920	0.7818	0.2323
67.5	0.7592	0.0815	0.3356	0.4639	0.5435	0.5878	0.6021	0.5878	0.5435	0.4639	0.3356	0.0815	0.7592
90.0	0.6422	-0.5998	-0.3316	-0.2528	-0.2340	-0.2380	-0.2419	-0.2380	-0.2340	-0.2528	-0.3316	-0.5998	0.6422
112.5	-0.0915	-0.8827	-0.7733	-0.8195	-0.9012	-0.9709	-0.9975	-0.9709	-0.9012	-0.8195	-0.7733	-0.8827	-0.0915
135.0	-0.7836	-0.7060	-0.7229	-0.7868	-0.8484	-0.8929	-0.9088	-0.8929	-0.8484	-0.7868	-0.7229	-0.7060	-0.7836
157.5	-0.9456	-0.4060	-0.4419	-0.3917	-0.3067	-0.2387	-0.2133	-0.2387	-0.3067	-0.3917	-0.4419	-0.4060	-0.9456
180.0	-0.8827	-0.2820	-0.3074	-0.1828	-0.0102	0.1228	0.1718	0.1228	-0.0102	-0.1828	-0.3074	-0.2820	-0.8827

Theta	Phi=0.0	INSIDE SHEAR STRESS FACTORS										Without Flanges	
		7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
22.5	0.1982	0.3937	0.2507	0.1711	0.1101	0.0546	0.0	-0.0546	-0.1101	-0.1711	-0.2507	-0.3937	-0.1982
45.0	0.2492	0.6127	0.4206	0.2978	0.1939	0.0963	0.0	-0.0963	-0.1939	-0.2978	-0.4206	-0.6127	-0.2492
67.5	0.1068	0.4998	0.3986	0.2994	0.1971	0.0973	0.0	-0.0973	-0.1971	-0.2994	-0.3986	-0.4998	-0.1068
90.0	-0.1299	0.0468	0.1140	0.1050	0.0694	0.0325	0.0	-0.0325	-0.0694	-0.1050	-0.1140	-0.0468	0.1299
112.5	-0.2892	-0.4736	-0.3168	-0.2210	-0.1460	-0.0738	0.0	0.0738	0.1460	0.2210	0.3168	0.4736	0.2892
135.0	-0.2814	-0.6765	-0.5850	-0.4444	-0.2894	-0.1406	0.0	0.1406	0.2894	0.4444	0.5850	0.6765	0.2814
157.5	-0.1569	-0.4519	-0.4538	-0.3566	-0.2286	-0.1081	0.0	0.1081	0.2286	0.3566	0.4538	0.4519	0.1569
180.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

		DIAMETER EXPANSION FACTORS										Without Flanges	
Theta	Phi=0.0	7.5	15.0	22.5	30.0	37.5	45.0	52.5	60.0	67.5	75.0	82.5	90.0
180.0	0.0	-4.568	-9.855	-15.086	-19.364	-22.121	-23.069	-22.121	-19.364	-15.086	-9.855	-4.568	0.0
													-27.652

